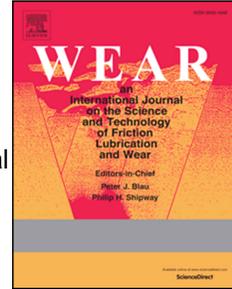


# Journal Pre-proof

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PII: S0043-1648(21)00123-X

DOI: <https://doi.org/10.1016/j.wear.2021.203734>

Reference: WEA 203734

To appear in: *Wear*

Received Date: 7 September 2020

Revised Date: 8 January 2021

Accepted Date: 25 January 2021

Please cite this article as: G.V. Messa, Y. Wang, M. Negri, S. Malavasi, An improved CFD/experimental combined methodology for the calibration of empirical erosion models (Abstract Ref. number: WEAR2021\_0102), *Wear*, <https://doi.org/10.1016/j.wear.2021.203734>.

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# An improved CFD/experimental combined methodology for the calibration of empirical erosion models (Abstract Ref. number: WEAR2021\_0102)

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## Abstract

A significant source of uncertainty in CFD-based erosion prediction is represented by the erosion model, used to turn the particle-wall impingement characteristics into an estimate of the removal of material. In engineering computations, frequent use is made of empirical erosion models obtained by fitting the experimental results of air-solid jet impingement tests. However, since the applicability conditions of these models are often unknown or unavailable to the users of CFD codes, they are frequently misapplied, producing not only uncertain, but at times also highly inaccurate wear estimates. As part of his PhD thesis defended at the University of Tulsa in 2016, Amir Mansouri proposed a methodology to calibrate the coefficients of an empirical erosion correlation by combining experimental data and CFD results of a slurry jet impingement test. This methodology provided an effective way to overcome the criticisms of using empirical erosion models taken from the literature without facing the complexity and the high computational burden of a multiscale, physically-based approach, in which the micro-scale mechanical interactions between the particles and the target surface are resolved according to the fundamental laws of damage mechanics. The present paper presents an improved formulation of Mansouri's methodology, which not only increases the calibration accuracy of the empirical erosion equation, but also enhances the reliability of the CFD-based erosion prediction model as a whole. To this purpose, numerical simulations were supported by in-house slurry jet impingement experiments on an aluminum sample and curved Glass Reinforced Epoxy samples.

*Keywords:* Computational Fluid Dynamics, erosion models, direct impact test, impact erosion

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## 1. Introduction

The numerical modelling of solid particle erosion in either dry or wet environments poses significant challenges due to the multi-scale nature of the wear process. Achieving reliable erosion estimates requires simulating the motion of the solid particles in the carrier fluid and accurately predicting the particle-wall impingement characteristics, namely, establish when and where the impacts occur, and what the velocity vector of each particle is at the stage of impingement. Achieving this task is extremely difficult for most scenarios of engineering interest, involving turbulent particle-laden flows with natural solids. The excessive computational cost of fully-resolved simulations based on the direct solution of the instantaneous flow equations, in fact, requires a number of modelling simplifications and assumptions, such as Reynolds-averaging of the governing equations, point-particle approximation, use of force models to model the interactions between the particles and the surrounding fluid (drag, pressure force, lift, etc). These modelling simplifications introduce sub-models and parameters which undermine the physical foundation of the mathematical model and, being often difficult-to-quantify, produce inherent uncertainty in the simulation results [1].

The most common approach in erosion prediction simulations based on Computational Fluid Dynamics (CFD) relies on the use of Reynolds-averaged based Eulerian-Lagrangian models, which solve for some Reynolds-averaged formulation of the fluid flow equations on an Eulerian grid of cells, and determine the

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Preprint submitted to *Wear of Materials* 2021

17 trajectories of a number of computational particles by integrating the Lagrangian particle equations of motion [2]. Such computational particles are often referred to as “parcels” (hereafter denoted by the letter “P”), which, in steady-state computations, are associated to a given solid mass flow rate,  $\dot{m}_p$ . Eulerian-Lagrangian models make it straightforward to evaluate, for each collision between a parcel and a wall boundary, the location of the impact, the modulus of the impact velocity,  $v_{p,imp}$ , and the angle formed by the impact velocity vector and the wall,  $\theta_{p,imp}$ . Typically, the interactions among the particles are ignored, and often also the influence of the particles on the fluid flow. This restricts the applicability of this approach to particle-laden flows with low solid concentration, say, not higher than about one percent by volume of mixture, which are, anyway, of interest for many engineering processes in the oil and gas sector. Indeed, some investigations have been reported in which particle-particle interactions were accounted for by Lagrangian inter-particle collision models (e.g. [3, 4]): these models extend the range of applicability to higher solid content, but they still remain an exception, probably because of the increase in the computational burden of the simulations and, not secondarily, also in the number of difficult-to-estimate parameters coming into play.

30 At the same time, predicting solid particle erosion requires being capable to model the wear phenomenon at the micro-scale level, determining the amount of material removal from a solid body subjected to subsequent collisions from the travelling particles. This relates with the concepts of damage and fracture mechanics of solids, but a physically-based approach is rarely adopted in CFD-based erosion prediction studies. Indeed, a number of papers have been published in which Finite Element Method (FEM) or alternative methods have been employed to simulate the particle-wall impingement processes at the micro-scale and accurately predict the removal of material for specific combinations of abrasive particles and target body, but they typically ignored the influence of the fluid on the particle transport at the macro-scale level [5–7]. Such an assumption might be reasonable for high-speed, gas-particle flows, but this is not likely to be the case for liquid-solid flows. Additionally, the results obtained in these studies allow a deep understanding of physical mechanisms at the basis of material removal, but they appear to be case-specific and difficult to integrate in a CFD-based erosion prediction model.

42 [Figure 1 about here.]

43 Among CFD modelers, who mainly focus on the particle transport side of the erosion phenomenon, it is a common practice to employ simple equations to turn the particle-wall impingement characteristics into an estimate of the amount of material removal. Usually, these equations are empirically obtained from dry direct impact tests, in which a particle-laden air jet is directed against a target sample for a period of time, producing an erosion hole (Fig. 1a). The fitting of the experimental results allows expressing the following functional relation in a simple mathematical form:

$$\Delta M = \dot{M}_s T \cdot f(V_{jet}, \theta_{jet}, \text{particle}, \text{target}) \quad (1)$$

49 where  $\Delta M$  is the difference in the mass of the sample before and after the test,  $\dot{M}_s$  is the mass flow rate of abrasives leaving the nozzle,  $T$  is the duration of the test,  $V_{jet}$  is the jet bulk-mean velocity,  $\theta_{jet}$  is the nozzle-to-specimen angle, and the terms “particle” and “target” are used to denote the geometrical and physical properties of the abrasives (e.g. size, shape, density, hardness etc.) and of the physical properties of target material (es. density, hardness etc.), respectively. Since the motion of the particles in a dry direct impact test is likely to be inertia-dominated, there is no dependence upon the nozzle-to-specimen distance  $H$  in Eq. 1. The same hypothesis on the fluid dynamic behavior of the particles provides a rationale for the coupling between Eq. 1 and an Eulerian-Lagrangian simulation. Since all particles are assumed to hit the target with an impact velocity approximately equal to the  $V_{jet}$  and with an impact angle approximately equal to  $\theta_{jet}$ , it is argued that the same functional relationship derived at the macro-scale of the jet applies also at the micro-scale of each parcel-wall impingement, that is

$$\dot{E}_{p,imp} = \dot{m}_p \cdot f(v_{p,imp}, \theta_{p,imp}, \text{particle}, \text{target}) \quad (2)$$

60 where  $\dot{E}_{p,imp}$  is the mass flow rate of removed material produced by the current parcel-wall impingement.

61 Undoubtedly, Eq. 2 is easy to incorporate in a CFD-based erosion prediction model based on Eulerian-Lagrangian simulations, and therefore, it provides a simple way to obtain erosion estimates even for arbitrarily complex particle-laden flows in dry and wet conditions. At the same time, the validity of such approach has serious limitations because of the following reasons. Firstly, it assumes that all particles used in the experiments are equal to each other, but this is not likely to be the case for natural materials like sand. Secondly, it ignores that, from a mechanical point of view, erosive wear is a highly nonlinear process and, therefore, the mass removal from the sample is not a simple superimposition of the effects of the single impingements. Thirdly, it ignores the nonlinear time effects caused by the change in particle-laden flow

69 field occurring as the erosion hole becomes deeper during the experiments. Fourthly, the jet velocity in dry  
 70 direct impact tests is usually higher compared with the impact velocities commonly found in slurry flows;  
 71 therefore, erosion models derived from dry direct impact tests are outside their calibration range when they  
 72 are applied to wet particle-laden flows. Finally, and partially related with the point above, the calibration  
 73 database and the applicability conditions of empirical erosion models are often unknown to CFD users and,  
 74 therefore, these models are frequently misapplied, yielding results of doubtful validity. The widely used  
 75 erosion model by Oka et al. [8, 9] lies in the category of empirical erosion models obtained from fitting  
 76 experimental results of dry abrasive jet impingement tests. Its form is as reported in Eq. 1, where the inde-  
 77 pendent variables related with “particle” and “target” are the particle diameter, the Vickers hardness of the  
 78 target, and the density of the target. A set of fitting constants is provided by the experimenters for different  
 79 abrasive materials (quartz, silicon carbide, glass beads). Two other parameters must be determined from  
 80 load relaxation tests on the target material.

81 Actually, valuable attempts have been made to conjugate accurate modelling of the particle-laden flow  
 82 at the macroscale with a physically based modelling of the erosive wear at the microscale. The multi-  
 83 scale approach of Leguizamon et al. [10] fits precisely into this framework. The approach relies on the  
 84 decoupling between the micro-scale and macro-scale processes underlying solid particle erosion. Firstly,  
 85 micro-scale simulations are performed to estimate the amount of material removal caused by a train of  
 86 sediments impinging against the target surface for different combinations of velocity and inclination angle.  
 87 Afterwards, a macro-scale model is employed to simulate the transport of the solid particles. Each time a  
 88 particle collides against a solid wall, the impact velocity and the impact angle are stored, and the eroded  
 89 mass is extracted from the database of the micro-scale results previously obtained. Compared to the em-  
 90 pirical erosion models derived from dry impingement tests, the main strength of the multi-scale approach  
 91 resides in the fact that the values of mass removal are not obtained from some interpolatory formula, but  
 92 from a database of physically-based simulation results. At the same time, a relevant drawback resides in  
 93 the high computational cost. In order to generate a comprehensive and reliable database of micro-scale  
 94 results, even limited to a specific target material, many combinations of particle impact velocities, particle  
 95 impact angles, and properties of abrasive particles (density, hardness, size, shape etc.) must be simulated.  
 96 In summary, the multi-scale approach of Leguizamon et al. [10] pushed the frontiers of the state-of-the-art  
 97 in CFD-based erosion prediction simulations, opening the way to the possibility of predicting erosion in  
 98 complex geometries, such as hydro-turbines, with a high degree of fidelity [11, 12]. At the same time, there  
 99 are still some concerns on the practical usability of this approach in a typical engineering context.

100 For the sake of completeness, it is worth mentioning that also phenomenological erosion models have  
 101 been proposed, which lie inbetween physically-based simulations of the erosive wear at the micro-scale  
 102 and fully empirical correlations derived from dry direct impact tests. Phenomenological erosion models  
 103 date back to the pioneering studies of Finnie [13] and Bitter [14, 15], in which the removal of material is  
 104 interpreted as a consequence of two erosion processes, namely, cutting wear and deformation wear. Cutting  
 105 wear is associated with a shearing action and it mainly occurs at low impact angles. Deformation wear,  
 106 instead, is typical of high impact angles, and it is associated with repeated particle impacts causing plastic  
 107 deformation, hardening, and sub-surface cracking. Several phenomenological erosion models have been  
 108 proposed across the last decades, but even the most recent ones (e.g. [16, 17]) deviated only slightly from  
 109 the original idea and formulation of Bitter. Phenomenological erosion models allow a certain insight into  
 110 the main mechanisms at the basis of the wear process, but the critical drawback resides in the high number  
 111 of parameters involved. Some of these parameters are not given a precise physical meaning, whereas others  
 112 are difficult-to-quantify. In the end, phenomenological erosion models have a lot of empirical content, and  
 113 they are formally more complicated than fully-empirical correlations. Given the current state of the art,  
 114 the authors of this paper are convinced that alternative strategies should be explored to obtain an effective  
 115 erosion prediction tool for engineers.

116 The considerations drawn before suggest that **a critical issue in CFD-based erosion prediction modelling**  
 117 **is how to convert the particle-wall impingement characteristics into estimates of material removal.** Few  
 118 **years ago, as part of his doctoral work at the University of Tulsa, Amir Mansouri proposed a methodology**  
 119 **to develop an erosion model of the form of Eq. 2 starting from a combination of experimental and simulation**  
 120 **results of a wet direct impact test. As sketched in Fig. 1b, the motion of the particles in a wet direct impact**  
 121 **test is strongly influenced by the carrier liquid and, as a result, there is a wide range of impact velocities**  
 122 **and impact angles. By combining the distributions of the parcel impact velocities, parcel impact angles,**  
 123 **and parcel impact number density (that is, the number of impingements per unit wall area), as predicted by**  
 124 **an Eulerian-Lagrangian model, and the local erosion depth, as measured by a profilometer, it is possible**  
 125 **to calibrate an algebraic equation representing the functional relation of Eq. 2. Mansouri described his**  
 126 **procedure in a co-authored research article published in the Wear journal in 2015 [18], and also in a chapter**  
 127 **of his PhD thesis, defended one year later [19]. Details of the operative procedure will be provided later**

128 in Section 4.1 of this paper. At this stage, it is simply noted that the methodology of Mansouri et al. [18]  
 129 represents a practical compromise to overcome the drawbacks of empirical erosion models derived from dry  
 130 direct impact tests. This methodology produces a fully empirical erosion model, which can be reasonably  
 131 employed for erosion prediction in flows involving the same materials, impact velocities, and impact angles,  
 132 of the wet direct impact test considered for the calibration.

133 In this paper, an upgraded version of the methodology of Mansouri [19] is proposed, with a twofold ob-  
 134 jective. On the one hand, the new methodology improves the accuracy of calibration of the erosion equation.  
 135 On the other hand, it gives the methodology a wider scope, as the new procedure allows not only finding  
 136 a case-specific erosion correlation, but also properly characterizing the effect of particle shape in the CFD-  
 137 based erosion prediction model, which is one of the most significant and critical sources of uncertainty. The  
 138 computational work was developed around two test cases, which were subject of an experimental campaign  
 139 conducted in the Hydraulic Laboratory of Politecnico di Milano. These are the normal impingements of a  
 140 slurry jet on a flat aluminium sample and on curved samples made of Glass Reinforced Epoxy (GRE). The  
 141 remainder of the paper is organized in four sections, followed by the conclusions. Section 2 is dedicated to  
 142 the experimental setup and methodology. Section 3 illustrates the CFD model for predicting the particle-  
 143 laden flow. Section 4 starts with an overview of the original methodology of Mansouri [19], and then  
 144 focuses on its upgraded formulation. Finally, in Section 5, the results for the two test cases are presented  
 145 and discussed.

## 146 2. Experimental campaign

### 147 2.1. Experimental setup

148 The experiments were performed using the wet direct impact test facility located in the Hydraulic Lab-  
 149 oratory of Politecnico di Milano. The setup consists of three main components, namely, a mixing tank,  
 150 a centrifugal pump, and the test cabin, connected by a stainless steel piping with internal diameter of 21  
 151 mm (Fig. 2). The mixing tank has a capacity of about 270 liters, and it is equipped with a variable-speed  
 152 impeller to keep the slurry in suspension. A first pipe segment connects the tank to the centrifugal pump,  
 153 which is capable in reaching pressures up to 9 bar and flow rates up to about 24 m<sup>3</sup>/h. Downstream of it,  
 154 a second pipe segment is interrupted by a valve, which is used to control the flow rate. Finally, the main  
 155 piping line ends inside the test cabin with a nozzle. Two nozzles have been used in the experiments, namely,  
 156 a stainless steel nozzle with inner diameter of 10.4 mm for the test on the flat aluminum specimen, and a  
 157 tungsten carbide nozzle with inner diameter of 8.0 mm for the tests on the curved GRE samples. The test  
 158 cabin encloses also the support of the eroding samples and a screw driven system to adjust the nozzle-to-  
 159 specimen distance,  $H$ , and the nozzle-to-specimen angle,  $\theta_{\text{jet}}$ . In all experiments,  $\theta_{\text{jet}}$  was set of 90°, the  
 160 nozzle was completely submerged by water, and maintaining a constant height of water in the upper tank  
 161 was possible by means of a second regulation valve placed in the flexible hose connecting the test cabin  
 162 back to the mixing tank.

163 [Figure 2 about here.]

### 164 2.2. Measured variables and instrumentation

165 Controlled variables were the jet bulk-mean velocity,  $V_{\text{jet}}$ , the nozzle-to-specimen distance,  $H$ , the  
 166 nozzle-to-specimen inclination angle,  $\theta_{\text{jet}}$ , and the testing time,  $T$ . The jet bulk-mean velocity was in-  
 167 ferred from the volumetric flow rate, measured using a Prosonic Flow 91WA1 ultrasonic flow metering  
 168 system installed in the pipe connecting the mixing tank to the pump. The measured values of  $V_{\text{jet}}$  were  
 169 combined with pressure measurements made by an absolute pressure transducer located upstream the noz-  
 170 zle, so that, by comparison with an experimentally-determined characteristic curve of the brand new nozzle,  
 171 it was possible to monitor the degradation of the nozzle due to erosion. The nozzle-to-specimen distance  
 172 and inclination angle were set manually using the screw driven system inside the test cabin;  $H$  was 12.7 mm  
 173 in the test on the aluminum sample, and 20.0 mm in the tests on the GRE samples; as already mentioned,  
 174  $\theta_{\text{jet}}$  was kept fixed at 90°. The testing time was measured by a standard digital clock.

175 The solid concentration was monitored during the tests by sampling the slurry exiting the hose connect-  
 176 ing the test cabin to the stirred tank (point S in Fig. 2). This required a graduate flask, a balance, and a  
 177 thermometer. The flask, which was used to estimate the sampling volume, had a capacity of 1.0 liter and  
 178 notches every 1 ml from 990 ml to 1010 ml, with an accuracy of 0.4 ml. The weight of the sampling vol-  
 179 ume was calculated as the difference between the weight of the flask with the sampling volume and that  
 180 of the empty and dry flask. The flask with the sampling volume was weighted using a laboratory balance  
 181 (PCE-BS 3000) with accuracy of 0.3 g. The empty and dry flask was previously weighted with a balance

182 with a much higher accuracy and periodically checked through the test sessions. Finally, the thermometer,  
 183 having accuracy of  $0.5^\circ \text{C}$ , was used to measure the temperature of the sample. By assuming the absence  
 184 of slip between the solid particles and the carrier fluid and the representativeness of the sampling point, the  
 185 solid concentration at the nozzle exit,  $C$ , could be obtained from the density of the slurry, as follows:

$$\frac{M_{\text{sample}}}{W_{\text{sample}}} = \rho_l (1 - C) + \rho_p C \quad (3)$$

186 where  $M_{\text{sample}}$  is the mass of the sample, calculated as a difference between the mass of the filled flask and  
 187 that of the empty flask,  $W_{\text{sample}}$  is the volume of the sample, obtained from the notches of the flask,  $\rho_l$  is the  
 188 density of the carrier liquid, inferred from temperature using a standard correlation for water [20], and  $\rho_p$  is  
 189 the density of the solid particles, provided by the retailer. The mass loss of the sample caused by erosion,  
 190  $\Delta M$ , was obtained by weighting the sample before and after the test using a Mettler Toledo AE200 balance,  
 191 with accuracy of 0.3 mg and full-scale value of 200 g. In order to determine the shape of the erosion crater,  
 192 the surface of the samples at the end of the tests was scanned with a resolution grid of 0.5 mm in both  
 193 directions. To this aim, use was made of a laser-triangulation CMOS CCD profilometer, in which the laser  
 194 was secured to two axes controlled by servomotors, and an optoelectronic system integrated with a digital  
 195 signal processor measured the displacements with respect to a target without contact with the object, by  
 196 means of a triangulation method (measuring range of 50 mm; linearity of  $\pm 0.2\%$ ; static resolution of  $5 \mu\text{m}$ ,  
 197  $25 \mu\text{m}$  dynamic resolution, 1 kHz measurement rate). For all tests, the values of  $\Delta M$  obtained from weight  
 198 measurements were about 20-30% larger than the corresponding estimates calculated by multiplying the  
 199 density of the target material by the volume of the erosion hole, which, in turn, was obtained by numerically  
 200 integrating the data acquired with the profilometer. Such deviation was judged satisfactory by the authors,  
 201 considering that errors might have occurred in estimating the surface of the undamaged specimens from the  
 202 profilometer data.

### 203 2.3. Sources of uncertainty in the experimental results

204 Uncertainty was evaluated for all experimental parameters coming into play in the comparison with the  
 205 numerical simulations. Starting from the indications provided in the datasheet of the flowmeter, the  
 206 uncertainty of the jet bulk-mean velocity,  $V_{\text{jet}}$ , was estimated as  $\pm 2.4\%$  of the measured value. The nozzle-  
 207 to-specimen distance,  $H$ , was set by turning a longitudinal screw manually, and verified using a caliber with  
 208 a tolerance of 0.05 mm. The nozzle-to-specimen angle,  $\theta_{\text{jet}}$ , was set by turning another screw and verified  
 209 with a goniometer, and it reasonable to assume an uncertainty of about  $\pm 1^\circ$ .

210 The uncertainty in the solid volume fraction at the nozzle outlet was a combination of three factors.  
 211 Firstly, the propagation of the uncertainties of the other terms in Eq. 3, which yielded  $\pm 3 \cdot 10^{-4}$ . Such  
 212 estimate was obtained by applying the error propagation law to Eq. 3, considering that uncertainty in the  
 213 particle density could not be quantified based on the only information available, that of the water density was  
 214 found negligible given the slight influence on temperature, and, finally, uncertainties in  $M_{\text{sample}}$  and  $W_{\text{sample}}$   
 215 were of the order of  $\pm 0.03\%$  and  $\pm 0.1\%$ , respectively. A second source of uncertainty in  $C$  was due to the  
 216 fact that the amount of solids carried along with the flow varied with time because particle accumulation and  
 217 resuspension could not be completely avoided. Following the sampling procedure illustrated in Section 2.2,  
 218 the concentration was determined approximately every two-to-three minutes during each test, showing a  
 219 variability with respect to the time-averaged values lower than  $\pm 7 \cdot 10^{-4}$ . Finally, the representativeness of  
 220 the sampling point should be taken into account. This aspect is not simple to investigate from a quantitative  
 221 point of view, but, since the solids are unlikely to deposit in the test cabin due to the high slope of its bottom  
 222 wall, the solid concentration at the sampling point S in Fig. 2 should not be too different from the value at  
 223 the nozzle outlet. Based on the above considerations, it seems reasonable to expect, for each experimental  
 224 test, an uncertainty of  $\pm 1 \cdot 10^{-3}$  on the solid volume fraction. This value is in agreement with the **estimate**  
 225 of Mansouri et al. [21] in previous wet direct impact tests. Note that, as the testing time increases, the  
 226 calculated average concentration value appears more and more reliable, as the higher number of samples  
 227 reduces the fluctuations in the solid volume fraction.

228 The overall measurement accuracy of the specimen mass was calculated considering the repeatability  
 229 (three times the standard deviation), which was 3.5 mg. Before each measurement, the specimen was  
 230 cleaned with an alcohol wet cloth. The much larger value of repeatability (3.5 mg) compared to the accuracy  
 231 of the balance (0.3 mg) was probably due to powder or micro-exfoliation when handling the specimens.  
 232 **The accuracy of the erosion depth measurements was estimated as the product between the linearity and the**  
 233 **measuring range of the profilometer, that is,  $\pm 0.2\% \cdot 50 \text{ mm} = 0.1 \text{ mm}$ .**

234 Finally, other sources of uncertainty are related with the properties of the solid particles. As already  
 235 mentioned, the particle density was not measured but reference was made to the value indicated by the  
 236 retailer. Nonetheless, application of the error propagation law showed that an error even up to  $100 \text{ kg/m}^3$

237 does not significantly affect the error on solid volume fraction, which is mostly influenced by the error on  
 238 the measurement of the sampling volume. Before running the tests, a rough sieve analysis was performed  
 239 to determine the particle size distribution of the coarsest sand used in the experiments, which was found in  
 240 reasonable agreement with the data provided by the retailer. Since it is well known that, in closed setups  
 241 involving a recycled use of the solids, the size and shape of the particles may vary with time due to particle  
 242 breaking and rounding [22], the sieve analysis was repeated after about 3 and 6 hours of operation of the  
 243 loop, suggesting a decrease in the mean value,  $d_{p,50}$ , as follows:  $d_{p,50}=0.371$  mm (new sand),  $d_{p,50}=0.348$   
 244 mm (after 3 hours and 25 minutes of operation),  $d_{p,50}=0.324$  mm (after 6 hours and 30 minutes of operation).  
 245 Qualitative comparison of microscope images of the new and the used sand grains indicated no significant  
 246 variation in grain shape. As it will be discussed later, the progressive reduction of particle size was subject  
 247 of concern when numerically simulating the experiments. **The key elements of the uncertainty analysis**  
 248 **described in this section are summarized in Table 1.**

[Table 1 about here.]

#### 250 2.4. Materials and testing conditions

251 Two different combinations of target specimen and abrasive particles were investigated. Firstly, prelim-  
 252 inary tests were run on samples made of aluminum alloy EN AW-6082, whose density was assumed as  
 253  $2700 \text{ kg/m}^3$  based on indications received from the retailer. The specimens were prisms with rectangular  
 254 cross-section and side lengths of 80 mm x 60 mm x 5 mm. In these tests, the abradant was a coarse silica  
 255 sand with the following properties, as obtained from the data provided by the retailer: particle density  $2500$   
 256  $\text{kg/m}^3$ ,  $d_{p,10}=0.25$  mm,  $d_{p,50}=0.35$  mm, and  $d_{p,90}=0.46$  mm.

257 Other experiments were conducted on samples made of a Glass Reinforced Epoxy (GRE) composite.  
 258 The specimens were obtained by cutting a circular pipe and, therefore, they have a surface curvature, which  
 259 is likely to affect the fluid dynamic behavior of the particle-laden flow. Their dimensions are indicated in  
 260 Fig. 3a. In order to be able to install the curved samples into the test cabin, a special holder was designed  
 261 and manufactured, as shown in Fig. 3b. The density of the GRE was  $2050 \pm 50 \text{ kg/m}^3$ , as estimated from  
 262 measuring the weight and the volume of the samples. In the GRE tests, use was made of fine silica sand  
 263 with density  $2500 \text{ kg/m}^3$ , and size distribution defined by  $d_{p,10}=0.075$  mm,  $d_{p,50}=0.125$  mm, and  $d_{p,90}=0.2$   
 264 mm. As for the coarse sand, also the particle properties of the fine sand reported above were obtained from  
 265 the datasheets provided by the retailer.

[Figure 3 about here.]

267 The experiments conducted on the EN AW-6082 specimens referred to a single testing condition, namely,  
 268  $V_{\text{jet}}=35.53 \pm 0.71 \text{ m/s}$ ,  $\theta_{\text{jet}} = 90 \pm 1^\circ$ ,  $C = 0.0085 \pm 0.001$ , and  $T = 5$  min. In this experiment, the abradant was  
 269 the coarse silica sand described above. Conversely, the GRE experiments was carried out using the fine sand.  
 270 The jet velocity was around 35 m/s, the nozzle-to-specimen angle was  $90 \pm 1^\circ$ , the solid volume fraction  
 271 was in the range  $0.0076 \div 0.0085$ , and the duration of the tests varied from 5 to 30 minutes. Repeatability  
 272 tests were conducted for some of the tested conditions, yielding, in all cases, successful results. **Relevant**  
 273 **details concerning the testing conditions are summarized in Table 2.**

[Table 2 about here.]

### 276 3. CFD modelling of the particle-laden flow

#### 277 3.1. Eulerian-Lagrangian equations

279 Owing to the relatively low solid volume fraction, and on the grounds of the findings obtained in a pre-  
 280 vious study [1], one-way coupling was assumed between the phases, thereby neglecting the influence of  
 281 the particles on the carrier fluid and any type of particle-particle interactions. As a result, it was possible  
 282 to decouple the solution of the carrier fluid flow and the calculation of the parcels' trajectories, with con-  
 283 siderable reduction of the simulation time. Firstly, the single-phase formulation of the Reynolds Averaged  
 284 Navier-Stokes equations was solved to determine the time-averaged pressure,  $P$ , and the velocity vector,  $U$ ,  
 285 of the carrier liquid [23]. These equations were coupled with the built-in Reynolds Stress Model (RSM) in  
 286 Ansys Fluent with default settings [24], which is expected to be more accurate than eddy viscosity based  
 287 turbulence models in reproducing the anisotropic turbulent flow occurring when a jet impinges a surface.

288 The scalable wall function option available in Ansys Fluent was employed to model the turbulence charac-  
 289 teristics in the near-wall cells. Secondly, the trajectories of the parcels were determined one after the other  
 290 by solving the following Lagrangian equation of motion for each parcel

$$m_p \frac{d\mathbf{v}_p}{dt} = \mathbf{F}_d + \mathbf{F}_l + \mathbf{F}_p + \mathbf{F}_{vm} \quad (4)$$

291 where  $m_p$  is the mass of a particle, and the four terms on the right hand side indicate the forces acting on a  
 292 parcel, which are drag force, lift force, pressure gradient force, and virtual mass force, respectively. Other  
 293 forces, such as gravity, buoyancy, and history force, were ignored because they are expected to play a minor  
 294 role in the case studies considered. The four forces in Eq. 4 were calculated as

$$\mathbf{F}_d = \frac{1}{2} \rho_l C_d \left( \pi \frac{d_p^2}{4} \right) |\mathbf{u}_{@P} - \mathbf{v}_p| (\mathbf{u}_{@P} - \mathbf{v}_p) \quad (5)$$

$$\mathbf{F}_l = 3.0844 J^* \frac{m_p}{\rho_p d_p} \sqrt{\frac{\nu_l}{|\Omega_{l,@P}|}} \cdot [\Omega_{l,@P} \times (\mathbf{U}_{@P} - \mathbf{v}_p)] \quad (6)$$

$$\mathbf{F}_p = -\frac{m_p}{\rho_p} \nabla P_{@P} \quad (7)$$

$$\mathbf{F}_{vm} = C_{vm} \rho_l \frac{m_p}{\rho_p} \left( \frac{d\mathbf{v}_p}{dt} - \frac{d\mathbf{U}_{@P}}{dt} \right) \quad (8)$$

295 where  $t$  is the Lagrangian integration time,  $\mathbf{v}_p$  is the instantaneous parcel velocity,  $d_p$  is the volume-  
 296 equivalent particle diameter, that is, the diameter of a sphere having the same volume of the actual particle,  
 297  $\mathbf{u}$  is the instantaneous velocity vector of the liquid,  $\nu_l$  is the kinematic viscosity of the liquid,  $\Omega_l = \nabla \times \mathbf{U}$   
 298 is the mean vorticity of the liquid,  $C_d$ ,  $J^*$ , and  $C_{vm}$  are dimensionless coefficients, and the subscript @P  
 299 means that  $\mathbf{u}$ ,  $\mathbf{U}$ ,  $\Omega_l$ , and  $\nabla P$  are interpolated at the current parcel position. The drag coefficient,  $C_d$ , is  
 300 obtained from the empirical correlation of Haider and Levenspiel [25], which expresses  $C_d$  as function of  
 301 the local mean parcel Reynolds number,  $Re_p = |\mathbf{U}_{@P} - \mathbf{v}_p| d_p / \nu_l$ , and the particle spherical coefficient,  $\varphi$ ,  
 302 which accounts for the effect of particle shape and it is defined as the ratio between the surface area of  
 303 the volume-equivalent sphere and the actual surface area of the particle. Recommended values for  $\varphi$  are  
 304 0.66, 0.76, and 0.86 for highly sharp, nearly rounded, and fully rounded grains [1], whereas a unit value  
 305 applies to the ideal case of perfectly spherical particles. The coefficient  $J^*$  in Eq. 6 was quantified through  
 306 an empirical correlation provided by Mei [26], which arises as an extension of the well known Saffman's  
 307 formula to finite particle Reynolds number and was implemented in the Ansys Fluent code through a User  
 308 Defined Function. Finally, the virtual mass coefficient,  $C_{vm}$ , was set to 0.5, as commonly assumed in the  
 309 literature.

310 Eq. 4 was solved in conjunction with another ordinary differential equation for the parcel position vector,  
 311  $\mathbf{r}_p$ , as follows:

$$\frac{d\mathbf{r}_p}{dt} = \mathbf{v}_p \quad (9)$$

312 and the numerical integration of the two equations yielded, for a discrete set of Lagrangian time instants,  
 313 the parcel position vector and the instantaneous parcel velocity vector. The instantaneous fluid velocity  
 314 vector,  $\mathbf{u}$ , which appears in Eq. 5, was evaluated from the RANS solution using a discrete random walk  
 315 stochastic model embedded in the Ansys Fluent code [24], leaving all default settings. **Relevant details on**  
 316 **the settings of the Eulerian-Lagrangian model are summarized in Table 3.**

317  
 318 [Table 3 about here.]

319

### 320 3.2. Computational domain and boundary conditions

321 A sketch of the computational domain, including the boundary conditions, is shown in Fig. 4 for  
 322 the curved GRE specimen case. The domain includes the nozzle with its actual thickness, modeled  
 323 as a straight pipe with length equal to 10 nozzle inner diameters, the specimen with its clamps, and  
 324 the surrounding environment (Fig. 4a). **A similar setup was defined for the flat aluminium case, where**  
 325 **the target wall was a plane circle surrounding the rectangular specimen (as in Fig. 4a of the earlier paper [1]).**

326  
 327 [Figure 4 about here.]

328 Boundary conditions were inlet, outlet, and solid walls. At the inlet, a fully-developed mean axial veloc-  
 329 ity profile was specified [27], its mean value being  $V_{\text{jet}}$ , whereas the two other mean velocity components  
 330 and all turbulence parameters were set as zero. In order to define the parcels' initial conditions, the mass  
 331 flow rate associated to each parcel was calculated first. Particularly, all parcels were attributed the same  
 332 mass flow rate,  $\dot{m}_p$ , determined in such a way to produce the desired solid volume fraction at the nozzle  
 333 exit, that is,

$$\dot{m}_p = \frac{\rho_p C V_{\text{jet}} \pi \frac{d^2}{4}}{N_p} \quad (10)$$

334 where  $d$  is the inner diameter of the nozzle, and  $N_p$  is the total number of tracked parcels. The inlet boundary  
 335 was triangulated, and the averaged fluid velocity and liquid mass flow rate were calculated for each triangle.  
 336 The number of injected parcels from each triangle was proportional to the local liquid mass flow rate, their  
 337 initial positions were extracted randomly from a uniform probability distribution over the triangle, and their  
 338 initial velocities were equal to the local averaged fluid velocity. In Ansys Fluent, the particle size associated  
 339 to each parcel must also be specified at the inlet section. As it will be discussed later, this was achieved  
 340 by sampling the values from a log normal probability distribution. At the outlet, the mean gauge pressure  
 341 of the fluid and the normal gradient of all other fluid-related variables were specified as zero, whereas the  
 342 parcels simply leave the domain. At the walls, the fluid velocity is zero and, as already mentioned, the  
 343 scalable wall function option is employed to model the turbulence characteristics in the near-wall cells.  
 344 The parcels rebound when they hit a solid wall. The normal and tangential parcel velocity components  
 345 after impingement are related with the corresponding incident values through two restitution coefficients,  
 346 which are expressed as polynomial functions of the parcel impact angle through the empirical correlations  
 347 of Forder et al. [28]. As it will be detailed later in Section 4.2, following an earlier study of two of the  
 348 authors of this paper [29], the parcel-wall impingement characteristics (velocity, angle, and position) were  
 349 not evaluated at a distance from the wall equal to half particle size, as an approximate way to take the  
 350 finite particle size into account in the point-particle framework. This was achieved through a User Defined  
 351 Function.

352

### 353 3.3. CFD solution strategy

354 The numerical simulations were performed using the commercial code Ansys Fluent version 19.2,  
 355 complemented with User Defined Functions. As already mentioned, owing to the one-way coupling  
 356 regime assumption, the single-phase flow of the carrier liquid was calculated first by solving the RANS  
 357 equations in the Eulerian, cell-based formulation. Unstructured tetrahedral meshes was initially created,  
 358 and then converted into polyhedral meshes in Ansys Fluent for speeding up the simulation time: as a  
 359 rule of thumb, in fact, the number of cells in a polyhedral mesh can be close to one fifth of that of the  
 360 original tetrahedral mesh. Fig. 5 shows the spatial discretization on a restricted view of the mid plane  
 361 in the case of the curved GRE specimen. The sketch clearly reveals that the cells are densified in the  
 362 core region of the domain, which is more prone to particle-wall impingements. Additionally, an inflation  
 363 layer is added adjacent to the surface of the specimen and the inner surface of the nozzle, based on a first  
 364 layer thickness of 0.05 mm. An extensive mesh sensitivity study was performed for the calibration case  
 365 presented in Section 5.2, referring to the normal jet impingement on a GRE specimen with  $V_{\text{jet}} = 35.53$  m/s,  
 366  $C = 0.0085$ , and  $T = 5$  min, the target parameter being the local mean fluid velocity profile along a  
 367 line near the sample surface. The solutions calculated on meshes with 0.18, 0.41, 0.72, 1.45, and 2.11  
 368 million elements were compared, and substantially grid-independent results were obtained with the 1.45  
 369 million cells. Starting from the results of this test case, the mesh sensitivity analysis was repeated for  
 370 all other computation domains, finding that, in the flat aluminum specimen case, about 1.26 million cells  
 371 were enough to attain mesh independence. Note that, since the parcel-wall impingement characteristics  
 372 were not evaluated at the wall but at a distance upstream of it equal to half particle size (Section 4.2),  
 373 they were only minorly affected by the near-wall mesh resolution and inflation layer properties. For  
 374 the numerical solution of the RANS equations, a coupled pressure-velocity scheme was used, with  
 375 Green-Gauss Node-Based gradient discretization, PRESTO! pressure discretization, and Second Order  
 376 Upwind schemes applied to all other solved variables. The convergence criterion was that the normalized  
 377 unscaled residuals must drop to below  $10^{-4}$  and, additionally, the area-averaged wall skin friction factor of  
 378 the specimen was verified to become constant. The under-relaxation factors were left to their default values.

379

[Figure 5 about here.]

381 Once the carrier fluid flow field was calculated, the parcel trajectories were determined sequentially  
 382 through the integration of the Lagrangian equations of motion. This was achieved through the DPM model  
 383 available in Ansys Fluent in the steady-state particle tracking mode, without accounting for the effect of  
 384 the parcels on the continuous phase. The default numerical settings of the DPM were used. The number  
 385 of tracked parcels was 50000, high enough to attain significant parcel-wall impingement statistics. A brief  
 386 sensitivity analysis was made for the flat specimen case discussed in Section 5.1, which indicated that even  
 387 doubling the number of trajectories did not produce any significant change to the curve fitting coefficients.  
 388 **Table 4 provides the key details on the numerical settings of the CFD model.**

389 [Table 4 about here.]

390

#### 391 4. Methodology for erosion model calibration

##### 392 4.1. The *original methodology of Mansouri [19]*

393 The methodology proposed by Mansouri allows calibrating an empirical erosion model having the fol-  
 394 lowing form

$$\dot{E}_{p,imp} = \dot{m}_p F_s K \cdot v_{p,imp}^n f(\theta_{p,imp}) \quad (11)$$

395 by combining the experimental measurements and the simulation results of a wet direct impact test (Fig. 1b).  
 396 In Eq. 11,  $F_s$  is a dimensionless coefficient related with particle shape, equal to 0.2, 0.53, and 1.0 for fully  
 397 rounded, nearly rounded, and fully sharp grains, respectively,  $K$  is a material-related constants, which  
 398 includes also the influence of particle size,  $n$  is a velocity exponent, and  $f(\theta_{p,imp})$  is a two-parameters  
 399 impact angle function, as follows,

$$f(\theta_{p,imp}) = (\sin \theta_{p,imp})^{b1} \cdot (1.5 - \sin \theta_{p,imp})^{b2} \quad (12)$$

400 The idea of Mansouri, as presented in his PhD thesis [19], is to divide the target wall into  $N_{cells}$  surface  
 401 cells and apply Eq. 11 is applied at the scale of each cell. This yields

$$\dot{E}_i = \dot{M}_i F_s K \cdot \langle v_{p,imp} \rangle_i^n f(\langle \theta_{p,imp} \rangle_i) \quad (13)$$

402 where the subscript  $i$  means that the following quantities refer to the generic cell:  $\dot{E}_i$  is the total mass  
 403 flow rate of removed material,  $\dot{M}_i$  is the impinging mass flow rate, equal to the product of the number of  
 404 impingements and the mass flow rate associated to each parcel,  $\langle v_{p,imp} \rangle_i$  is the average of the parcel impact  
 405 velocities, and  $\langle \theta_{p,imp} \rangle_i$  is the average of the parcel impact angles. After dividing by the product of the target  
 406 material density,  $\rho_t$ , and the area of the cell,  $\Delta A_i$ , and multiplying by the testing time,  $T$ , the area-averaged  
 407 erosion depth of each cell,  $\langle h \rangle_i$ , can be obtained as

$$\langle h \rangle_i = \frac{\dot{E}_i T}{\rho_t \Delta A_i} = \frac{\dot{M}_i F_s K \cdot \langle v_{p,imp} \rangle_i^n f(\langle \theta_{p,imp} \rangle_i)}{\rho_t \Delta A_i} \cdot T \quad (14)$$

408 and, finally, Eq. 14 can be rearranged as

$$K \cdot f(\langle \theta_{p,imp} \rangle_i) = \frac{\langle h \rangle_i}{T \cdot F_s \cdot \langle v_{p,imp} \rangle_i^n} \cdot \frac{\rho_t \Delta A_i}{\dot{M}_i} \quad (15)$$

409 For each cell  $i$ , the erosion depth  $\langle h \rangle_i$  can be measured using a profilometer. Conversely,  $\dot{M}_i$ ,  $\langle v_{p,imp} \rangle_i$ ,  
 410 and  $\langle \theta_{p,imp} \rangle_i$  are difficult to evaluate experimentally, but they can be easily estimated from an Eulerian-  
 411 Lagrangian CFD simulation. By combining the numerical and experimental results, a list of  $N_{cells}$  pairs of  
 412 coordinates  $[\langle \theta_{p,imp} \rangle_i, K \cdot f(\langle \theta_{p,imp} \rangle_i)]$  can be determined, and the coefficients  $K$ ,  $b1$ , and  $b2$  can be obtained  
 413 by applying some regression procedure. This requires the velocity exponent  $n$  to be known, but this aspect  
 414 has not completely clarified by Mansouri et al. [18], who simply assumed  $n = 2.4$  based on previous  
 415 experimental findings and did not provide any criteria for deciding the value of this parameter. According to  
 416 the authors, this aspect deserves more attention but, unfortunately, the experimental data currently available  
 417 did not allow gaining more insight. Therefore, this task was shelved for future research, and, in this study,  
 418 the value of  $n$  was left to 2.4. **A flowchart of the original methodology of Mansouri [19] is shown in Fig. 6.**

419 [Figure 6 about here.]

420

421 *4.2. An improved methodology*

422 Three main modifications were made to the methodology of Mansouri [19] in order to improve the  
423 accuracy of erosion model calibration and widening its scope.

424 Firstly, the particle shape coefficient  $F_s$  is included in the calibration constant  $K$ , and, therefore, the  
425 formulations of the erosion model applied at the single impact scale and at the cell scale are, respectively,

$$\dot{E}_{p,imp} = \dot{m}_p K \cdot v_{p,imp}^n f(\theta_{p,imp}) \quad (16)$$

426 and

$$\dot{E}_i = \dot{M}_i K \cdot \langle v_{p,imp} \rangle_i^n f(\langle \theta_{p,imp} \rangle_i) \quad (17)$$

427 Although this change might appear purely formal, it is indeed important for several inter-related reasons, as  
428 follows. When combining the current Eulerian-Lagrangian model with the original erosion model formu-  
429 lation of Mansouri, particle shape is accounted for via two parameters. The former is the particle spherical  
430 coefficient,  $\varphi$ . This parameter appears in the drag coefficient correlation of Haider and Levenspiel [25] to  
431 quantify the influence of particle shape on the resistance encountered by the particles travelling in the carrier  
432 fluid, and, therefore, it affects the parcel trajectories and the parcel-wall impingement characteristics. The  
433 latter parameter is the coefficient  $F_s$  in the erosion model (Eq. 11), which quantifies the effect of particle  
434 shape on the mass removal produced by a single impinging particle, for the same impact velocity and angle.  
435 As already said, recommended values of  $\varphi$  and  $F_s$  for fully sharp, nearly rounded, and fully rounded grains  
436 are (0.66, 0.76, 0.86) and (1.0, 0.53, 0.2), respectively. However, these parameters should not be treated  
437 as independent with each other, as they are associated with the same geometrical property of the grains.  
438 Avoiding the explicit appearance of  $F_s$  stresses this aspect.

439 In addition, particle shape is difficult to quantify for natural sands, and, therefore,  $\varphi$  (and  $F_s$ ) are critical  
440 features of a CFD-based erosion prediction model. An earlier study by Messa and Malavasi [1] showed that,  
441 when simulating a wet direct impingement test, the value of  $\varphi$  has a significant influence on the particle-wall  
442 impingement characteristics. Based on the above considerations, it is here proposed to use the consistency  
443 between the map of the particle-wall collisions, obtained from the CFD simulation, and the shape of the  
444 erosion hole found in the experimental test as a criterion to decide the value of  $\varphi$  within the triplet (0.66,  
445 0.76, 0.86).

446 The last reason why it was decided to include  $F_s$  in the calibration constant  $K$  is to make the applicability  
447 limits of the calibrated model evident. In fact, leaving  $F_s$  at the denominator of Eq. 15, as Mansouri  
448 et al. [18] did, implies that the calibration constants  $K$ ,  $b1$ , and  $b2$  would be valid for different particle  
449 shapes, since the effect of this parameter could be accounted for a posteriori by changing  $F_s$ . Instead, it  
450 is here recommended to employ the calibrated formula only for cases involving the same type of abrasive  
451 particles, also because particle shape, particle size and particle material are usually strongly interrelated,  
452 and, therefore, it seems reasonable to include all these parameters into  $K$ ,  $b1$ , and  $b2$ .

453 The second modification made to Mansouri's strategy is to change, in the analogous of Eq. 15 for the  
454 new formulation, the term related with the impact velocity from  $\langle v_{p,imp} \rangle_i^n$  to  $\langle v_{p,imp}^n \rangle_i$ . The rationale behind  
455 this change is to make the calibration of the model more consistent with the post-processing of the parcel-  
456 wall impingement characteristics through the E-CODE, which is the in-house MATLAB library for erosion  
457 calculation developed with the authors' research group. In the E-CODE, each impact point is associated to  
458 the centroid of the nearest surface cell, and the erosion model in the form of Eq. 11 is applied to find the  
459 mass flow rate of removed material caused by the current impingement. The sum over all impingements  
460 associated to the same cell yields the mass flow rate of removed material of that cell,  $\dot{E}_i$ . Note that, for the  
461 sake of clarity, the generic cell is denoted once again by the subscript  $i$ , since the surface discretization in  
462 the E-CODE is the same considered when illustrating the calibration methodology in Section 4.1; therefore,  
463 the number of impingements occurring in that cell is referred to as  $N_{imp,i}$ . Since each cell is attributed a  
464 single erosion depth, this can be interpreted as the area-averaged value over the cell. Using Eq. 16, the  
465 area-averaged erosion depth of cell  $i$  after a time  $T$ , called  $\langle h \rangle_i$ , can be obtained as:

$$\langle h \rangle_i = \frac{\dot{E}_i}{\rho_t \Delta A_i} T = \frac{T}{\rho_t \Delta A_i} \cdot \left[ \sum_{p=1}^{N_{imp,i}} \dot{m}_p K v_{p,imp}^n f(\theta_{p,imp}) \right] \quad (18)$$

466 and the equation above can be rearranged as:

$$\langle h \rangle_i = \frac{\dot{M}_i K \langle v_{p,imp}^n f(\theta_{p,imp}) \rangle_i}{\rho_t \Delta A_i} \cdot T \quad (19)$$

467 where the impinging mass flow rate over the cell is calculated as  $N_{imp,i}\dot{m}_p$ . Clearly, it is not possible to turn  
 468 Eq. 19 into an equation for the product  $K \cdot f(\langle\theta_{p,imp}\rangle_i)$ , since the area-averaging operator is applied to the  
 469 term  $v_{p,imp}^n f(\theta_{p,imp})$ . However, if the following assumption is made:

$$\langle v_{p,imp}^n f(\theta_{p,imp}) \rangle_i \approx \langle v_{p,imp}^n \rangle \cdot f(\langle\theta_{p,imp}\rangle_i) \quad (20)$$

470 then, an explicit expression for  $K \cdot f(\langle\theta_{p,imp}\rangle_i)$  can be found, as follows,

$$K \cdot f(\langle\theta_{p,imp}\rangle_i) = \frac{\langle h \rangle_i}{T \cdot \langle v_{p,imp}^n \rangle_i} \cdot \frac{\rho_t \Delta A_i}{M_i} \quad (21)$$

471 and the formula here above is used instead of Eq. 15 to find the calibration constants  $K$ ,  $b1$ , and  $b2$ . Indeed,  
 472 Eq. 20 does not have a clear theoretical justification, but it represents a practical way to make the area-  
 473 averaged approach of Mansouri's procedure consistent with the parcel-based approach at the basis of the  
 474 E-CODE. In order to obtain the original formulation of Mansouri, a second assumption should be made  
 475 alongside with Eq. 20, that is,  $\langle v_{p,imp}^n \rangle_i \approx \langle v_{p,imp} \rangle_i^n$ . However, not only this additional approximation would  
 476 be further unjustified from a theoretical point of view, but it would also have an adverse effect on the  
 477 calibration accuracy, as it will be shown later in Section 5.2. Finally, for the sake of completeness, it is  
 478 mentioned that, in the E-CODE, the overall mass loss after a given time  $T$  is obtained as

$$\Delta M = \sum_i \dot{E}_i T \quad (22)$$

479 where the summation is over all surface cells.

480 The last difference between the current investigation and the works by Mansouri [18, 19] is not related  
 481 with the calibration procedure in itself, but, rather, in the evaluation of the parcel-wall impingement char-  
 482 acteristics from the output of the Eulerian-Lagrangian CFD simulation.

483 Eulerian-Lagrangian models for engineering calculations, such as the one employed in this study, rely  
 484 on the already mentioned point-particle approximation, in which the parcels are treated as point sources of  
 485 momentum with zero dimension. As a result, the integration of the parcel trajectories is performed up to  
 486 the wall boundary and the impingement characteristics (that is, position of the impact, impact velocity, and  
 487 impact angle) are evaluated at the wall. As discussed by two authors of this paper in a previous study [29],  
 488 such computational procedure seriously affects the reliability of the CFD-based erosion prediction model  
 489 for two reasons. On the one hand, it produces an underestimation of the impact velocity, because the high  
 490 drag experienced by the parcels in the near-wall region, where the fluid velocity is very low owing to the  
 491 presence of the boundary layer, is a mere consequence of ignoring the actual size of the solids. On the other  
 492 hand, it makes the particle-wall impingement characteristics strongly dependent on the mesh resolution  
 493 close to the wall as well as on the near-wall treatment approach used. Since a fully-resolved treatment of  
 494 the solid phase would result in prohibitive computational cost, two practical strategies were proposed by  
 495 researchers.

496 Zhang et al. [30] recommended to use different mesh sizes and different near-wall treatment methods  
 497 according to the size of the particles. Undoubtedly, this approach is very easy to implement, but it suffers  
 498 from lack of applicability when considering particle-laden flows with broad grain size distributions.  
 499 Conversely, Messa and Wang [29] proposed to evaluate the parcel-wall impingement characteristics at the  
 500 particle centroid, that is, at a distance from the wall equal to half particle diameter (Fig. 7). Note that this  
 501 alternative approach does not impose the particle rebound to occur at the particle centroid, but it extracts  
 502 from a point-particle trajectory the parcel position, velocity and angle half particle diameter away from  
 503 the wall, and take these values as the parcel-wall impingement characteristics to be used in the erosion  
 504 calculations. The method by Messa and Wang [29], employed in this study, has a clear physical meaning,  
 505 and it is applicable to either mono-dispersed or poly-dispersed particle-laden flows. Additionally, it makes  
 506 the parcel-wall impingement characteristics (and, therefore, the erosion estimates) less sensitive to the  
 507 near-wall mesh resolution and wall treatment method compared to the traditional point-particle method. At  
 508 the same time, its implementation in the Ansys Fluent code required writing a library of UDFs.

509  
510

[Figure 7 about here.]

511 The flowchart of the improved methodology is shown in Fig. 8. Note that the upgraded procedure allows  
 512 overcoming some limitations of CFD-based erosion prediction highlighted by Parsi et al. [31], namely, the  
 513 problems with commercial particle tracking subroutines which rely on the point-particle approximation, the  
 514 strong influence of the mesh on the erosion predictions, and the uncertainty related with the value of the  
 515 particle spherical coefficient.

[Figure 8 about here.]

516

517

#### 518 4.3. A formulation for statistically axi-symmetric flows

519 The first case study considered in this paper is the normal impingement of a slurry jet on a flat  
 520 specimen, in which the particle-laden flow is statistically axi-symmetric, and, therefore, also the erosion  
 521 hole is axi-symmetric as well. When addressing this test case, use was made of a formulation of the  
 522 calibration methodology which is valid only for statistically axi-symmetric flows, in which the parcel-wall  
 523 impingement characteristics and the local erosion depth depend only on a radial coordinate measured from  
 524 the center of the specimen. In the axi-symmetric formulation, the surface of the specimen is no longer  
 525 subdivided in square cells but, rather, into  $N_{\text{rings}}$  concentric rings with radius  $r_i$  and thickness  $\Delta r_i$  (Fig. 9).

526

[Figure 9 about here.]

527

528 When the upgraded methodology, illustrated in Section 4.2, is applied at the ring scale, Eq. 21 becomes:

$$K \cdot f(\langle \theta_{p,\text{imp}} \rangle_{\text{ring},i}) = \frac{\langle h \rangle_{\text{ring},i}}{T \cdot \langle v_{p,\text{imp}}^n \rangle_{\text{ring},i}} \cdot \frac{\rho_t \Delta A_{\text{ring},i}}{M_{\text{ring},i}} \quad (23)$$

529 where the new subscript “ring,  $i$ ” means that all quantities refer to the generic ring with center radius  $r_i$ .  
 530 Provided that  $\Delta r_i$  is sufficiently small, the ring area  $\Delta A_{\text{ring},i}$  could be estimated as  $2r_i\pi\Delta r_i$ .

531 Applying Eq.23 requires evaluating the ring-based values of an experimentally-determined variable, that  
 532 is, the ring-averaged erosion depth, as well as of three parameters obtained from the CFD simulation,  
 533 namely, the ring-averaged parcel impact velocity elevated to power  $n$ , the ring-averaged parcel impact  
 534 angle, and the total impinging mass flow rate in each ring. Using the E-CODE, evaluating the parcel wall  
 535 impingement characteristics at the scale of the ring simply required implementing in the MATLAB subrou-  
 536 tine a proper discretization of the specimen surface. Conversely, some interpolation of the experimentally-  
 537 determined erosion depth values was necessary, since these data were acquired by the profilometer on a  
 538 cartesian grid of measuring points.

539 Obviously, the cell-based methodology might be directly applied also to statistically axi-symmetric  
 540 flows. The reason behind the authors’ choice to develop a case-specific, ring-based formulation is twofold.  
 541 On the one hand, expressing the statistics of the parcel-wall impingements as well as the erosion depth as  
 542 a function of a single space coordinate, that is, the radial one, appears the most straightforward choice and  
 543 makes the results more easily accessible and interpretable. On the other hand, it is easier to get statistically  
 544 significant statistics on a large ring rather than on a small elementary square cell. That is, taking advan-  
 545 tage of the known axi-symmetric behavior of the mean flow, it is possible to reduce the number of tracked  
 546 parcels and, therefore, the computational burden of CFD simulation and data processing.

## 547 5. Results and discussion

548 The experimental and numerical results obtained in the present study are reported and discussed hereafter.  
 549 Particularly, in Section 5.1, the calibration of the erosion correlation for a normal jet impingement against  
 550 a flat aluminum specimen is described, as performed adopting the **ring-based formulation presented in**  
 551 **Section 4.3**. Section 5.2 illustrates the development of an erosion correlation from the case of normal  
 552 jet impingement against a curved GRE specimen. Since the phenomenon is no longer axisymmetric, use  
 553 was made of the **more general cell-based approach**. **Although** this study almost entirely focused on the  
 554 calibration of the erosion model, examples of application of the calibrated formula to GRE experiments  
 555 with longer testing times are presented and discussed.

556 Before illustrating the results, it is noted that, for different reasons, both the experimentally-measured and  
 557 the numerically-obtained erosion depth distributions had to be filtered **a posteriori** to smooth **out noise**. In  
 558 the experimental data, such **noise** mainly arose from measurement errors and/or lack of complete control of  
 559 the experimental conditions, whereas, in the numerical solution, **it was a consequence of the discrete nature**  
 560 **of particle tracking in Eulerian-Lagrangian models**, in addition to the fact that, as already explained in  
 561 **Section 4.2**, the E-CODE calculates the erosion of each surface cells considering only the impacts occurring  
 562 **in that cell, and it does not perform weight-averaging of the impacts in neighboring cells as well**. After  
 563 testing different types of smoothing filters implemented in MATLAB, it was decided to employ a nine-  
 564 point moving average filter. Another aspect to consider when processing the experimental data was that  
 565 the point of intersection between the nozzle axis and the surface of the specimen, virtually corresponding  
 566 to the center of the erosion hole in the normal jet cases, had to be determined. However, for all test cases,  
 567 the location of the point could be inferred from the shape of the smoothed erosion depth distribution with  
 568 reasonable accuracy.

569 *5.1. Test case 1: flat aluminum sample*

570 In the first test case, the normal impingement of a slurry jet against a flat aluminum specimen with  
 571 density  $2700 \text{ kg/m}^3$  was considered. The test parameters were  $V_{\text{jet}} \approx 35 \text{ m/s}$ ,  $C \approx 0.0085$ , and  $T = 5 \text{ min}$ .  
 572 An erosion correlation was calibrated for this specific flow condition, **employing the improved methodology**  
 573 **in the ring-based formulation**. The width of the rings used to average the experimental and the numerical  
 574 results was determined in such a way that 2500 parcel-wall impingements occurred in each ring, since this  
 575 criterion produced meaningful statistics and a sufficiently high profile resolution.

576 As already said, the density of the coarse sand particles was  $2500 \text{ kg/m}^3$ , whereas two difficult-to-  
 577 evaluate parameters were the grain size distribution and the shape of the grains. In this regard, it has  
 578 already been noticed that the particles tended to become progressively finer during their recycled use, re-  
 579 ducing their mean size of about 12% after 6 hours and 40 minutes of operation. In order to explore whether  
 580 this effect can adversely affect the accuracy of calibration, the simulations were repeated by sampling the  
 581 grain size of the injected parcels from four different particle size distributions, namely, the one provided by  
 582 the retailer and those obtained from the sieve analyses on the brand new sand, after 3 hours and 25 minutes,  
 583 and after 6 hours and 30 minutes. The results indicated that choosing one particle size distribution instead  
 584 of another had very little effect on the goodness-of-fit of the  $K \cdot f(\langle \theta_{p,\text{imp}} \rangle_{\text{ring},i})$  function using Eq. 12. In-  
 585 spection of the parcel-wall impingement characteristics revealed that such variations in particle size are still  
 586 too mild to produce significant changes in the behavior of the particle-laden flow. The results presented in  
 587 the remainder of this section refer to the particle size distribution after 3 hours and 25 minutes of operation,  
 588 having  $d_{p,50} = 0.348 \text{ mm}$ .

589 **Since the coefficient  $F_s$ , which appears explicitly in the erosion model formulation considered by Man-**  
 590 **souri (Eq. 11), is here included in the calibration constant  $K$ , the CFD-based erosion prediction model**  
 591 **includes only one parameter related with particle shape. This is the particle shape coefficient in the drag**  
 592 **coefficient correlation of Haider and Levenspiel [25], recommended equal to 0.66, 0.76, and 0.86 for highly**  
 593 **sharp, semi-rounded, and highly rounded particles, respectively [1]. It is clear that reducing the complexity**  
 594 **of the shapes of a mixture of natural sand grains to a single scalar variable is a big challenge, if not im-**  
 595 **possible. Therefore,** it is not surprising to see that the value of  $\varphi$  is a significant source of uncertainty in  
 596 CFD-based erosion prediction and that, to some extent,  $\varphi$  might be treated as a calibration parameter. **As**  
 597 **already mentioned in Section 4.2, one of the key ideas of this study is to exploit the combination of CFD**  
 598 **and experimental results not only to calibrate an empirical erosion model, but also to gather indications on**  
 599 **the proper value of  $\varphi$  to use in the numerical simulations. This is exemplified hereafter.**

600 [Figure 10 about here.]

601 Figures 10a-c show the ring-based parcel-wall impingement characteristics for the three values of  $\varphi$ ,  
 602 whereas Fig. 10d depicts the experimentally determined ring-averaged erosion depth profile. Whilst the  
 603 impact velocity distribution is rather insensitive to the value of  $\varphi$  (Fig. 10a) and the impact angles show a  
 604 only slight increase with increasing  $\varphi$  (Fig. 10b), the number of impacts per ring area is strongly affected by  
 605 the particle spherical coefficient: in fact, as clearly shown in Fig. 10c, rounded particles are more prone to  
 606 impinge in the central part of the specimen, whilst sharp grains tend to impinge at a certain distance from  
 607 the axis of the jet (note that, in order to obtain the number of impacts in each ring, the curves in Fig. 10c should  
 608 be multiplied by the ring area, which increases with increasing ring radius). This finding is not surprising  
 609 because, compared to rounded grains, sharp particles experience higher drag force and, therefore, they have  
 610 a stronger tendency to follow the fluid streamlines, which spread away from the stagnation region.

611 [Figure 11 about here.]

612 The scatter plots in Figs. 11(a-c) show, for the three values of  $\varphi = 0.66, 0.76, 0.86$ , the ring-averaged  
 613 parcel impact angle,  $\langle \theta_{p,\text{imp}} \rangle_{\text{ring}}$ , obtained from the CFD simulations, versus the product  $K \cdot f(\langle \theta_{p,\text{imp}} \rangle_{\text{ring}})$ ,  
 614 calculated by combining the numerical and experimental results in Eq. 23. The regression curves are also  
 615 shown. As already evident from visual inspection of the plots, and qualitatively confirmed by the  $R^2$  values,  
 616 a good fit can be obtained only for  $\varphi = 0.66$  (Fig. 11a). Conversely, the trendline is a poor representation of  
 617 the scattered points for nearly rounded grains with  $\varphi = 0.86$  (Fig. 11c). The motivation of this behavior can  
 618 be found by combining the information reported in Fig. 10 with Eq. 23. When  $\varphi = 0.86$ , the preferential  
 619 location of the particle wall impingements, close to the center of the specimen, is not compatible with the  
 620 observed erosion depth profile. This is quite evident when the attention is turned to point P at  $r \approx 10$   
 621 mm, where the ring-averaged impact angle is around  $11^\circ$  (Fig. 10b). At this location, there is significant  
 622 erosion depth (Fig. 10d), but only a limited number of impingements per unit area (black curve in Fig. 10c).  
 623 When Eq. 23 is applied, such unphysically low value of  $M_{\text{ring}}/\Delta A_{\text{ring}}$  produces an unphysically high value of  
 624  $K \cdot f(\langle \theta_{p,\text{imp}} \rangle_{\text{ring}})$ , which is clearly visible in Fig. 11c. At the same time, if the point Q of maximum erosion

625 depth is considered, the high  $\dot{M}_{\text{ring}}/\Delta A_{\text{ring}}$  (black curve in Fig. 10c) yields low  $K \cdot f(\langle\theta_{\text{p,imp}}\rangle_{\text{ring}})$  (Fig. 11c).  
 626 As a result, the regression curve produces severe underestimation of the erosion depth in P, whereas the  
 627 erosion depth in Q is overestimated (black curve in Fig. 11d). The fitting is considerably improved when  $\varphi$   
 628 is reduced owing to the outward shift of the preferential location of the parcel-wall impingements (Fig. 10c).  
 629 When  $\varphi = 0.66$ , as characteristics of sharp grains, the maximum erosion depth is accurately captured by the  
 630 calibrated erosion model (red curve in Fig. 11d).

631 The considerations above indicate that the accuracy of calibration is strictly related with the value of the  
 632 particle spherical coefficient. Therefore, it is here recommended to repeat the numerical simulation of the  
 633 wet direct impact experiment used for calibrating the erosion model three times, with  $\varphi$  equal to 0.66, 0.76,  
 634 and 0.86, and chose the value of  $\varphi$  for which the predicted erosion depth profile is in closest agreement with  
 635 the experimental data, by referring, for instance, to the deviation in the maximum value. Apparently, also  
 636 the goodness-of-fit of the  $K \cdot f(\theta_{\text{p,imp}})$  regression model, quantified, for instance, by the  $R^2$  value, might  
 637 be taken as a criterion to decide the value of  $\varphi$ . However, it will be shown that  $R^2$  is no longer a suitable  
 638 indicator in the GRE tests and, therefore, referring to the maximum erosion depth appears a better choice.  
 639 Note that, in any case, the tuning of  $\varphi$  must not be intended in an absolute sense, since the type of grain  
 640 shape assumed in the simulations (highly sharp, semi-rounded, or highly rounded) must be consistent with  
 641 the particles used in the experiments, if any information in this sense is available.

642 As a final note, the mass reduction of the aluminum specimen obtained from the balance was 0.973 g.  
 643 Volume integration of the experimentally-determined erosion depth profile and subsequent multiplication by  
 644 the aluminum density yielded 0.744 g. **The scope of comparing the two types of measurements was simply  
 645 to provide some degree of confidence in the reliability of the experimental data. Indeed, the difference  
 646 between the two values is not negligible, and it was interpreted by considering that, as already discussed in  
 647 Section 2.3, the uncertainty of the erosion depth measurements is about  $\pm 0.1$  mm, which is not negligible  
 648 compared to the measured values in many cells (Figs. 10d). Therefore, the mass loss estimate obtained  
 649 from the profilometer measurement is expected to be less accurate than that of the balance.** The numerically  
 650 obtained mass losses, calculated through Eq. 22, were 0.827 g, 0.788 g, and 0.925 g, when  $\varphi$  was equal to  
 651 0.66, 0.76, and 0.86, respectively.

## 652 5.2. Test case 2: curved GRE samples

653 In the second test case, slurry jet impingements on the curved GRE samples were considered. The  
 654 calibration of the erosion correlation was made for the following testing conditions:  $V_{\text{jet}}=35.53$  m/s,  $C =$   
 655  $0.0085$ , and  $T=5$  min. The density of GRE and of the abrasive particles were  $2050$  kg/m<sup>3</sup> and  $2500$  kg/m<sup>3</sup>,  
 656 respectively. As far as the size distribution of the abrasive grains is concerned, the only information available  
 657 came from the tabular data provided by the retailer, previously detailed in Section 2.4, because, unlike for  
 658 the coarse sand used in the aluminum test, no sieve analysis was made on the fine sand particles of the GRE  
 659 experiments. Based on the results obtained for the coarse grains, which **suggested that moderate changes in  
 660 the particle size distribution are not likely to affect the accuracy of calibration**, the GRE simulations were  
 661 run by sampling the particle sizes from a log-normal distribution with mean value equal to 0.125 mm, built  
 662 from the data provided by the retailer. No information could be obtained about the shape of the fine sand  
 663 particles. **Therefore**, the particle spherical coefficient  $\varphi$  was treated as a calibration parameter, estimated by  
 664 combining the experimental and CFD results, as explained in the previous section.

665 [Figure 12 about here.]

666 The experimentally-determined erosion depth map over the inner surface of the GRE specimen is de-  
 667 picted in Fig. 12, where the local values have been normalized by the maximum erosion depth to preserve  
 668 confidentiality. Since the normal impingement against the curved GRE specimen is not an axisymmetric  
 669 phenomenon, averaging over concentric rings was no longer meaningful, and the **general, cell-based ap-  
 670 proach** had to be adopted. Particularly, since the erosion depth measurements were obtained spanning over  
 671 a Cartesian grid with resolution of 0.5 mm in both directions, the **surface cells were squares** with edge size  
 672 of 0.5 mm.

673 [Figure 13 about here.]

674 The simulations were repeated three times with  $\varphi$  equal to 0.66, 0.76, and 0.86 and the parcel-wall  
 675 impingement statistics were calculated over each **cell**. The results indicated that, as for the first test case,  
 676 the value of  $\varphi$  has a minor influence on the locally-averaged impact velocity and impact angle (Figs. 13a-f),  
 677 but it strongly affects the impact number density map (Figs. 13g-k). Particularly, the impacts are mainly  
 678 concentrated in the central part of the specimen when  $\varphi=0.86$ , whereas they tend to be more uniformly  
 679 distributed as  $\varphi$  decreases, due to the larger spreading of the trajectories around the stagnation region. It

680 is rather intuitive that, compared to those obtained for  $\varphi=0.66$  and  $0.76$ , the impact pattern obtained for  
 681  $\varphi=0.86$  (Fig. 13k) is the most likely to produce the experimentally observed erosion map (Fig. 12), thereby  
 682 suggesting that this is the value of the particle spherical coefficient to be used in the numerical simulations.

683 [Figure 14 about here.]

684 In order to confirm this guess, the extended methodology described in Section 4.2 was applied to the  
 685 three values of  $\varphi$ , thereby determining three different impact angle functions (Figs. 14a,d,g). The  $R^2$  value  
 686 is very high and almost identical for the three fitting functions, thereby indicating that, unlike in the flat  
 687 specimen case, this parameter cannot be taken as an indicator of the accuracy of calibration. However,  
 688 the comparison between the predicted and measured erosion maps clearly reveals that  $\varphi=0.86$  produces  
 689 more accurate calibration compared to the two other values of this parameter. This is evident not only  
 690 by inspecting the maximum predicted erosion depth, which exceeds the experimental one by 36%, 29%,  
 691 and 22% for  $\varphi$  equal to 0.66, 0.76, and 0.86 (Figs. 14b,e,h), but also from the shape of the erosion crater,  
 692 which, as  $\varphi$  decreases, tends to be elongated along the horizontal axis and deviates from the evidence  
 693 (Figs. 14c,f,k). Further looking at Figs. 14c,f,k, it can be noted that, unlike the experimentally-determined  
 694 ones, the predicted erosion profiles show two local peaks for all values of  $\varphi$ . This is due to the fact that,  
 695 owing to the drag and pressure gradient forces, the trajectories are pushed away from the stagnation region,  
 696 and, therefore, the maximum impact number density is found at a certain distance from the center of the  
 697 specimen. This effect is reduced for  $\varphi=0.86$ , but it cannot be completely avoided. As a consequence,  
 698 a U-shaped crater is impossible to obtain with the current CFD model. Given the influence of  $\varphi$  on the  
 699 configuration of the particle trajectories, it would not be surprising to find better match with the observed  
 700 erosion hole with a higher  $\varphi$ . Nonetheless, values of  $\varphi$  larger than 0.86 would be out of the recommended  
 701 range for natural sands [1], and, therefore, they might produce an unphysical solution.

702 [Figure 15 about here.]

703 As explained in Section 4.2, one of the proposed modifications to the methodology of Masouri consists in  
 704 elevating the impact velocity modulus to power  $n$  before cell-averaging instead of applying the cell-average  
 705 operator first. The motivation behind this change was to make the procedure consistent with the E-CODE  
 706 library used for processing the parcel-wall impingement characteristics produced as output by Fluent. As  
 707 a practical counterpart to the mathematical derivation in Section 4.2, cell-averaging the impact velocity  
 708 modulus after elevating to power  $n$  was found to have a positive effect on the accuracy of calibration for  
 709 the GRE test case. This is evident in Fig. 15b, which is the same as Fig. 14k including also the two green-  
 710 colored profiles obtained when the cell-averaged impact velocity is applied to power  $n$ . The erosion depth  
 711 distribution corresponding to the green-colored profiles, shown in Fig. 15a, indicates a maximum predicted  
 712 erosion depth around 36% higher than the maximum measured value. The underestimation is therefore  
 713 much worse compared to that obtained when the elevation to power  $n$  is performed before cell-averaging  
 714 (Fig. 14h).

715 [Figure 16 about here.]

716 The results discussed so far indicate that, although the concept behind the calibration procedure of Man-  
 717 souri et al. [18, 19] and its local generalization is rather straightforward, achieving accurate predictions  
 718 might not be simple even for the calibration case. In fact, the predicted parcel-wall impingement charac-  
 719 teristics must be necessarily consistent with the observed erosion depth distribution, and this requires both  
 720 careful control of the experimental tests and reliable modelling of the particle-laden flow close to solid  
 721 walls. Nonetheless, it is observed that, despite the differences in the location of the maximum erosion, the  
 722 case-specific correlation shows much better performance compared to the comprehensive erosion model of  
 723 Oka et al. [8, 9], calibrated with respect to dry direct impact test experiments. Figure 16 shows the erosion  
 724 depth map predicted by the erosion model of Oka et al. [8, 9], and it clearly reveals that, even with  $\varphi=0.86$ ,  
 725 this model does not produce a wear pattern consistent with the experiment. The estimates in Fig. 16 refer  
 726 to a Vickers hardness equal to 30 GPa, but a sensitivity analysis revealed that the shape of the erosion hole  
 727 is substantially unaffected by the value of this parameter.

728 The comparison between calculated and measured erosion characteristics can be extended also to the  
 729 overall mass loss. If the mass reduction obtained from the balance is attributed a unit value, that ob-  
 730 tained from the integration of the experimentally determined erosion map was 0.754. Referring to the same  
 731 normalization, the calculated values for  $\varphi$  equal to 0.66, 0.76, and 0.86 were 2.197, 1.461, and 0.925, re-  
 732 spectively. This further confirms the conclusions reached by inspecting the surface of the erosion crater  
 733 in Fig. 14, namely, that, compared to 0.66 and 0.76, 0.86 is the most suitable value of particle spherical  
 734 coefficient for the current test case.

735

[Figure 17 about here.]

736 The calibrated erosion model for  $\varphi=0.86$  was finally employed to simulate the two other experimen-  
 737 tal tests, in which the testing time was increased keeping all other parameters essentially constant. Fig-  
 738 ures 17a,b are analogous to Fig. 14k for  $T$  equal to 15 and 30 minutes, respectively. These plots do not  
 739 represent a validation since, when using a one-way coupled steady-state CFD model, the wear profiles  
 740 scales linearly with testing time and concentration and, therefore, the CFD curves in Figures 17a,b can be  
 741 immediately obtained from those in Fig. 14k. Nonetheless, it is interesting to observe that, as the testing  
 742 time increases, the CFD model progressively overestimates the measured erosion depth, indicating that the  
 743 erosion hole **deepens** less-than-linearly with time. Such behaviour is well known in the literature, and it  
 744 was observed, for instance, by Nguyen et al. [22] when performing normal slurry jet impingement tests on  
 745 a flat stainless steel sample. With the aid of CFD simulations, Nguyen et al. [22] argued that the deviation  
 746 from the steady-state erosion was the consequence of the changes in the particle-laden flow produced by the  
 747 eroding surface, and this interpretation was recently confirmed by Messa and Malavasi [32], Parsi et al. [31]  
 748 and Duarte and De Souza [33]. Such unsteady effect cannot be captured by the steady-state CFD model em-  
 749 ployed in the present study and, therefore, the results in Figs. 17a,b are not surprising. The erosion profiles  
 750 in Figs. 17a,b have a clear correspondence with the time evolution of the overall mass reduction, shown in  
 751 Figs. 17c. Whereas the predicted mass loss obviously linearly increases with time, both experimental data  
 752 series (obtained using the balance and from volume integration of the profilometer data) increase less than  
 753 linearly, in agreement with the earlier findings of Nguyen et al. [22].

## 754 6. Conclusions and future developments

755 The present study **presents** an improved formulation of a methodology for the calibration of empirical  
 756 erosion models previously introduced by Mansouri et al. [18, 19], which relies on a combination of exper-  
 757 imental data and CFD results of a normal wet direct impact test. In the authors' opinion, the methodology  
 758 might have a strong impact on the engineering simulation of slurry erosion since it represents a practical  
 759 compromise to obtain, at least for specific flow conditions, reliable erosion predictions, avoiding, on the  
 760 one hand, the high computational burden of micro-scale erosion simulations solving for the physical laws  
 761 of damage and fracture mechanics and, on the other hand, the inherent uncertainty of mechanistic erosion  
 762 models and empirical erosion models based on dry direct impact tests.

763 **As a result of an extensive computational work and referring to original experimental data, the upgraded**  
 764 **methodology allows overcoming some well-known drawbacks of CFD-based erosion prediction [31].** The  
 765 main achievements are as follows.

- 766 • It has been demonstrated that the consistency between the parcel-wall impingement characteristics,  
 767 predicted by the CFD model, and the measured erosion distribution is a necessary requirement to  
 768 achieve accurate calibration of the erosion correlation. In this regard, the role played by the particle  
 769 spherical coefficient,  $\varphi$ , in the drag coefficient correlation is fundamental. It has been proposed to  
 770 take the calibration accuracy of the erosion model as a criterion to decide the proper value of  $\varphi$ ,  
 771 which is very difficult to quantify for natural abrasives. Particularly, it has been recommended to  
 772 run the numerical simulations for three reference values of particle spherical coefficient (0.66, 0.76,  
 773 and 0.87), thereby determining three different erosion correlations, and identify the value of  $\varphi$  which  
 774 produces the lowest deviation from the experiments in terms of maximum erosion depth. It has been  
 775 also shown that, in some situations (**but not always**), the goodness-of-fit of the regression impact angle  
 776 curve, quantified by the  $R^2$  value, might be also taken as a criterion to identify the most suitable value  
 777 of  $\varphi$ . **This contributed to widening the scope of the combined CFD/experimental methodology, which**  
 778 **does not only allow calibrating an empirical erosion formula, but it also gives some indication on the**  
 779 **proper value of  $\varphi$  to be set in the CFD model.**
- 780 • **In the new methodology, the coefficient  $F_s$  does no longer appear explicitly in the erosion formula but**  
 781 **it is included in the coefficient  $K$ . This avoids the need to handle two different parameters related with**  
 782 **particle shape, namely,  $\varphi$  and  $F_s$ , which are not independent with each other as they are associated with**  
 783 **the same geometrical feature of the particle phase. Additionally, removing  $F_s$  clearly underlines that**  
 784 **the developed erosion formula cannot be applied to other particles than those used for the calibration.**
- 785 • **In the upgraded equation for the calibration of the impact angle function (Eq. 21), the impact velocity**  
 786 **term is elevated to power  $n$  before cell-averaging, and not after cell-averaging as in the original formu-**  
 787 **lation. This made the procedure more consistent with the post-processing of the parcel-wall impact**  
 788 **data made by the E-CODE library, resulting in an improved calibration accuracy.**

- 789 • In CFD model, the parcel-wall impingement characteristics were not evaluated at the wall boundaries,  
790 as implied by the point-particle approximation, but at an upstream distance from the wall equal to half  
791 of the particle diameter, following a procedure previously developed by two authors of this study [29].  
792 This allowed not only improving the accuracy of the numerical evaluation of the impacts, but it also  
793 reduced their dependence upon the near-wall mesh resolution and the near-wall treatment method.
- 794 • A ring-based formulation of the methodology was proposed to make its application to statistically  
795 axi-symmetric flows more straightforward.

796 Although the present study represents a step forward in the development of reliable and practically use-  
797 ful CFD-based erosion prediction models, two tasks remain unresolved and have been shelved for future  
798 research. The former is the characterization of the velocity exponent  $n$ , which was here assumed equal  
799 to 2.4 as initially recommended by Mansouri et al. [18, 19] but it is likely to be dependent on the mate-  
800 rials involved in the process. Criteria for deciding the proper value of  $n$  should be established, and this  
801 requires disposing of experimental data for different jet velocities, abrasive particles, and target materials.  
802 The second drawback of this work is the lack of a validation of the calibrated erosion correlation. This  
803 study focused on the calibration phase and showed that even achieving a good calibration accuracy is more  
804 challenging than it might appear at first sight. Given the experimental database available, only exploring  
805 the effect of testing time was possible. However, even if the findings are in line with previous experimental  
806 and numerical results, this did not represent a real validation, owing to the assumptions underlying the CFD  
807 model used which make the CFD predictions at different testing time not independent from each other.  
808 Validating the predictions of the calibrated formula against other experimental tests (e.g. jet impingements  
809 at different velocity and nozzle-to-specimen angles, or different types of flows involving the same com-  
810 bination of materials) will be definitely required to demonstrate the effectiveness of the methodology for  
811 engineering purposes.

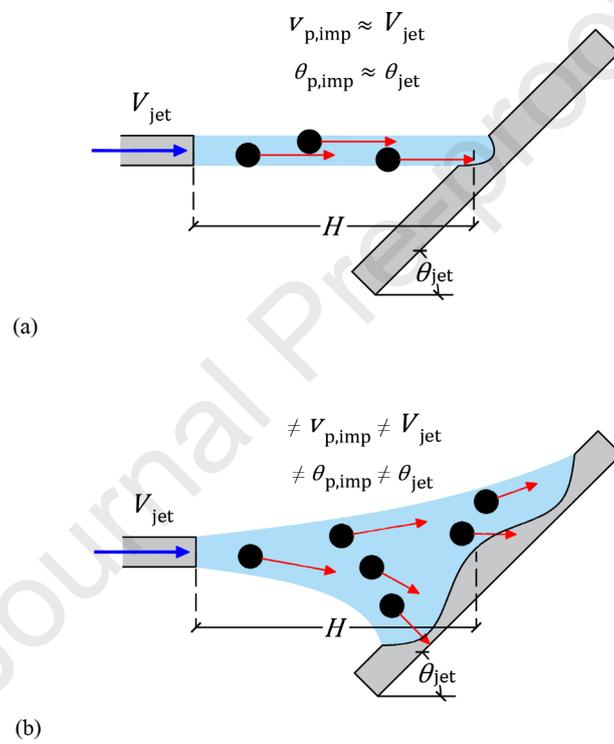
## 812 Acknowledgements

813 The numerical simulations performed in this study were run on the GALILEO supercomputer at  
814 CINECA as part of the IS CRA research project IsC78 (MASS - Modelling Anisotropic flows in Single-  
815 phase and Slurry conditions). Authors would like to acknowledge Cristina Tedeschi from the Laboratory  
816 for Diagnostic and Investigation on Built Heritage Materials of Politecnico di Milano, who scanned the  
817 specimens and performed the sieve analysis of the abrasive particles used in the aluminum experiments,  
818 ENI S.p.A., which partially supported this research, and NOV Wellbore Technologies, which provided the  
819 GRE specimens.

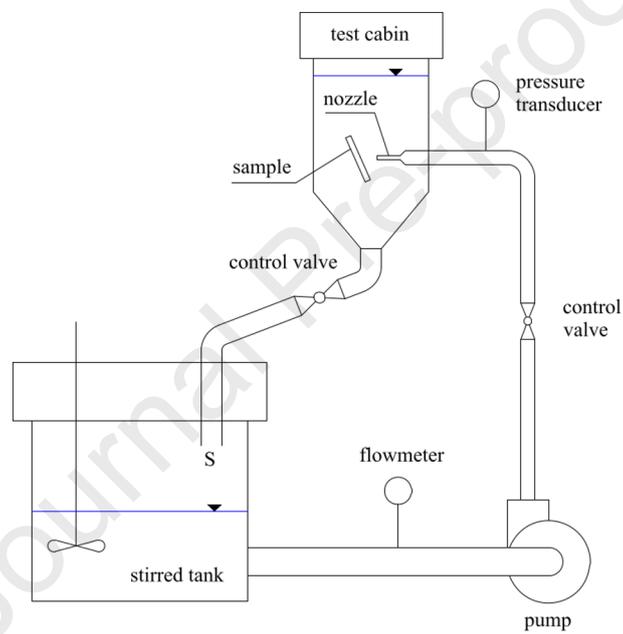
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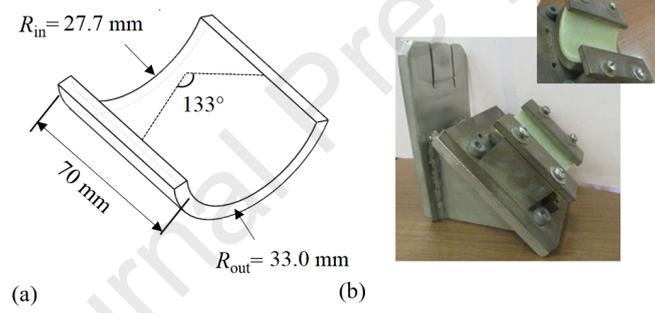
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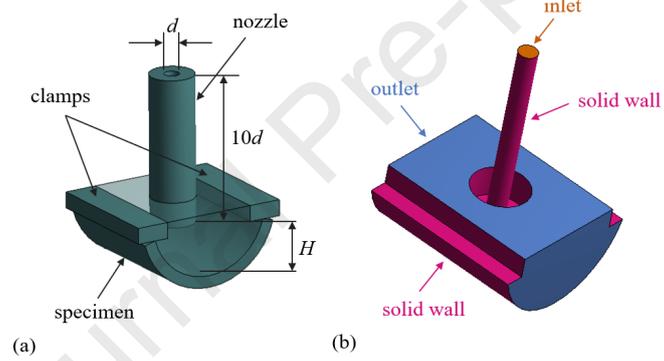
**Fig. 1.** Sketch of (a) a dry direct impact test and (b) a wet direct impact test.



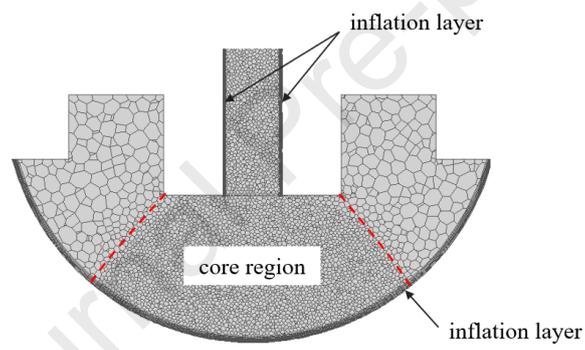
**Fig. 2.** Simplified sketch of the wet direct impact test facility at the Hydraulic Laboratory of Politecnico di Milano.



**Fig. 3.** (a) the GRE specimens and (b) their clamping holder.



**Fig. 4.** (a) computational domain and (b) boundary conditions for the curved GRE specimen case.



**Fig. 5.** restricted view of the discretization of the mid plane for the curved GRE specimen case.

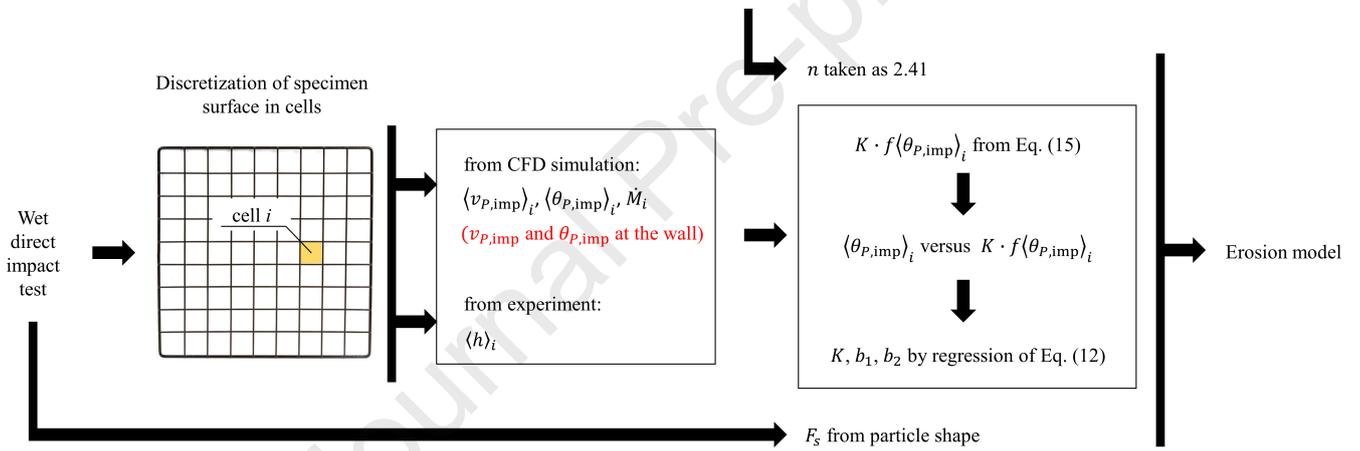
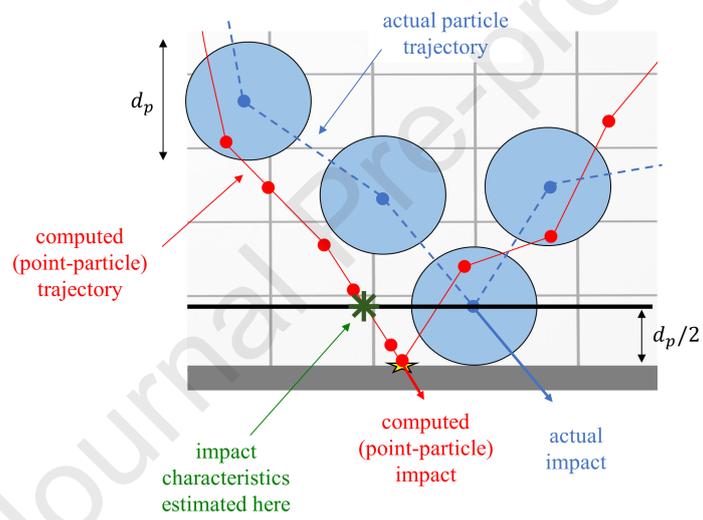


Fig. 6. Flowchart of the original methodology of Mansouri [19].



**Fig. 7.** evaluation of the parcel-wall impingement characteristics according to the strategy of Messa and Wang [29].

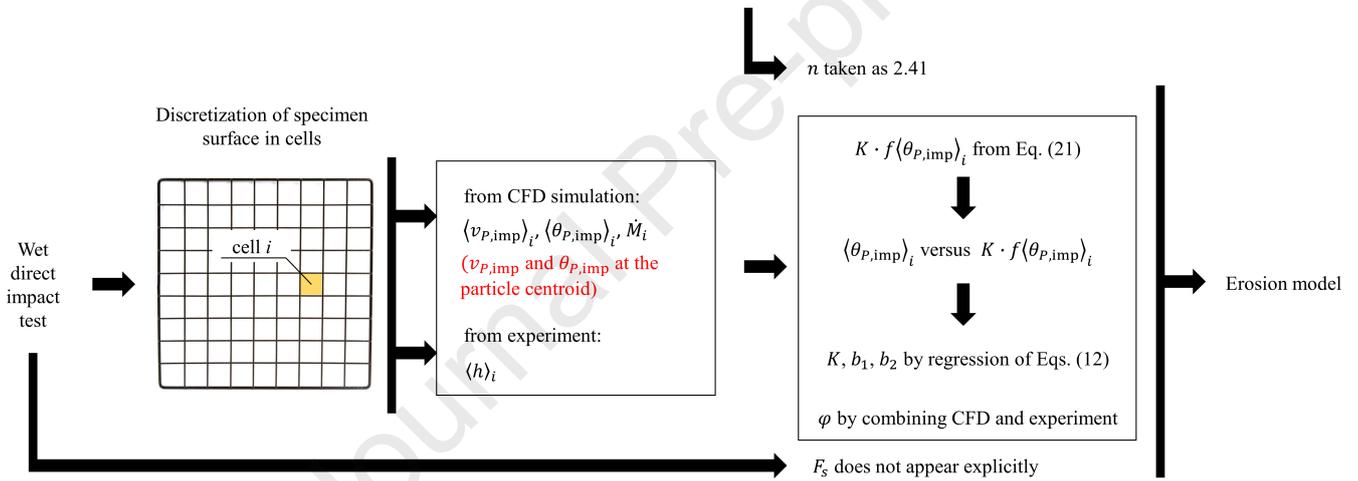
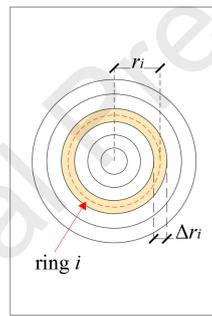
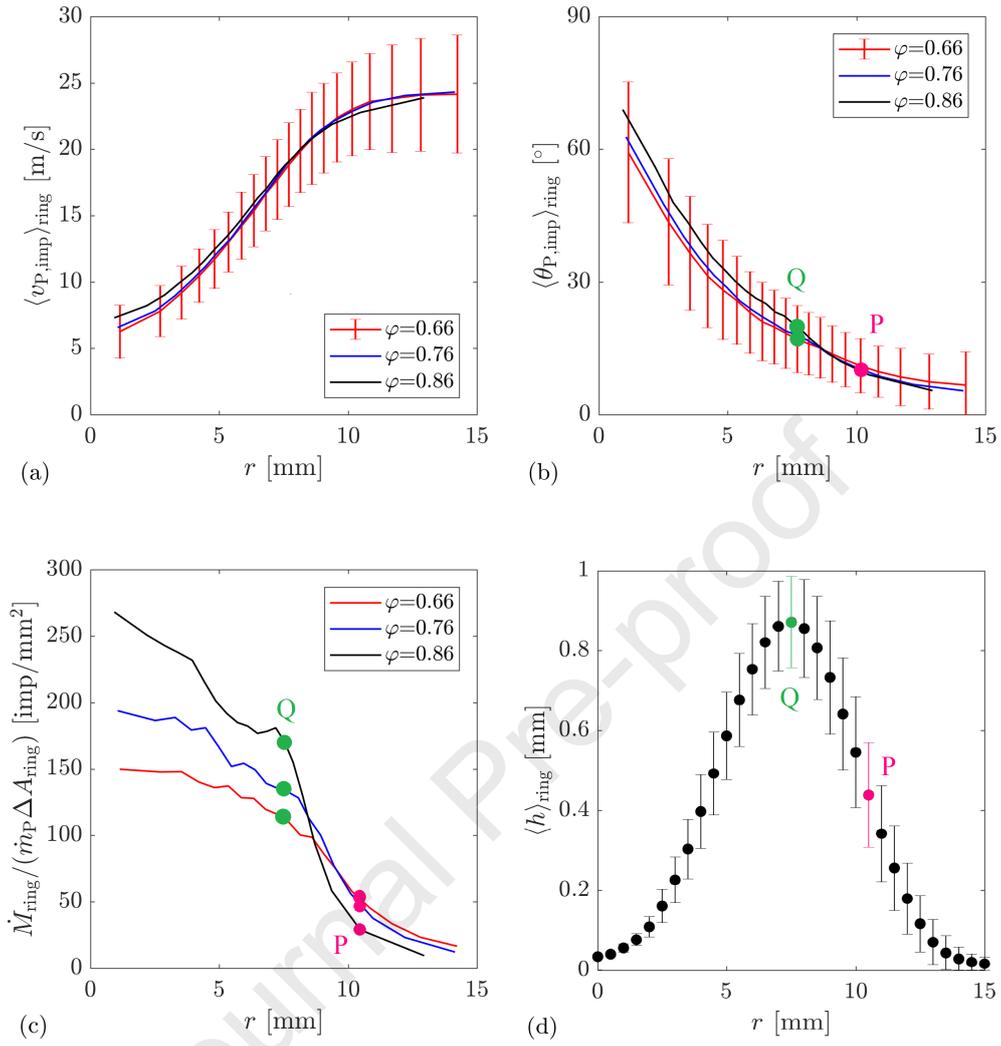


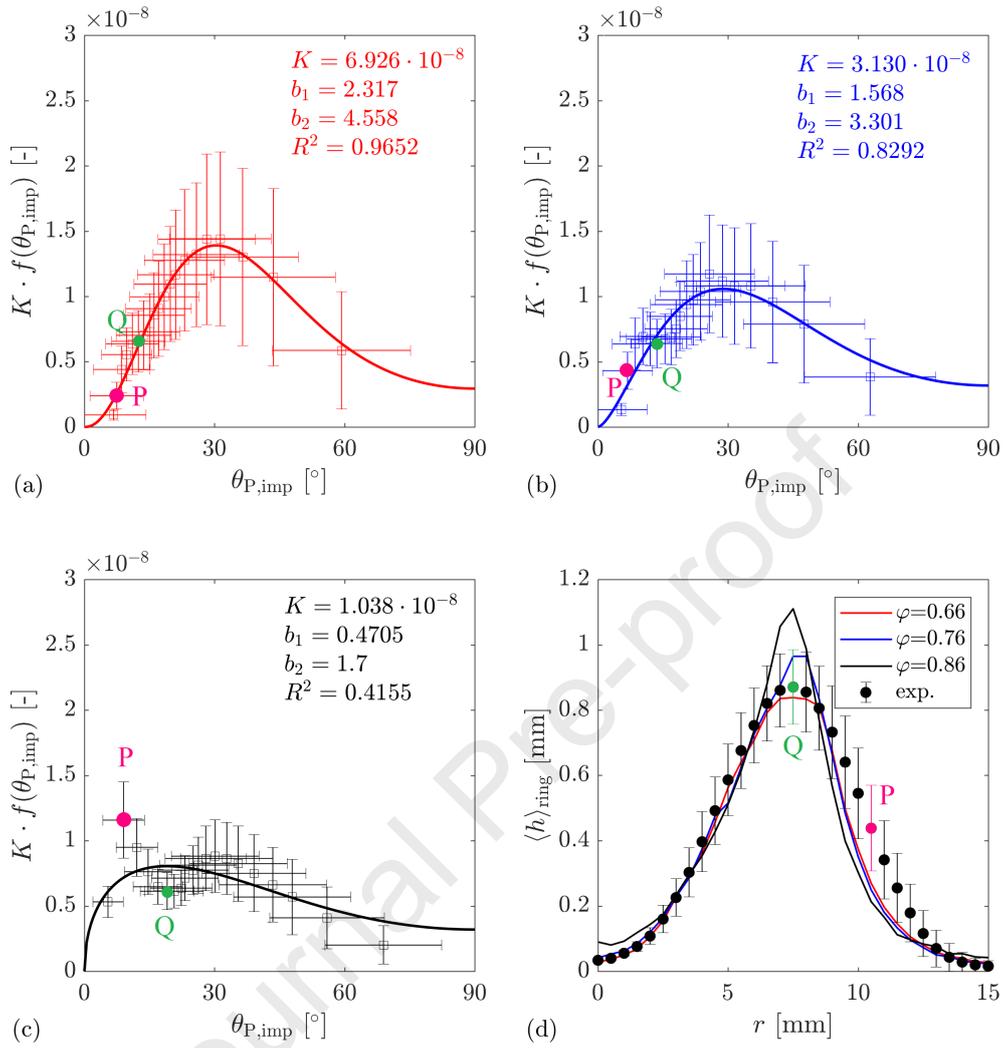
Fig. 8. Flowchart of the improved methodology.



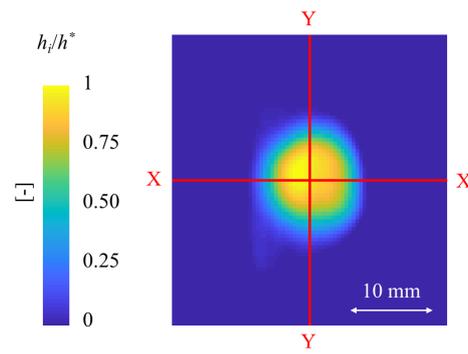
**Fig. 9.** subdivision of the surface of the specimen into concentric rings.



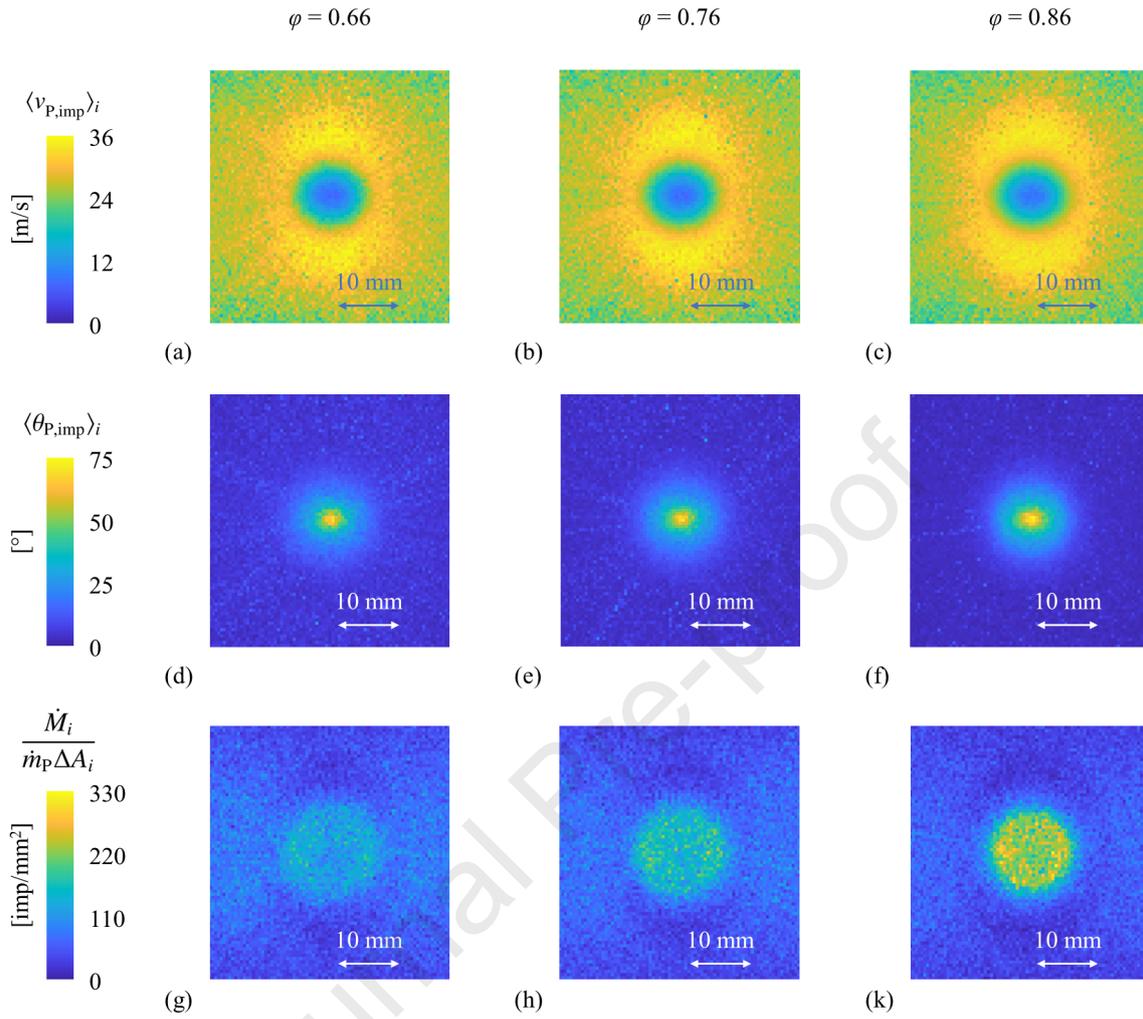
**Fig. 10.** (a,b,c) output of the Eulerian-Lagrangian simulations: ring-averaged profiles of (a) mean parcel-wall impact velocity, (b) mean parcel-wall impact angle, (c) impact number density. (d) experimentally-determined ring-averaged erosion depth profile. In plots (b,c), the error bars, equal to the standard deviation over the impingements occurring in each ring, are similar for the three values of  $\varphi$ , and they have been plotted only for  $\varphi = 0.66$  for the sake of clarity. In plot (d), the vertical bars represent the azimuthal variability of the erosion depth over each ring.



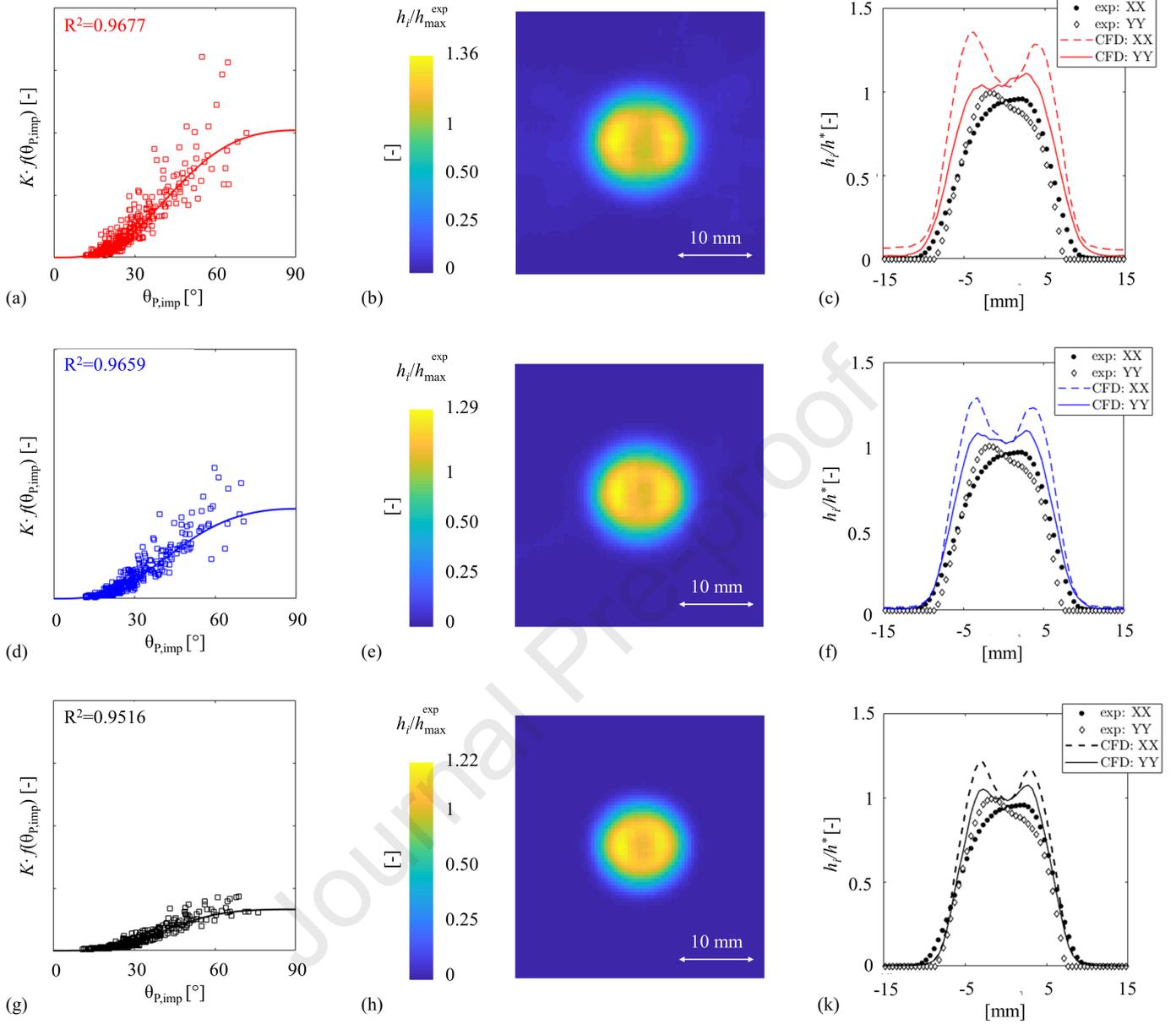
**Fig. 11.** calibration of the impact angle function  $K \cdot f(\theta_{P,imp})$  for  $\varphi$  equal to (a) 0.66, (b) 0.76, and (c) 0.86. (d) erosion depth profiles obtained from the three regression curves and from the ring-averaged profilometer data. The horizontal and vertical bars in plots (a-c) indicate the variability of the  $\theta_{P,imp}$  and  $K \cdot f(\theta_{P,imp})$  values in each ring, respectively. In plot (d), the vertical bars represent the azimuthal variability of the erosion depth over each ring.



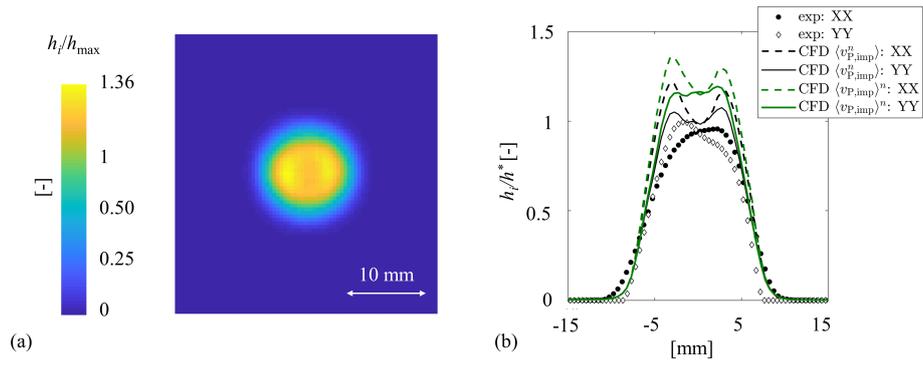
**Fig. 12.** experimentally-determined wear depth distribution over the inner surface of the GRE specimen for case  $V_{jet}=35.53$  m/s,  $C = 0.0085$ , and  $T=5$  min. In order to preserve confidentiality, the data have been normalized with respect to the maximum value, referred to as  $h^*$ . Lines XX and YY represent the horizontal and vertical axes of the specimen.



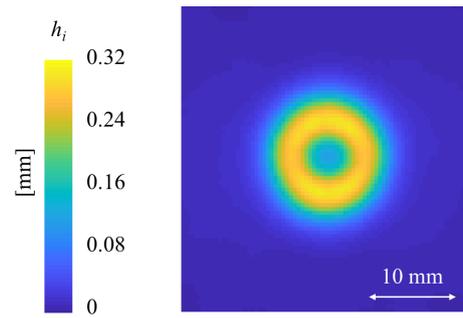
**Fig. 13.** calculated parcel-wall impingement statistics over the inner surface of the GRE specimen for case  $V_{jet}=35.53$  m/s,  $C = 0.0085$ , and  $T=5$  min: (a-c) locally averaged impact velocity (d-f) locally averaged impact angle (g-k) impact number density. The columns from left to right refer to increasing values of the particle spherical coefficient.



**Fig. 14.** results of the second test case for  $\varphi$  equal to (a-c) 0.66, (d-f) 0.76, and (g-k) 0.86. Left column: calibration of the impact angle function  $K \cdot f(\theta_{p,imp})$  by combining the experimentally determined erosion depth map in Fig. 12 and the calculated parcel-wall impingement statistics in Fig. 13. The limits of the vertical axes are the same for all plots, but they have not been shown to preserve confidentiality. Center column: predicted erosion depth map, normalized by the maximum experimental value in Fig. 12. Right column: comparison of predicted and measured erosion depth profiles along the horizontal (XX) and the vertical (YY) axes, normalized by the maximum experimental value in Fig. 12.

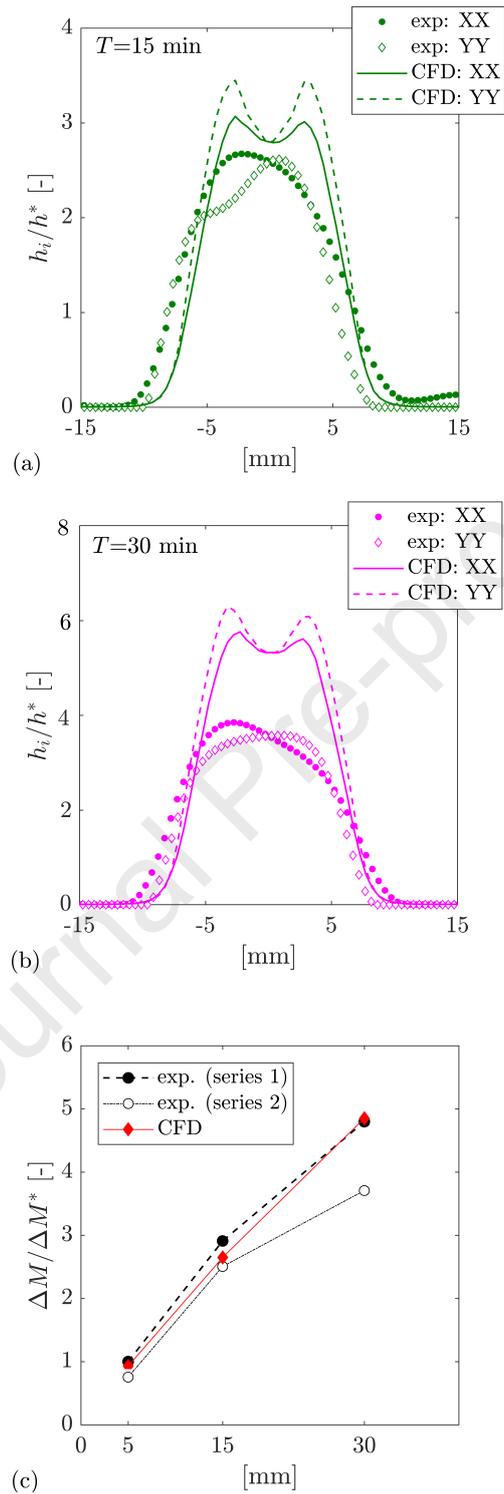


**Fig. 15.** (a) analogous of Fig. 14h when, in Eq. 21, the impact velocity is elevated to power  $n$  after cell averaging; (b) analogous of Fig. 14k including also the erosion depth profiles obtained by applying Eq. 21 with impact velocity elevated to power  $n$  after cell averaging.



**Fig. 16.** wear depth distribution over the inner surface of the GRE specimen for case  $V_{\text{jet}}=35.05$  m/s,  $C = 0.0085$ , and  $T=5$  min obtained by applying the erosion model of Oka et al. [8, 9] with Vickers hardness of 30 GPa.

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**Fig. 17.** (a,b) analogue of Figs. 14c,f,k for testing time equal to (a) 15 minutes and (b) 30 minutes. The normalization factor,  $h^*$ , is the maximum measured erosion depth for the 5 minutes test. (c) Overall mass loss versus time for the three test case. Series 1 and 2 are the values obtained from the balance and from volume integration of the profilometer data, respectively. The normalization factor,  $\Delta M^*$ , is the data obtained from the balance for the 5 minutes test.

Eroding sample	Erosion depth	$\approx 0.1$ mm
	Mass of specimen	$\pm 3.5$ mg
	Target density, $\rho_t$	$\pm 50$ kg/m <sup>3</sup> (GRE samples)
Abradant	Particle density, $\rho_p$	Substantially ineffective
	Mean particle diameter, $d_{p,50}$	Reduced by about 13% during the test on aluminum specimen
Testing conditions	Jet bulk-mean velocity, $V_{jet}$	$\pm 2.4\%$
	Nozzle-to-specimen angle, $\theta_{jet}$	$\pm 1^\circ$
	Nozzle-to-specimen distance, $H$	$\pm 0.05$ mm
	Solid volume fraction, $C$	$\pm 0.001$

**Table 1**  
Key elements of the uncertainty analysis.

Eroding sample	Type of material	Aluminum alloy EN AW-608	Glass Reinforced Epoxy (GRE)
	Target density, $\rho_t$ [kg/m <sup>3</sup> ]	2700	2050±50
	Surface of specimen	Flat	Curved
	Dimensions of specimen [mm]	80x60x5	See Fig. 3a
Abradant	Type of material	Coarse silica sand	Fine silica sand
	Particle density, $\rho_p$ [kg/m <sup>3</sup> ]	2500	2500
	$d_{p,10}$ [mm]	0.25	0.075
	$d_{p,50}$ [mm]	0.35	0.125
	$d_{p,90}$ [mm]	0.46	0.200
Testing conditions	Jet velocity $V_{jet}$ [m/s]	35.53±0.71	≈35
	Nozzle-to-specimen angle, $\theta_{jet}$ [°]	90±1	90±1
	Nozzle-to-specimen distance, $H$ [mm]	12.7±0.05	12.7±0.05
	Solid volume fraction, $C$ [-]	0.0085±0.001	0.0076 to 0.0085
	Testing time, $T$ [min]	5	5, 15, 30

**Table 2**  
Summary of the experimental conditions.

Particle-laden flow	Phase coupling regime	One-way coupling
Carrier fluid	Flow equations Turbulence model Near-wall treatment	RANS RSTM Scalable wall function
Particle phase	Forces included in the parcel equation of motion  Particle turbulent dispersion model Parcel-wall restitution coefficients Evaluation of impact characteristics	Drag force (Haider and Levenspiel [25]) Lift force (Mei [26]) Pressure gradient force Virtual mass force $C_{vm}=0.5$ Random Walk Model Forder et al. [28] At particle centroid

**Table 3**

Modelling features of the Eulerian-Lagrangian model. Unspecified settings are the default ones in Ansys Fluent version 19.2.

Carrier fluid	Type of mesh Number of inflation layers [-] First inflation layer thickness [mm] Number of cells [millions] Type of solver Gradient discretization method Pressure discretization scheme Discretization scheme for all other variables Convergence criterion Under-relaxation factors	Polyhedral with inflation layer 5 0.05 $\approx 1.26$ (test case 1), $\approx 1.45$ (test case 2) Coupled, pressure-based Green-Gauss Node-based PRESTO! Second Order Upwind $10^{-4}$ (normalized residuals) Default settings
Particle phase	Tracking mode Number of tracked parcels [-] Tracking parameters [-]	Steady-state 50000 Default settings

**Table 4**

Key numerical settings of the CFD model. Unspecified settings are the default ones in Ansys Fluent version 19.2.

# An improved CFD/experimental combined methodology for the calibration of empirical erosion models (Abstract Ref. number: WEAR2021\_0102)

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## Highlights

- An existing CFD/experimental methodology for calibrating erosion models is improved
- The new methodology allows for higher accuracy of calibration
- It also allows for fine tuning of the particle spherical coefficient
- The methodology is applied to slurry jet tests on flat and curved samples

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**Declaration of interests**

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.

The authors declare the following financial interests/personal relationships which may be considered as potential competing interests:

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