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DI INGEGNERIA CIVILE, CHIMICA  
E AMBIENTALE

# Comparison of nasal anatomies using computational fluid dynamics

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# Introduction

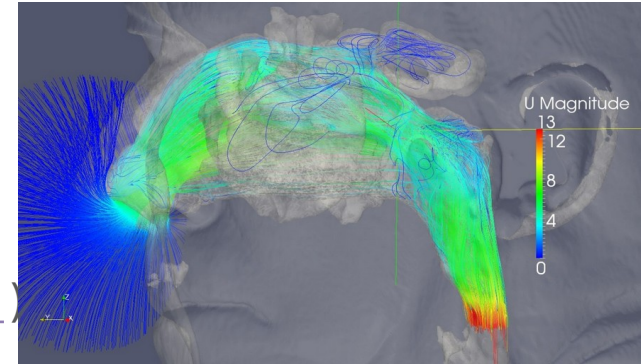
- Nasal breathing difficulties (NBD) are as **widespread** and important as **difficult to diagnose**;
- Surgical corrections are often necessary, but the **failure rate is high** (more than 50%) resulting in higher health care costs.

*D. A. Campbell, M. G. Moghaddam, J. S. Rhee, and G. J. M. Garcia.*

*Narrowed Posterior Nasal Airway Limits Efficacy of Anterior Septoplasty. 2021*

# Introduction - OpenNOSE

- The OpenNose community is active since 2011 and a Website ([www.open-nose.org](http://www.open-nose.org)) will be launched in 2023;
- Multidisciplinary community (~30 people) and PoliMi (Milan, Italy) is the group leader.
- It's driven by clinical problems and ENT surgeons are integral part of the group;
- It aims at developing virtual surgeries to give support to surgeons by evaluating different virtual operations before the real one.
- This project will foster the emerging community of Computational Rhinology.
- DNS, experimental and anatomy data will be freely available to anyone to create a standard.





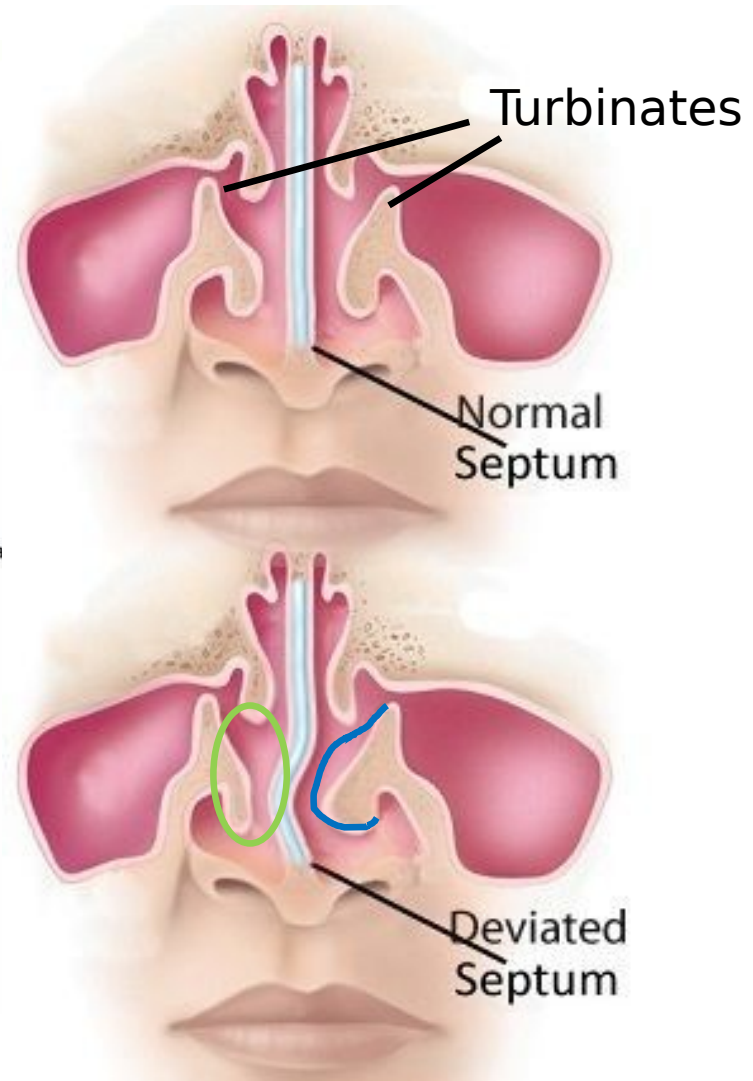
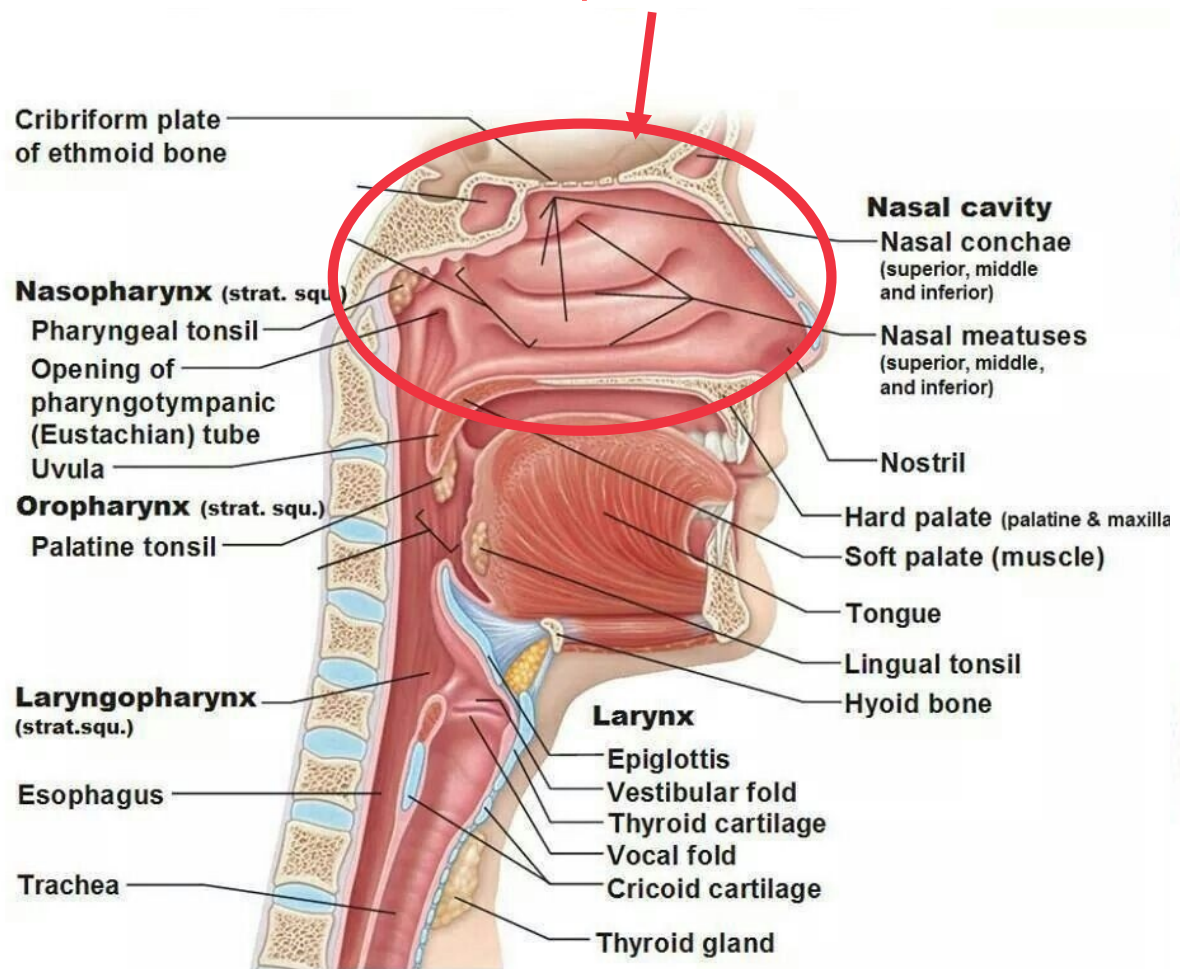
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**Some background.**

# Medical background

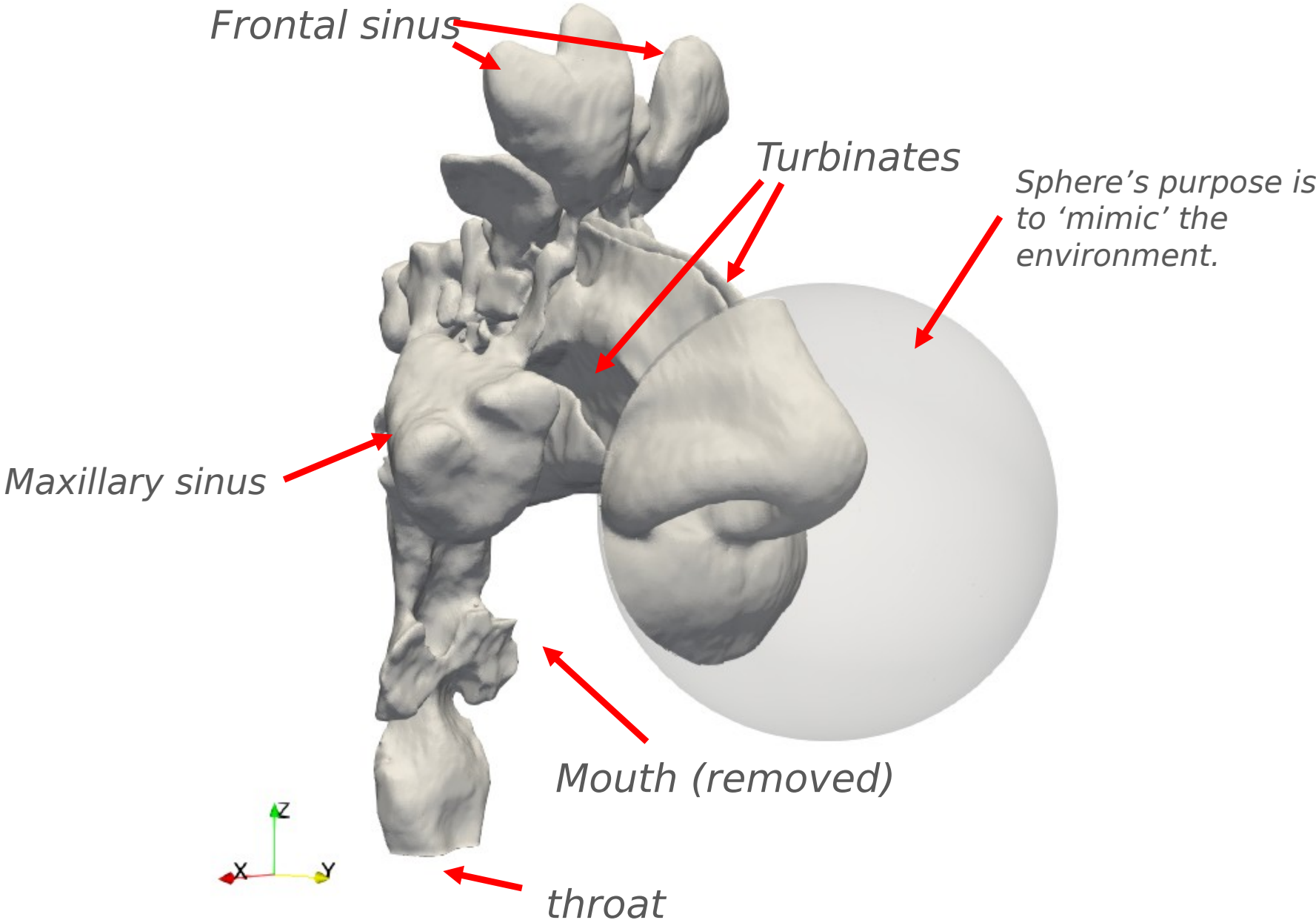
## Deviated septum



Reduction of the area where air can flow, in a nostril

Hypertrophy of the turbinate of the other nostril

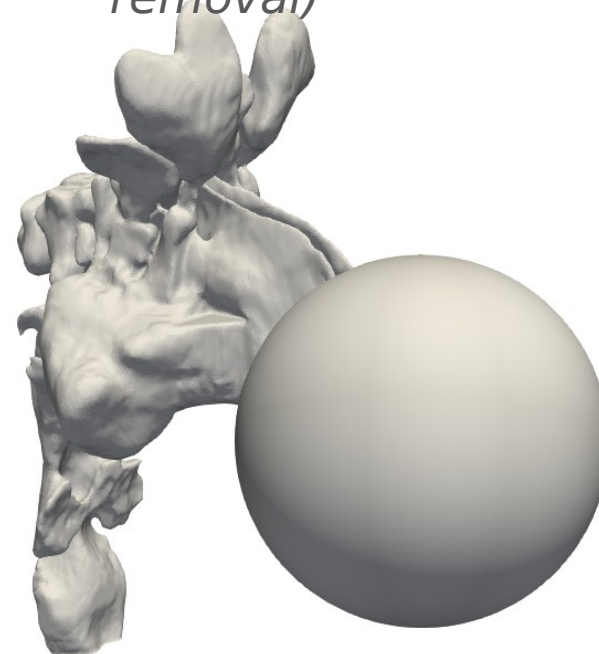
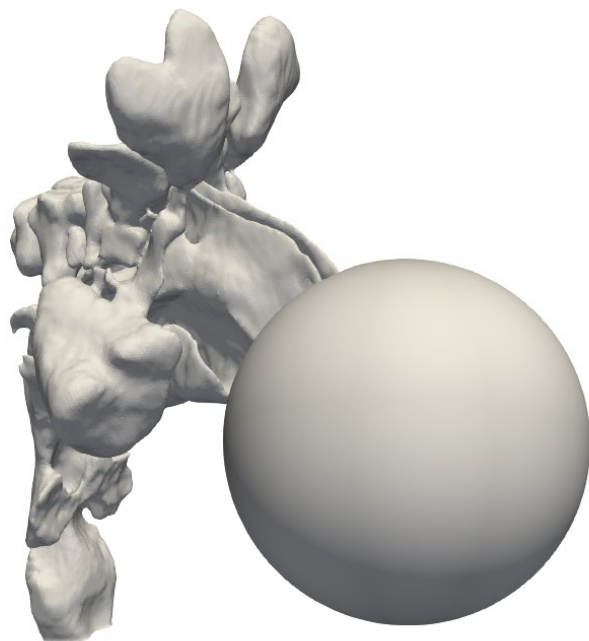
# Pre-surgery anatomy



# Virtual surgery effect

*before the surgery*

*after maxillectomy (maxillary sinus removal)*



# Open questions

Some of the open questions which we are working on:

1. Which numerical model makes the comparison more realistic, especially when comparing pre- and post-op;
2. How to measure well-being of a patient;
3. How to compare well-being before and after the surgery;
4. How to properly share the results with surgeons.
5. ...

# Open questions

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2. How to measure well-being of a patient;
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5. ...

# First question

*FLOW CONTROL  
FOR TURBULENT  
DRAG REDUCTION*



*FLOW CONTROL  
FOR NBD  
CORRECTION*

- The analogy can be made due to previous experience with flow control.
- In both cases the flow is manipulated and it must be compared before and after the application of control.
- The application of control alters the natural state of flow.

# First question

*FLOW CONTROL  
FOR TURBULENT  
DRAG REDUCTION*

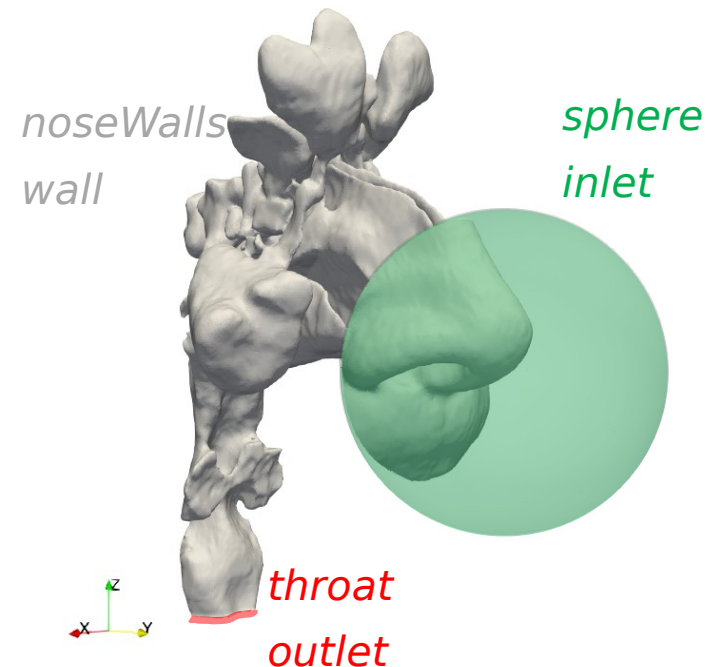


*FLOW CONTROL  
FOR NBD  
CORRECTION*

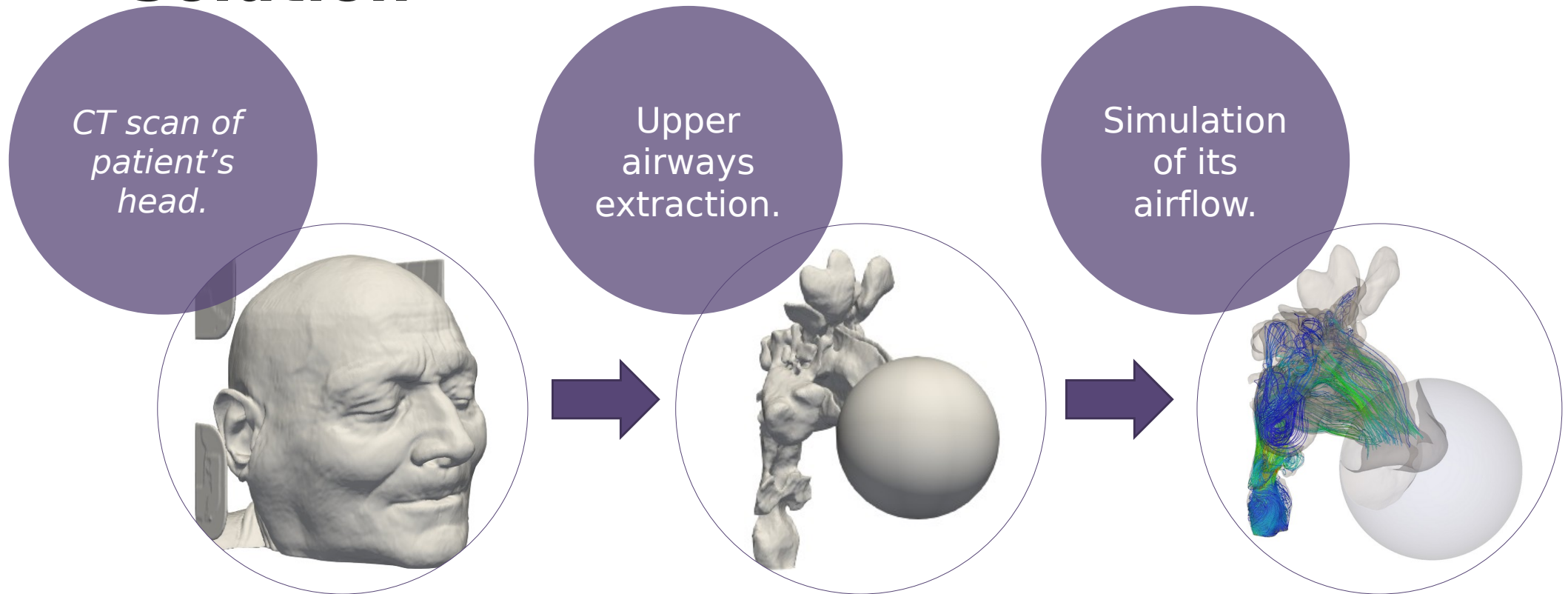
To compare pre- and post- surgery anatomies we can apply the following boundary conditions:

- **Constant Pressure ;**
- **Constant Flow Rate ;**
- **Constant Power Input**

*Hasegawa, Quadrio, Frohnapfel, J. Fluid Mech. (2014)*



# Workflow: CT Scan to Numerical Solution



- *3D slicer: used to create geometry*
- *STL file: obtained from an hospital based in Milan.*

- *Hounsfield Unit (HU)*
- *Interaction with Medical Doctors important to create realistic geometry.*

- *LES – WALE model*
- *Unsteady Transitional flow*

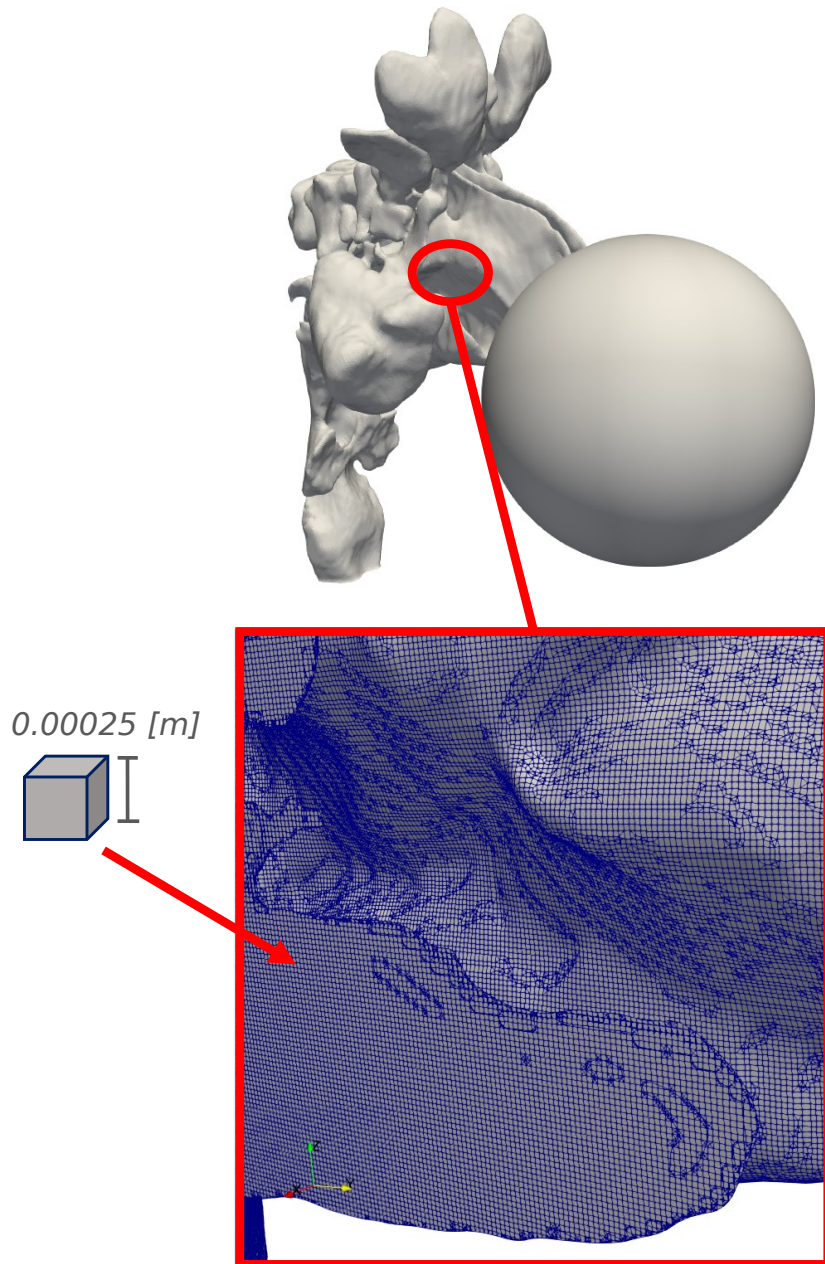


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# Meshing.

# Meshing



Mesh stats:

Cells ~ 15000000

of which:

Hexahedra ~ 14500000

Prisms ~ 50000

Wedges: 0

Pyramids: 0

Tet wedges: ~ 22

Tetrahedra: 0

Polyhedra: ~ 175000

Mesh quality:

Non-ortho: Max: ~ 60 , average: ~ 4

Max skewness: ~ 3



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# Numerical schemes and solver.

# fvSchemes & fvSolution

ddtSchemes: backward

gradSchemes: Gauss linear

divSchemes: default Gauss linear

laplacianSchemes: Gauss linear limited  
0.777

interpolationSchemes: linear

snGradSchemes: limited 0.777

fluxRequired: no

wallDist: method meshWave

solvers

p

—solver: GAMG

—tolerance: 1e-5

—smoother: DICGaussSeidel

U

—solver: smoothSolver

—smoother: symGaussSeidel

PIMPLE

momentumPredictor: yes

nOuterCorrectors: 1

nCorrectors: 2

nNonOrthogonalCorrectors: 1

relaxationFactors: fields & eqns to 0.9



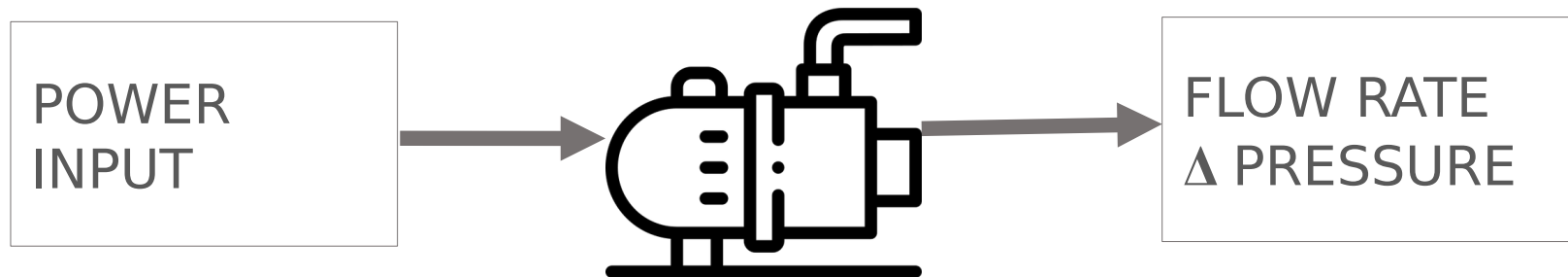
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# Constant Power Input setup.

# Boundary condition explained

- Constant Power Input boundary condition is typical for pumps where a certain power comes from the shaft and pressure gradient and flowrate comes after that.



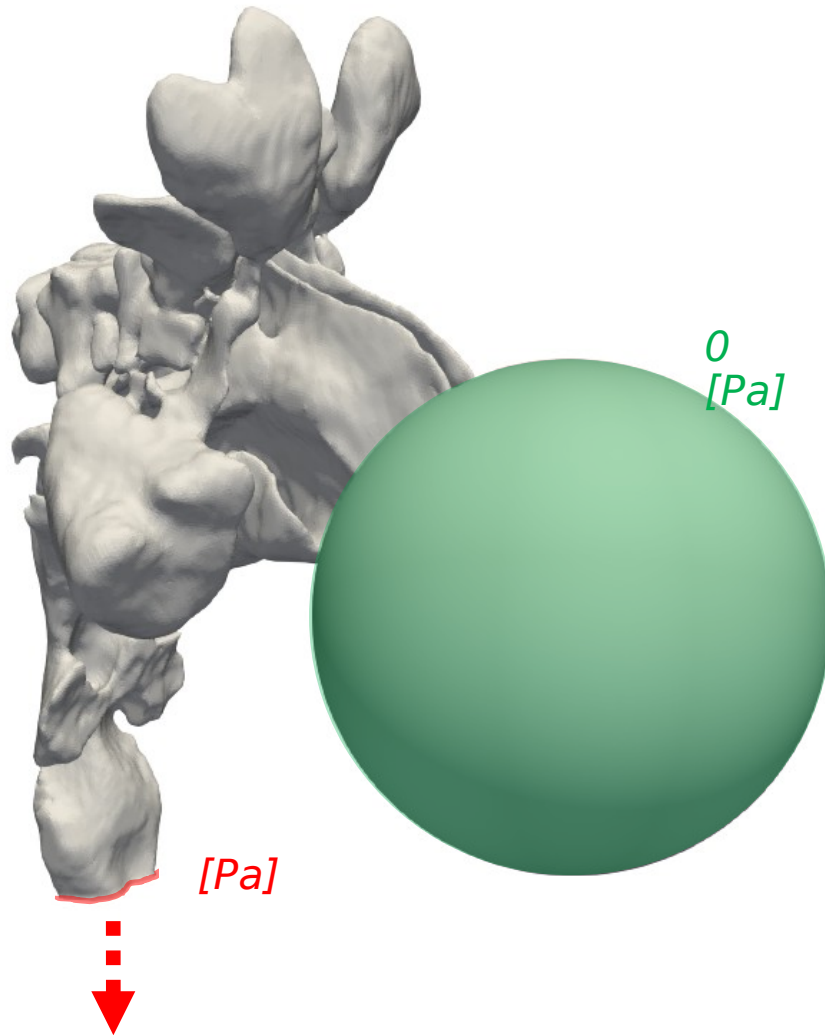
- The boundary condition was implemented in the ESI version using *swak4foam* which allowed us to write a clean, straightforward and readable code.

# Implementation

- Power Input boundary condition is expressed as:
  
- The boundary condition was written as follows:

```
throat
{
  type groovyBC;
  variables
  (
    "CPI=-0.00533";
    "Qdot=sum(phi)";
  );
  valueExpression "CPI/Qdot";
  value uniform -20;
}
```

# Implementation



1. **Initial guess** for pressure is the result of previous calculations.
2. then, with this value of pressure, **a timestep is run** and a certain **flowrate will be calculated** on the throat.
3. This flowrate, by using the expression of the **power input**, gives us a new estimate for the throat pressure and, with that, the loop restarts.



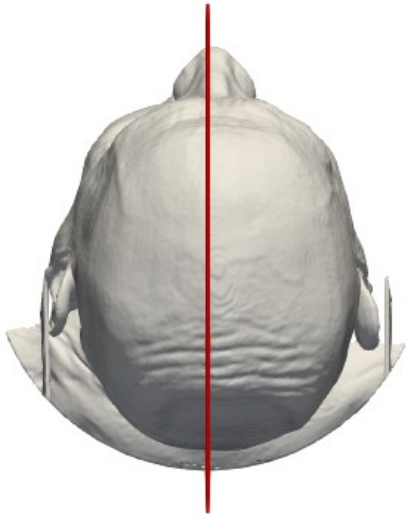
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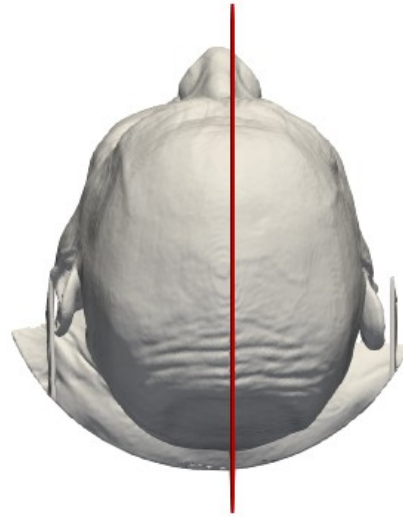
# **Pre-surgery (preop) results and comparison.**

# Slices

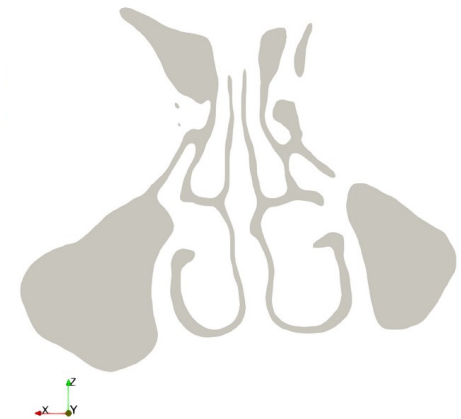
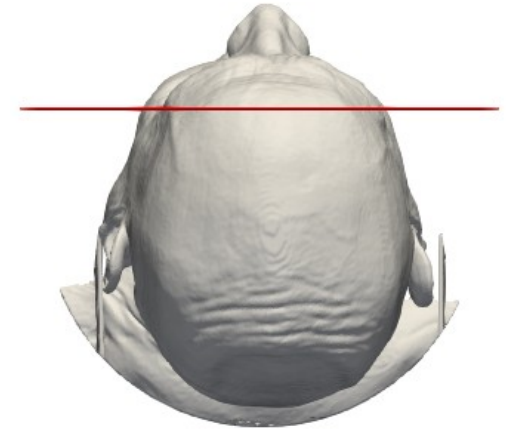
*sagittal plane 1*



*sagittal plane 2*



*coronal plane*



Comparisons will be carried on on the planes highlighted in red.

# Pre-surgery – CPG boundary conditions

noseWalls

```
type    fixedValue;
value   uniform (0 0 0);
```

sphere

```
type    pressureInletOutletVelocity;
value   uniform (0 0 0);
```

throat

```
type    pressureInletOutletVelocity;
value   uniform (0 0 0);
```

$p$

noseWalls

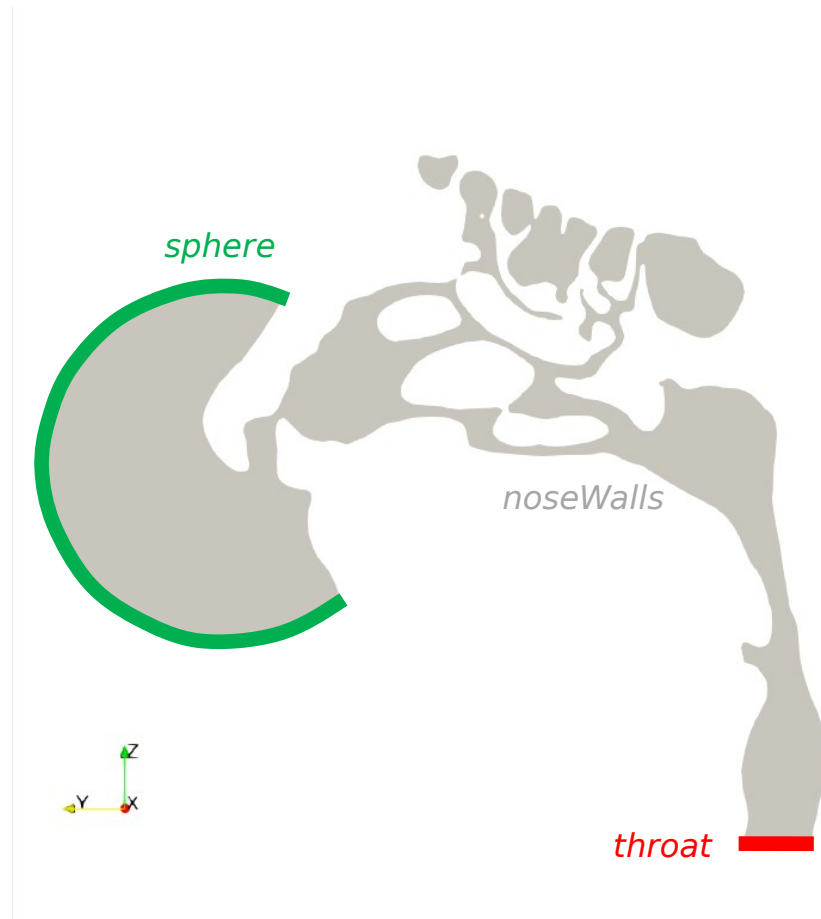
```
type    zeroGradient;
```

sphere

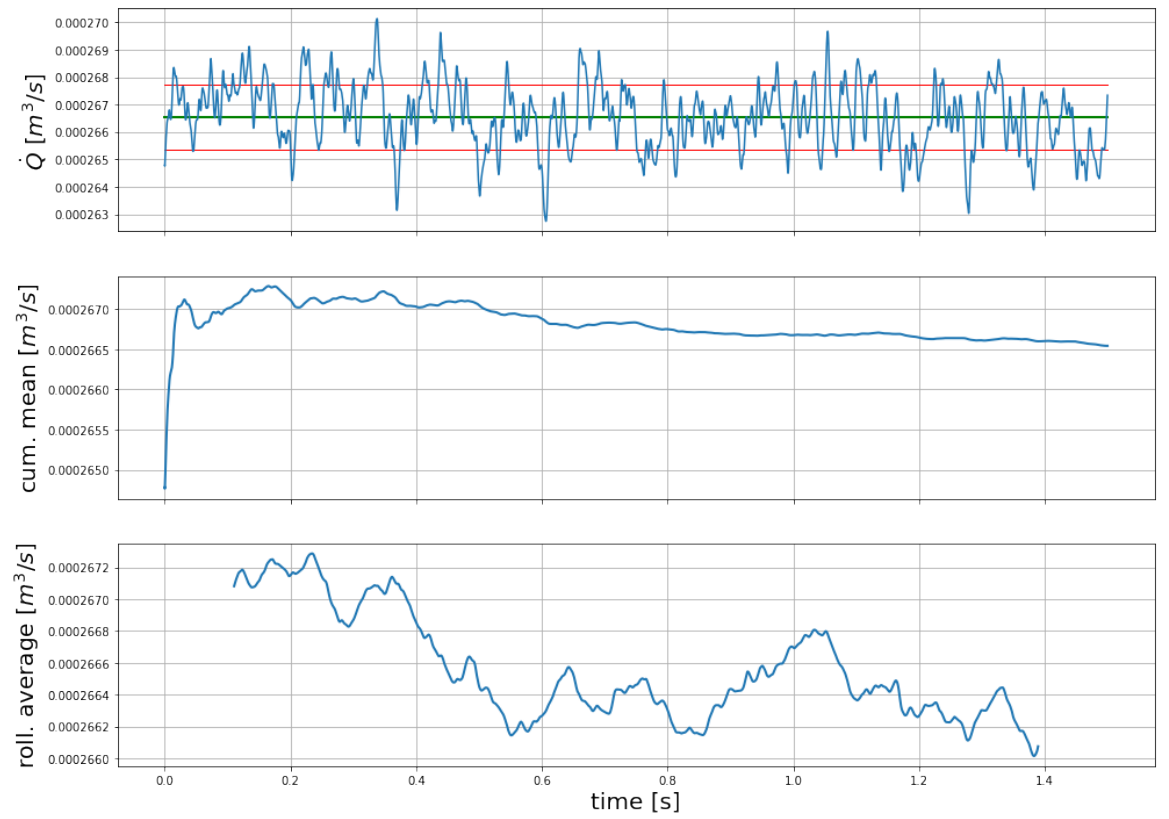
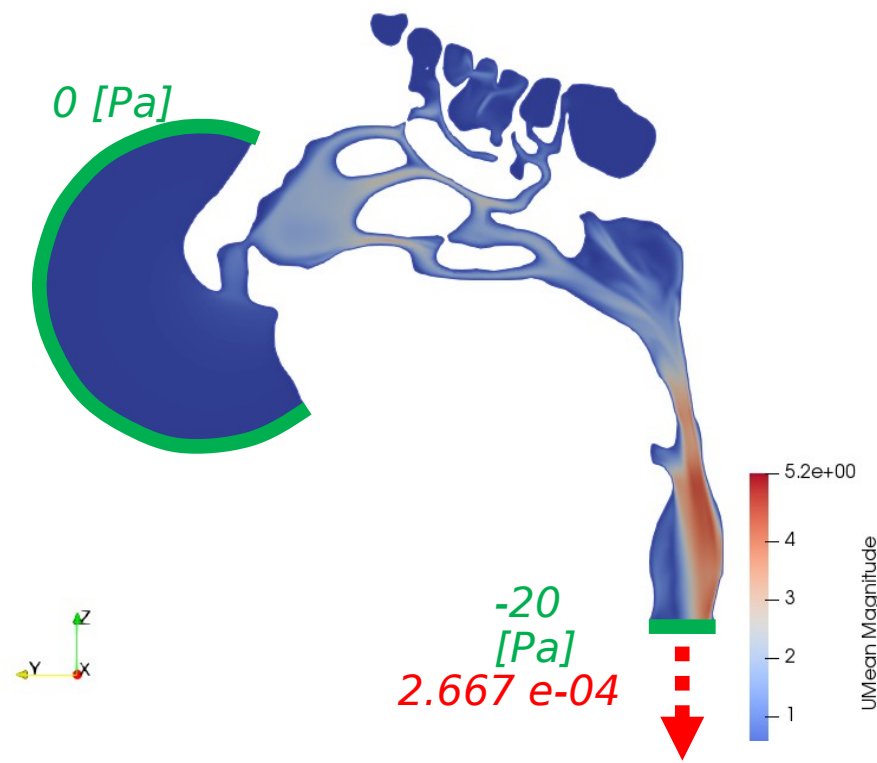
```
type    totalPressure;
p0      uniform 0;
value   uniform 0;
```

throat

```
type    totalPressure;
p0      uniform -20.;
value   uniform 0;
```



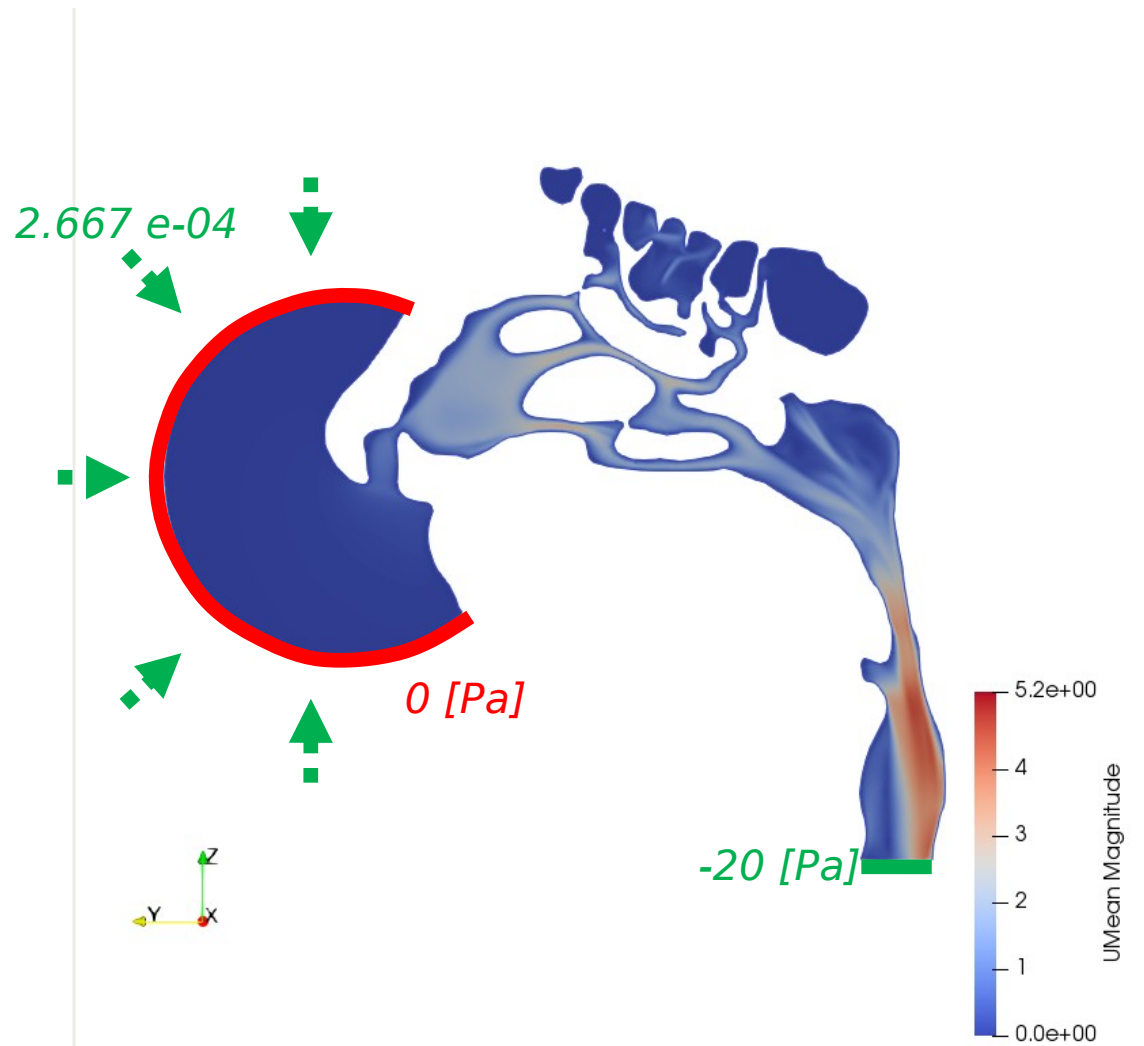
# Pre-surgery – CPG results & statistics



The condition of statistically steady solution is reached since the simulation is steady inside the 0.1%.

# Pre-surgery – CFR boundary conditions

noseWalls	type	fixedValue
	value	uniform (0 0 0)
sphere	type	flowRateInletVelocity
	volumetricFlowRate	constant 2.667e-04
	value	uniform (0 0 0)
throat	type	pressureInletOutletVelocity
	value	uniform (0 0 0)
		$p$
noseWalls	type	zeroGradient
sphere	type	zeroGradient
throat	type	totalPressure
	p0	uniform -20
	value	uniform 0



# Pre-surgery – CPI boundary conditions

$U$

```

noseWalls
  type    fixedValue
  value   uniform (0 0 0)

sphere
  type    inletOutlet
  value   uniform (0 0 0)

throat
  type    inletOutlet
  value   uniform (0 0 0)
  
```

$p$

```

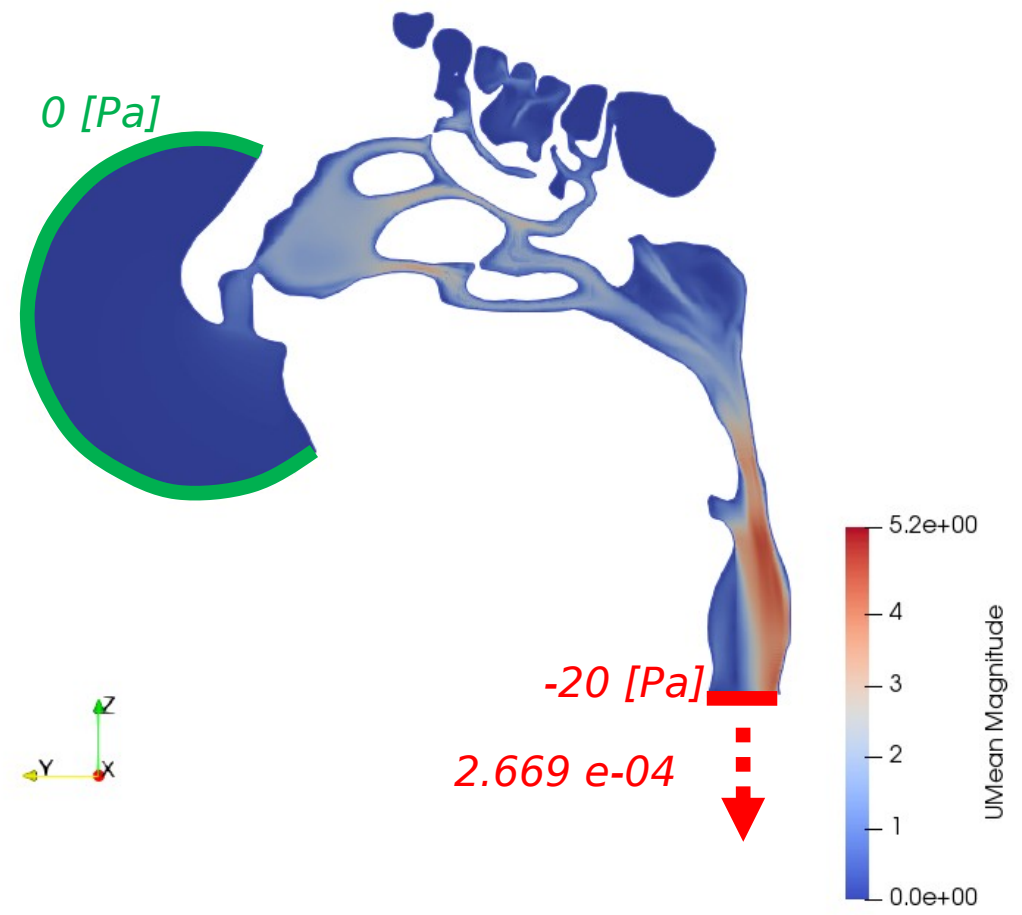
noseWalls
  type    zeroGradient

sphere
  type    totalPressure;
  p0      uniform 0;
  value   uniform 0;

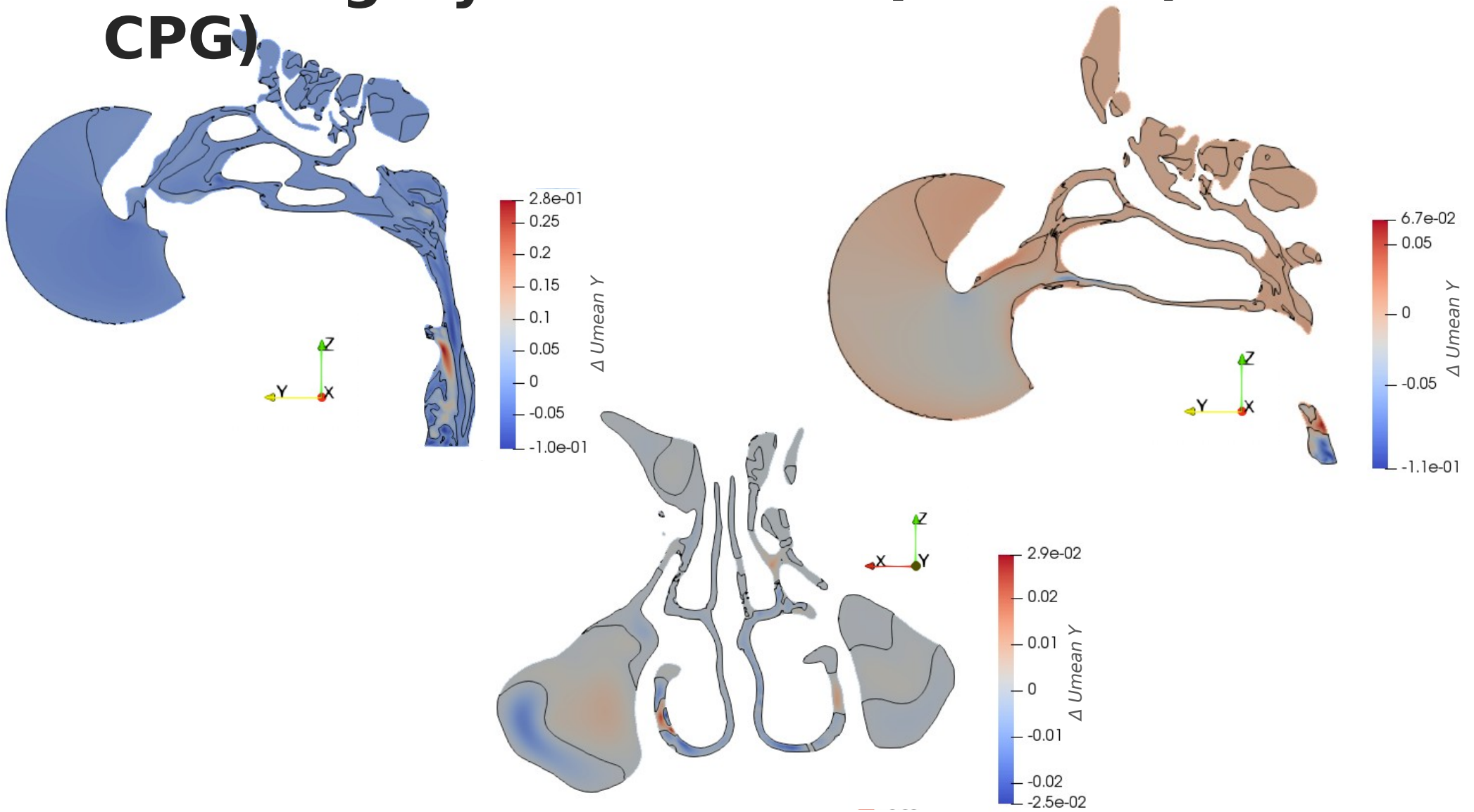
throat
  type    groovyBC;
  variables ( "CPI=-0.005334;" "Qdot=sum(phi);" );
  valueExpression "CPI/Qdot ";
  
```

UniGeValUA uniform -20;

Power Input:  $-0.005334 \pm 6.7e-09$  [W]



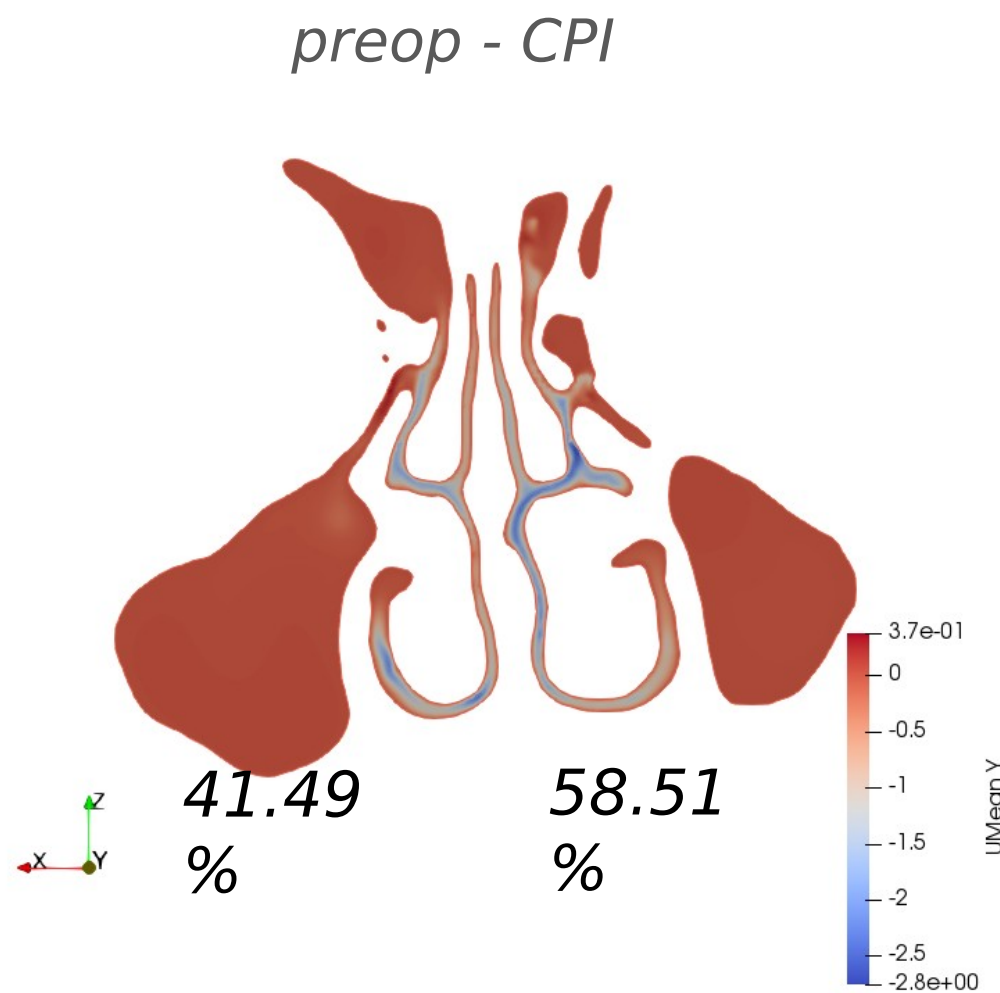
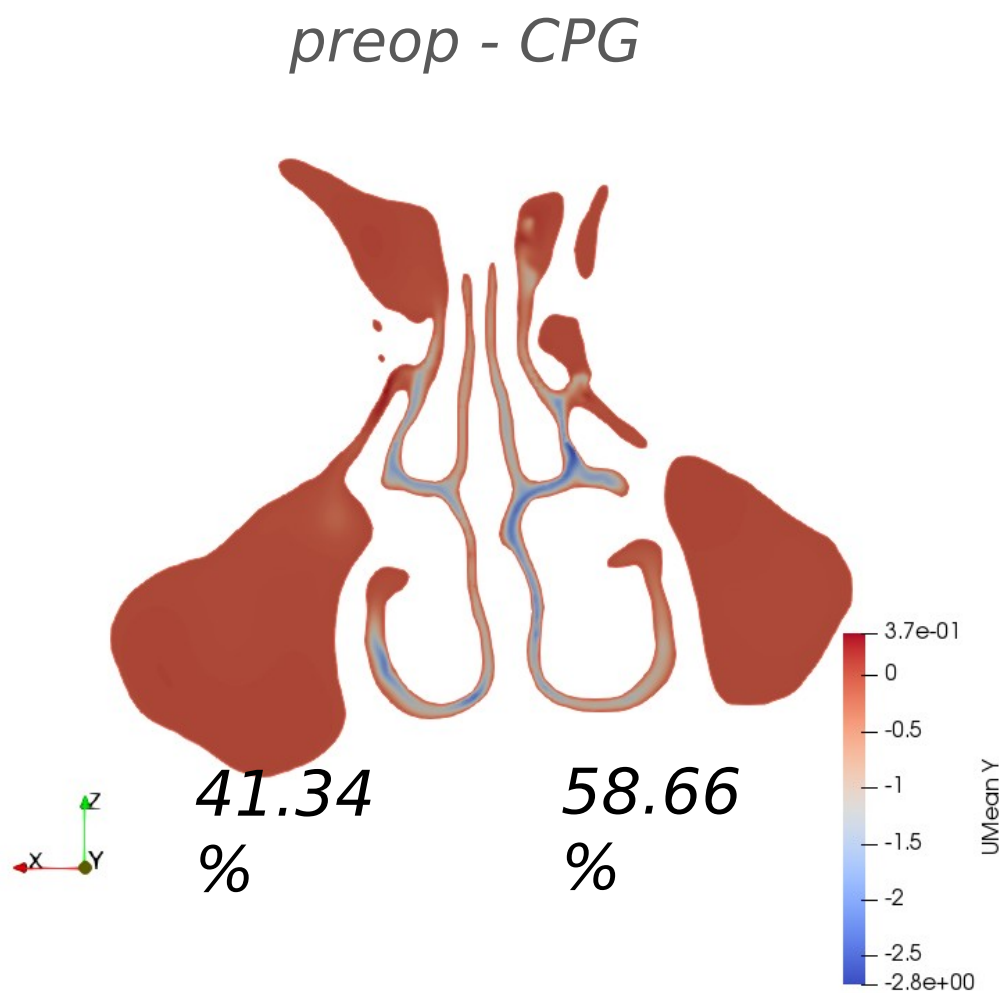
# Pre-surgery results comparison (CPI – CPG)



The figures show the velocity difference between CPI and CPG. In general the differences are small.

Black lines indicate zero-difference

# Results – pre surgery flow distribution



The difference in the distribution of the flowrate for the preop case with CPG and CPI boundary condition is 0.145% .



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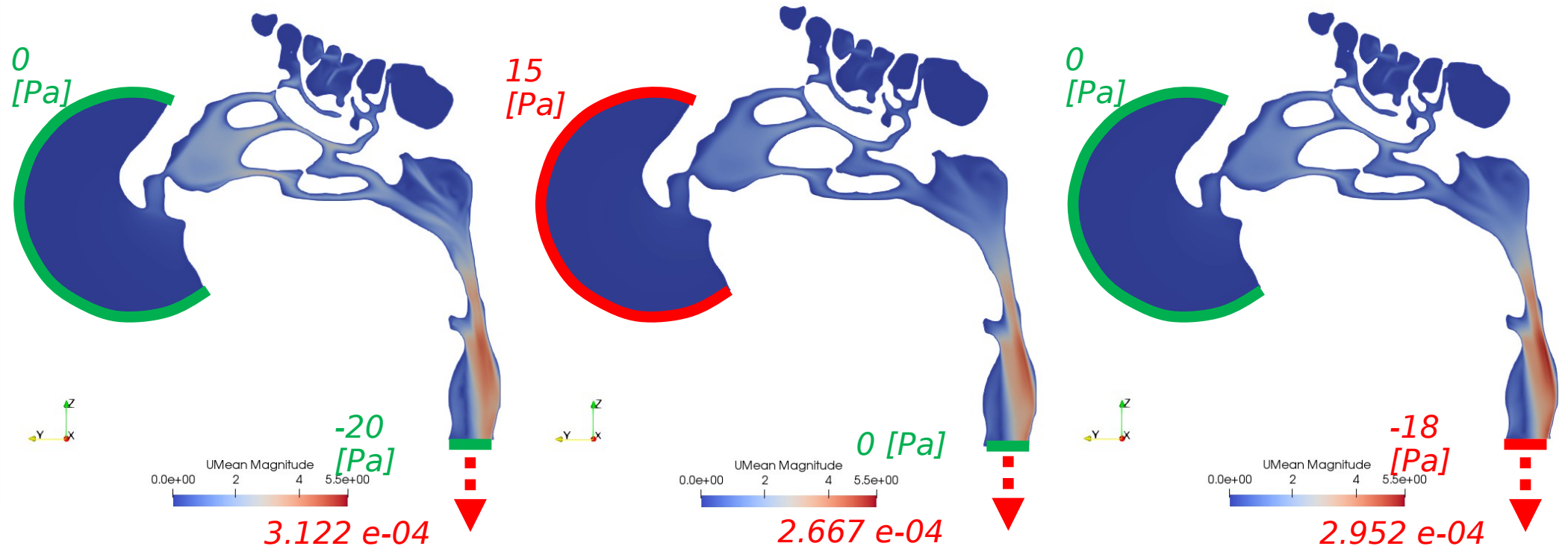
# Post-surgery (postop) results

# Post-surgery results

Postop - CPG

Postop - CFR

Postop - CPI



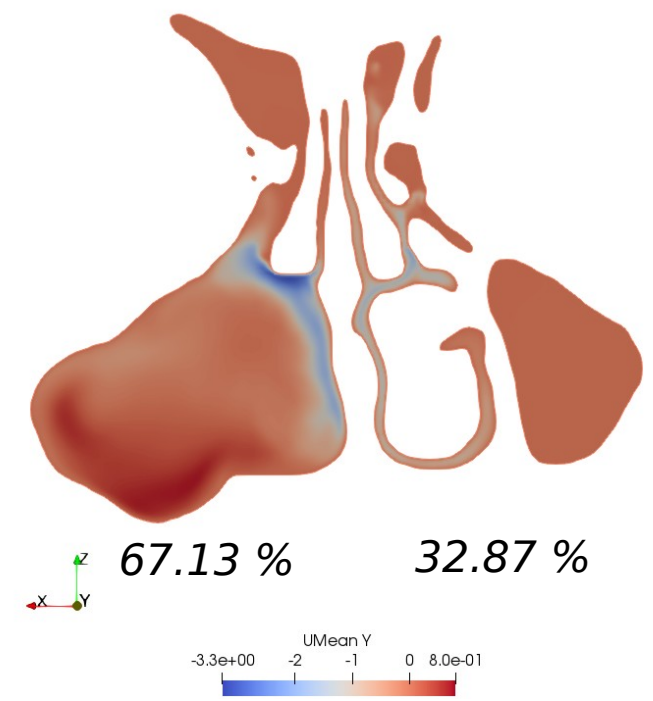
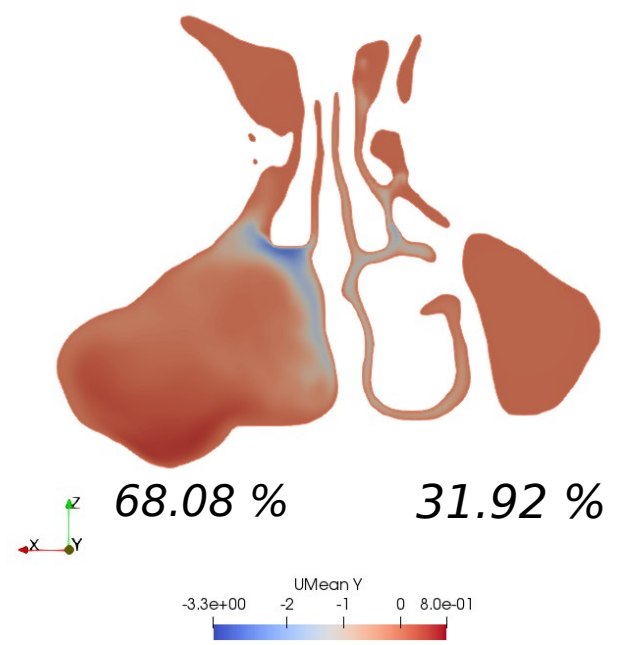
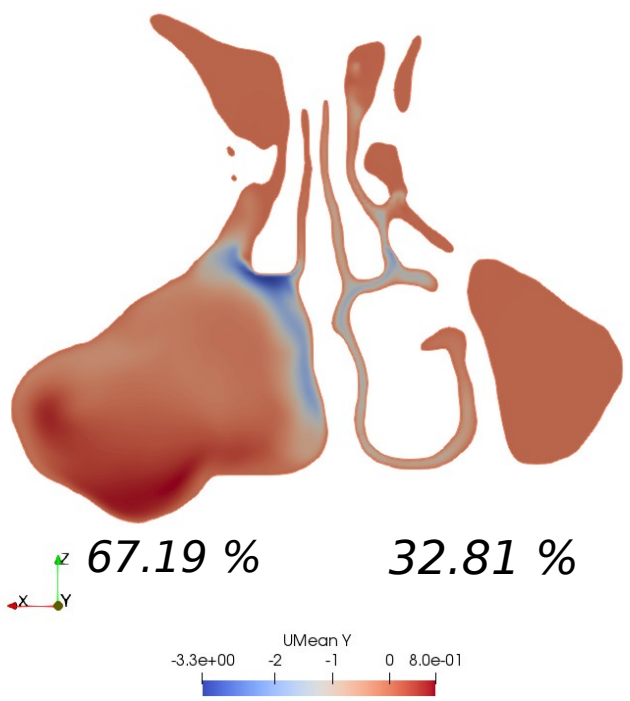
The three boundary conditions shows quite different outcomes.

# Post-surgery results

Postop - CPG

Postop - CFR

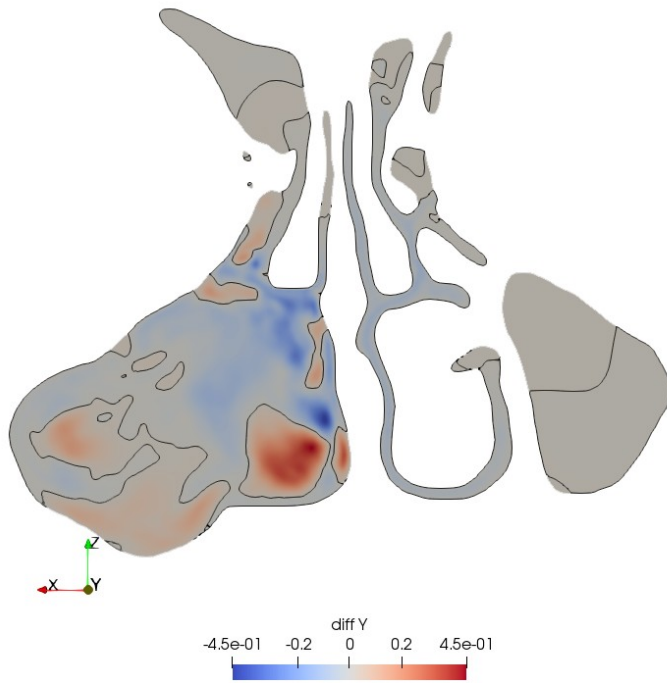
Postop - CPI



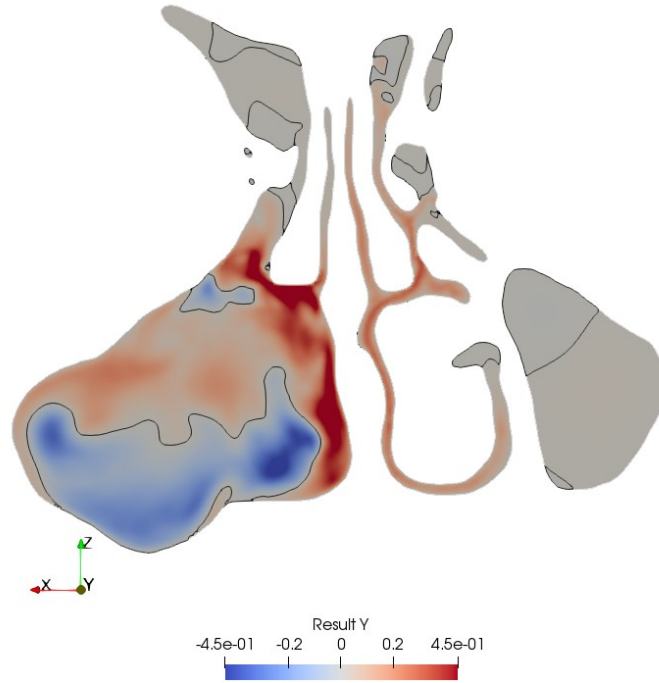
*Flowrate distribution among the nostrils shows almost the same imbalance in all the solutions.*

# Post-surgery results comparison

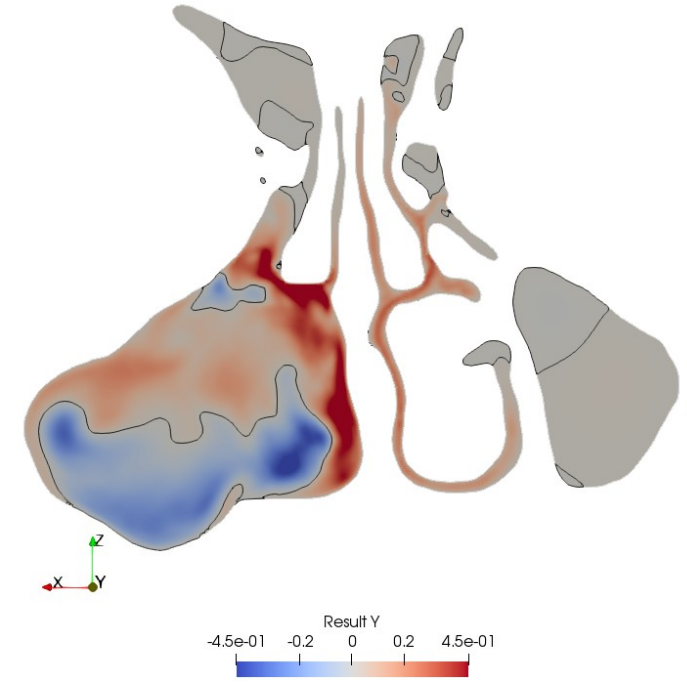
*CPI - CPG*



*CPG - CFR*



*CPI - CFR*



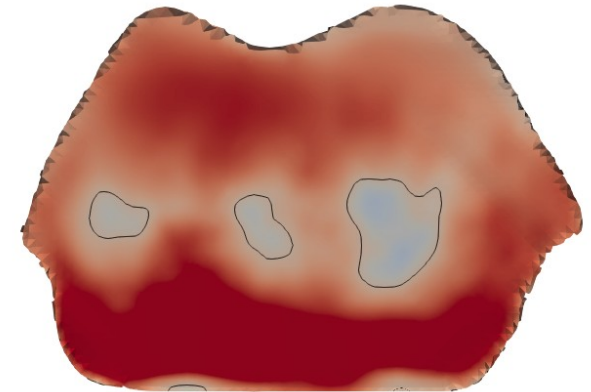
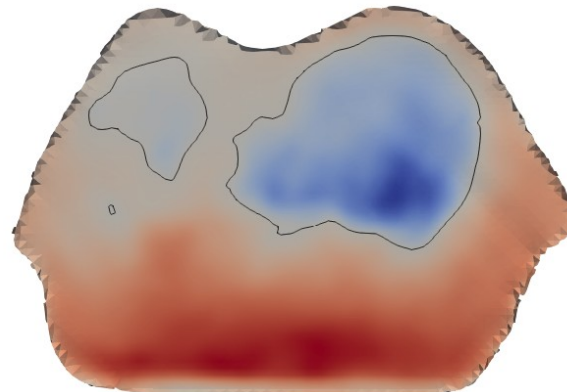
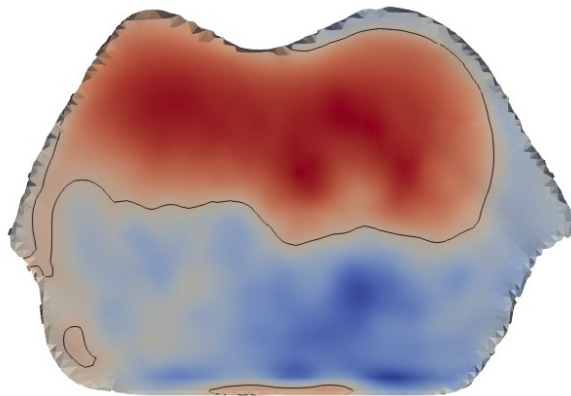
*Y-Umean differences are larger than in the preop case.*

# Post-surgery results comparison

*CPI - CPG*

*CPG - CFR*

*CPI - CFR*



*CPI and CPG boundary conditions show different behaviour at the outlet patch: in the CPI the outflow is more concentrated on the back of the Trachea.*

*It can be noted the mesh degradation close to the patches.*

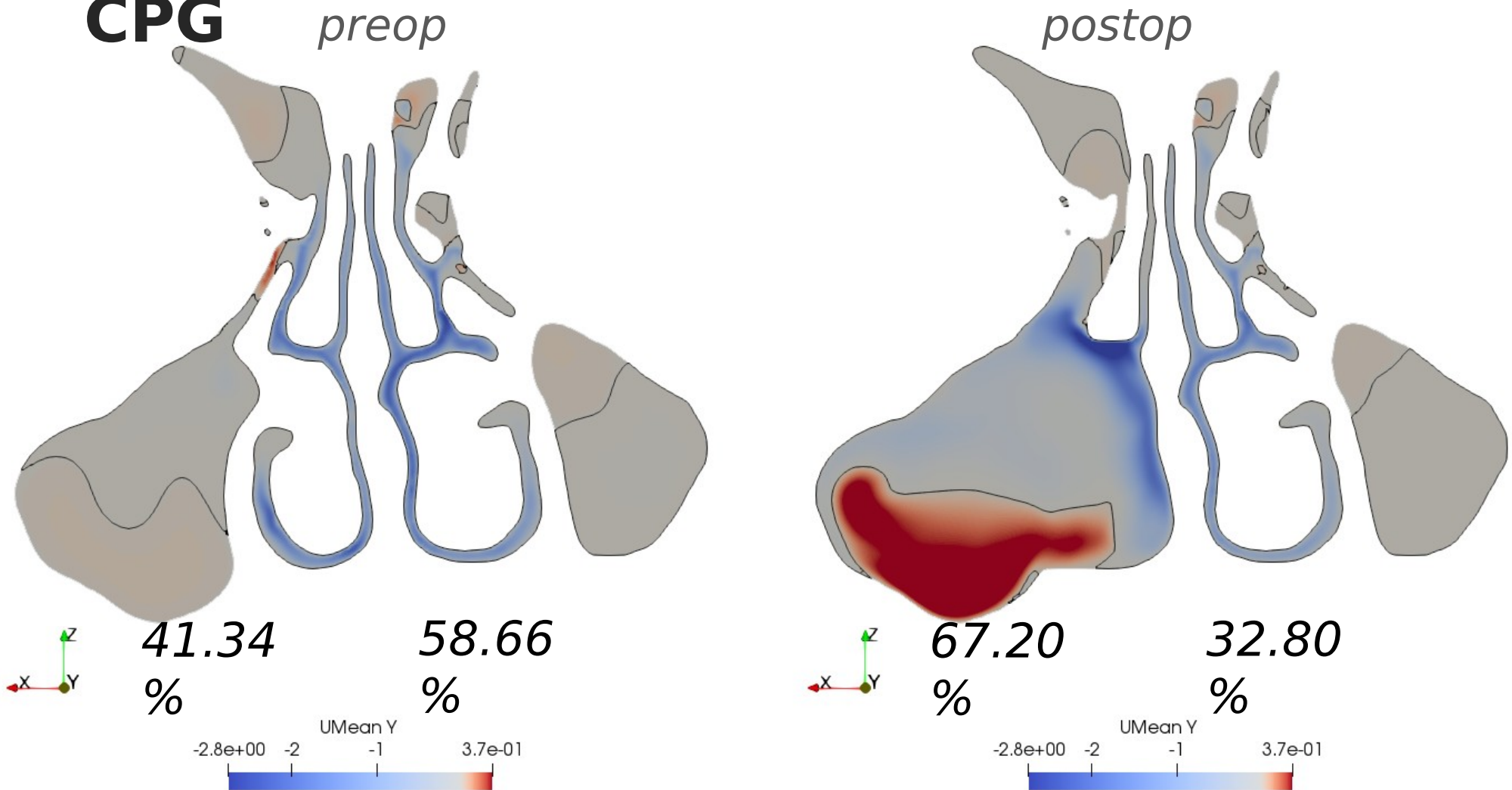


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# Comparison of preop and postop

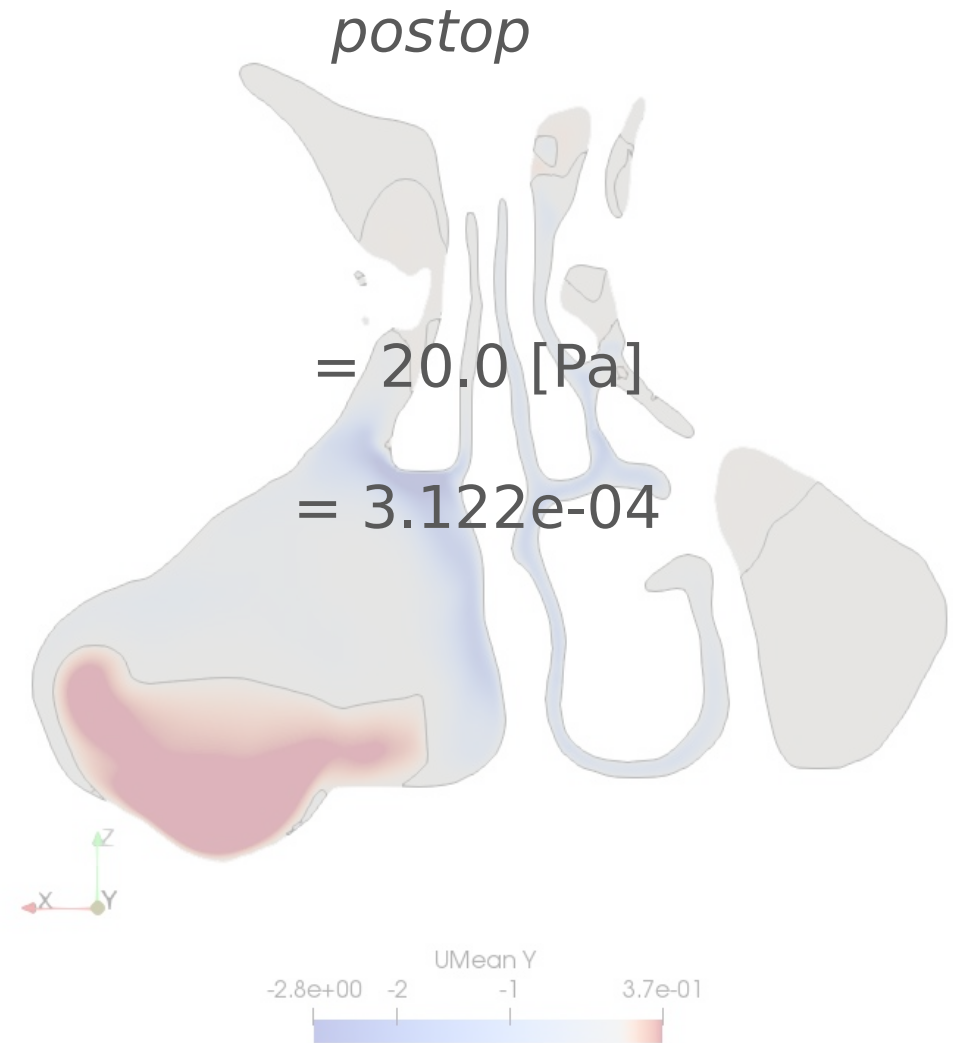
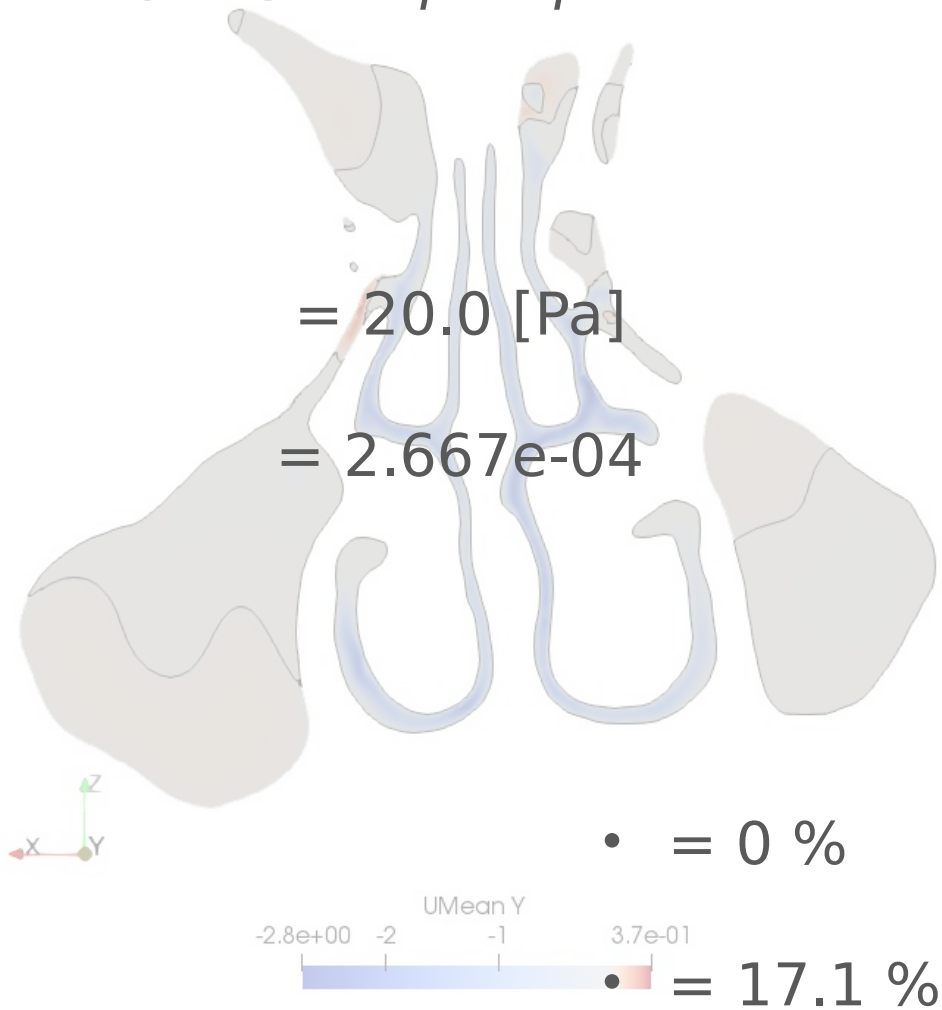
# Results – the effects of surgery with CPG



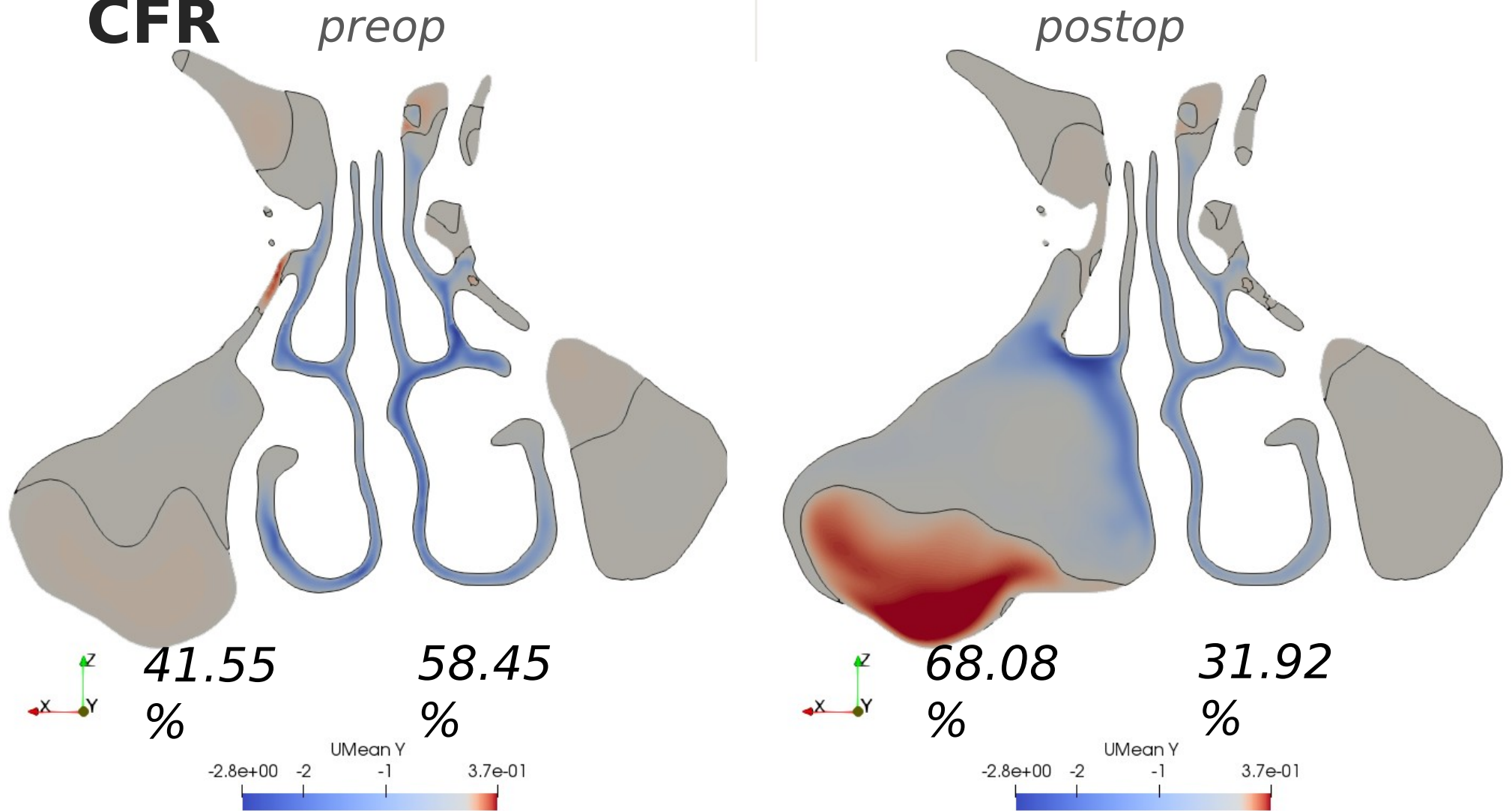
The slices are coloured with the UMean along Y axis. Black line is the zero velocity line.

Flowrate distribution among the nostrils. Huge imbalance after the surgery.

# Results – the effects of surgery with CPG



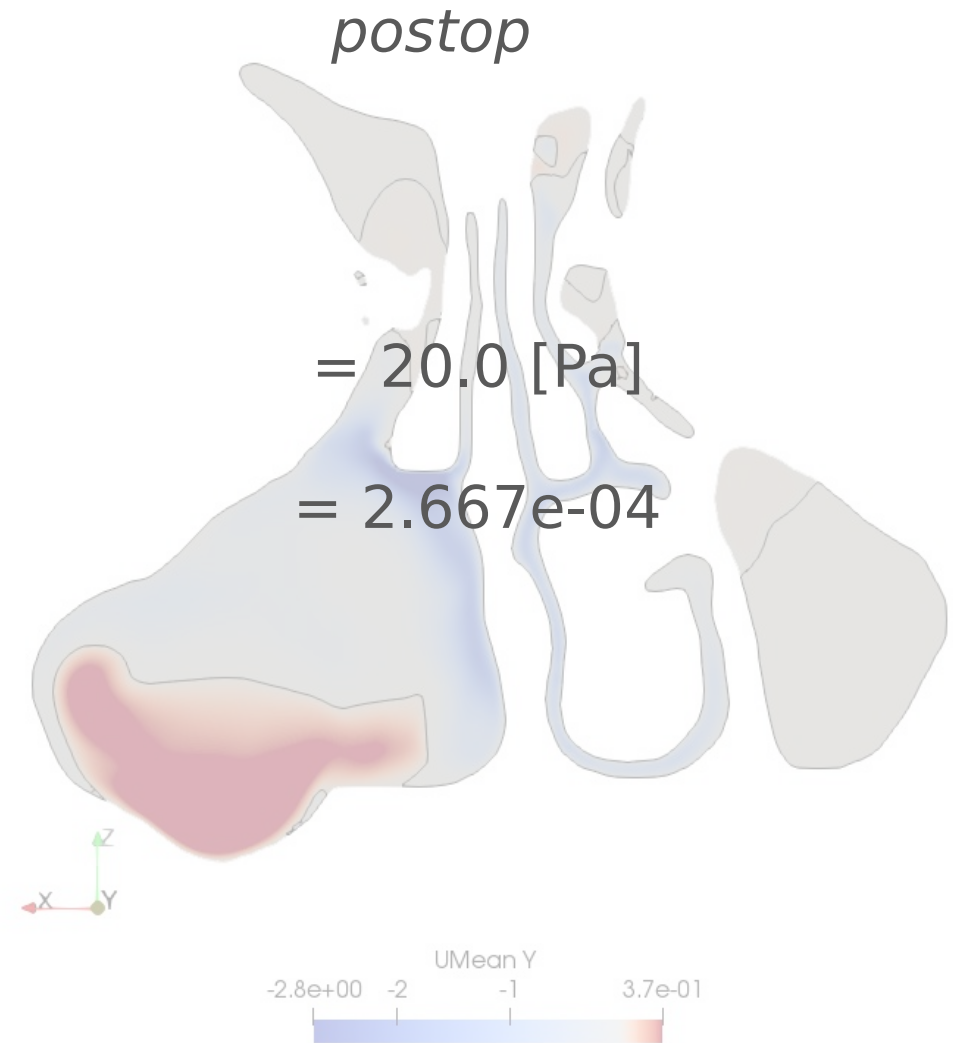
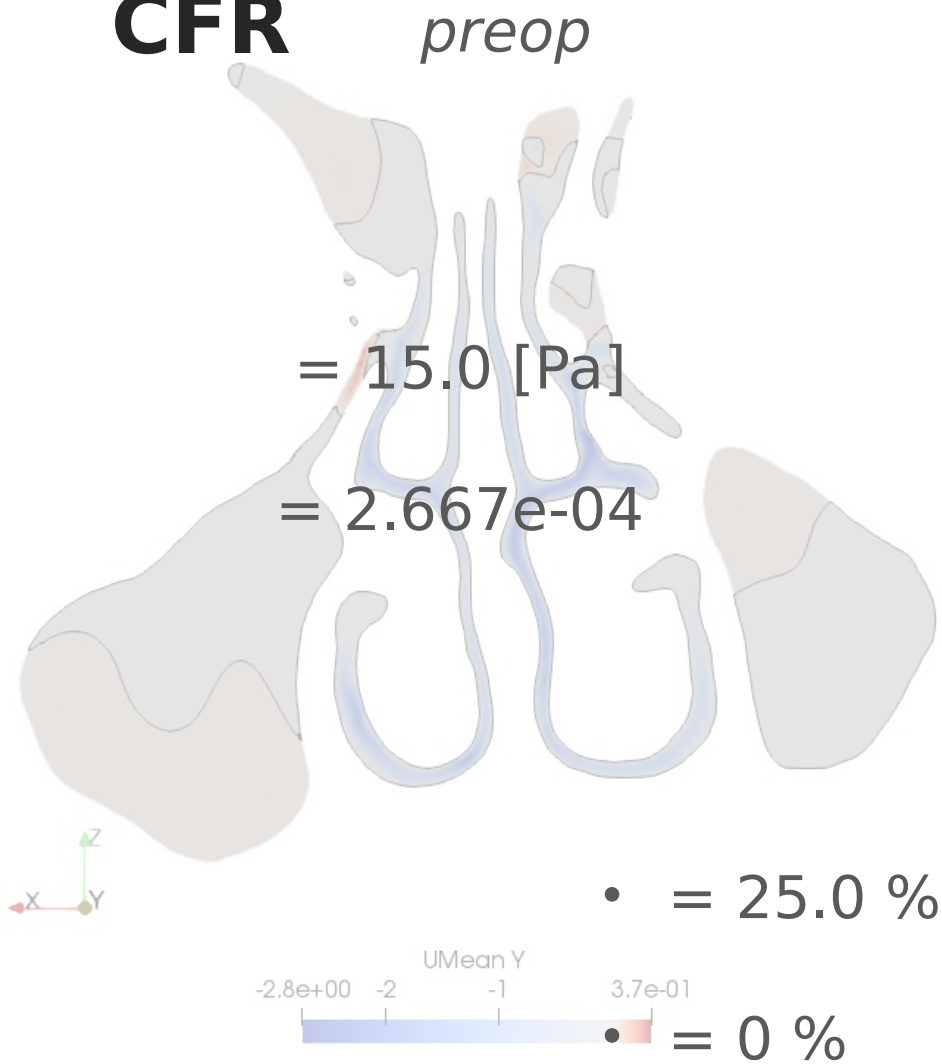
# Results – the effects of surgery with CFR



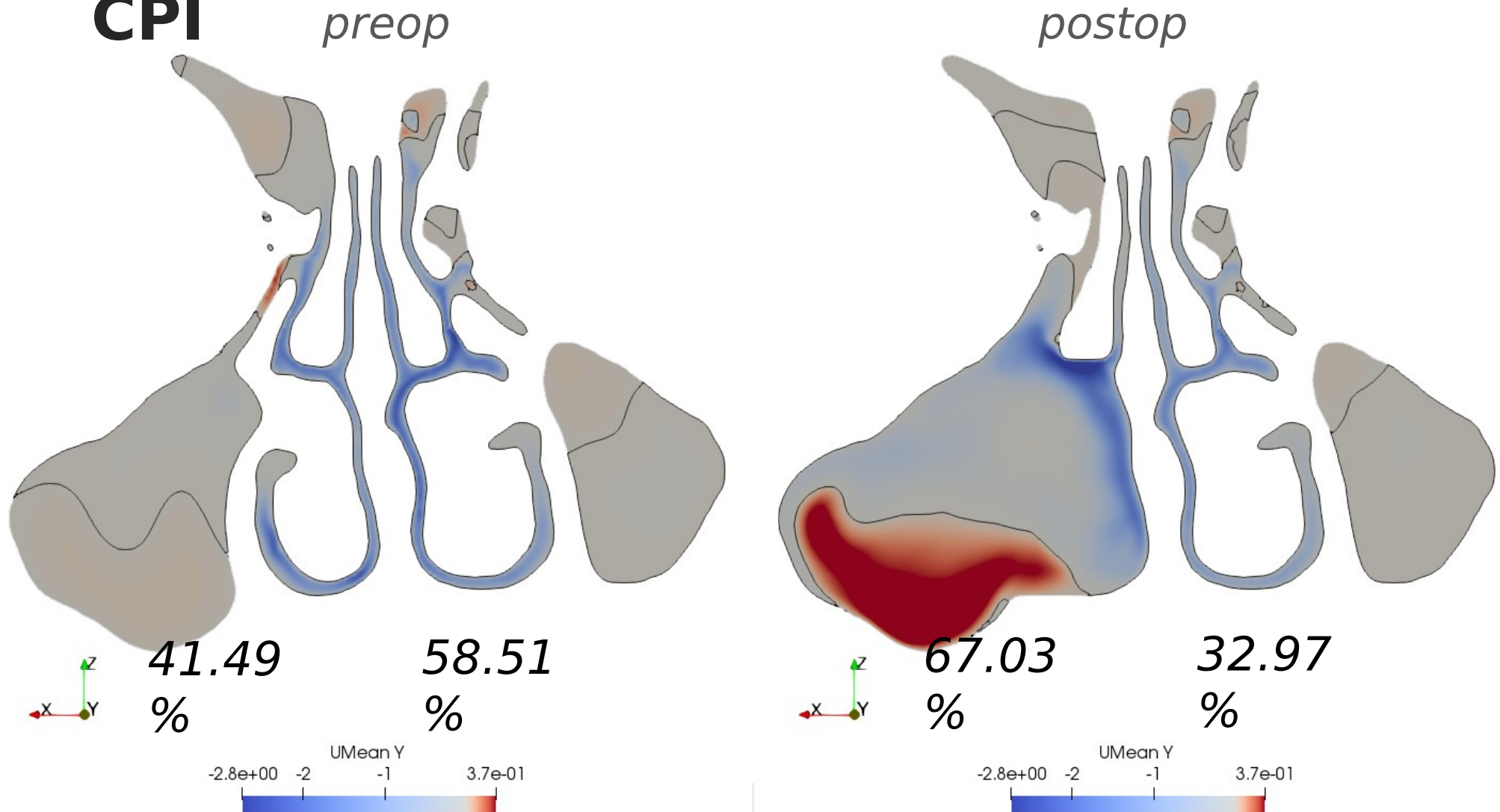
The slices are coloured with the UMean along Y axis. Black line is the zero velocity line.

Flowrate distribution among the nostrils. Huge imbalance after the surgery.

# Results – the effects of surgery with CFR



# Results – the effects of surgery with CPI



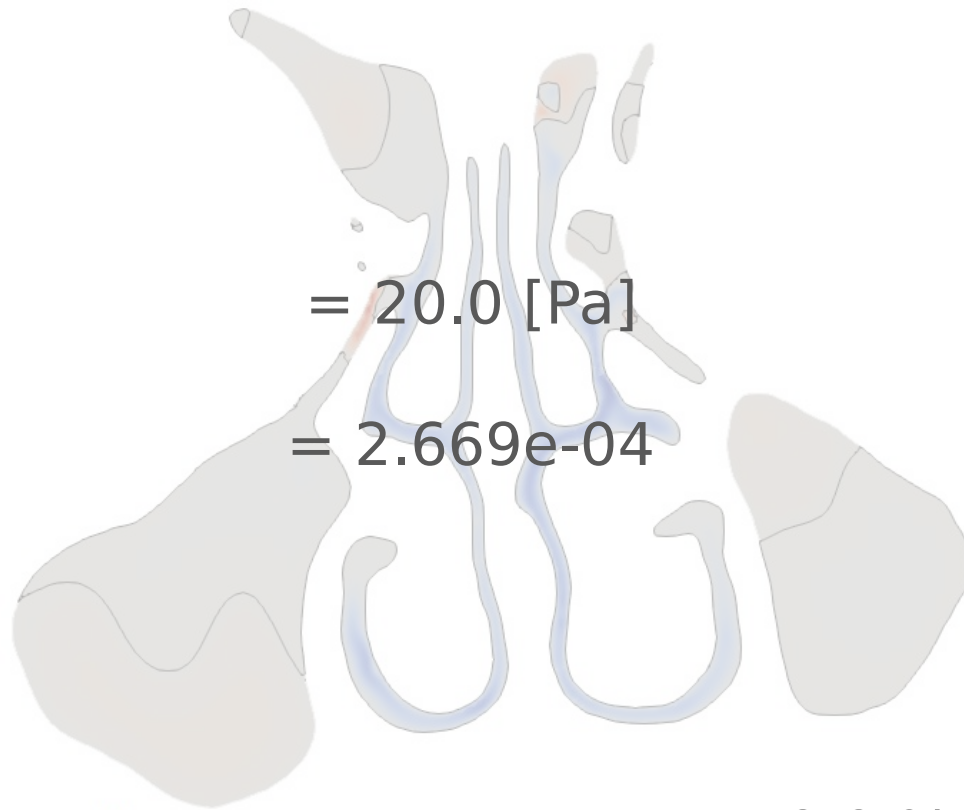
The slices are coloured with the UMean along Y axis. Black line is the zero velocity line.

Flowrate distribution among the nostrils. Huge imbalance after the surgery.

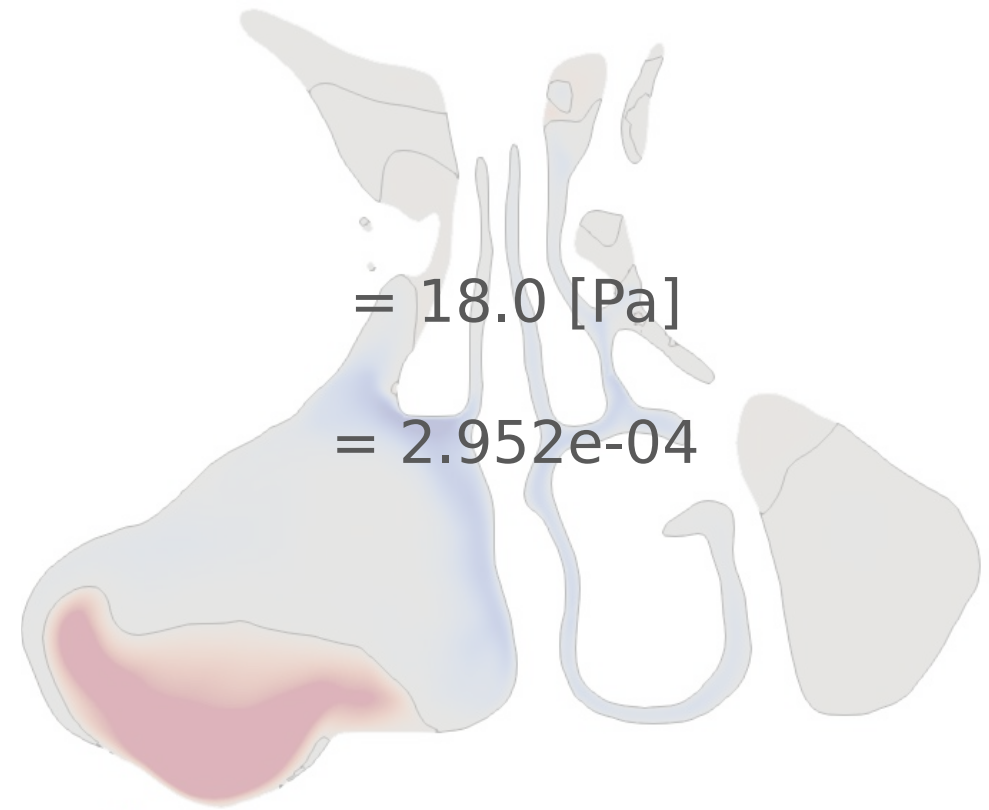
# Results – the effects of surgery with CPI

*preop*

*postop*



= 20.0 [Pa]  
= 2.669e-04



= 18.0 [Pa]  
= 2.952e-04



• = 10.0 %



• = 10.6 %



# Results summary

	CPG preop	CPG postop	CFR preop	CFR postop	CPI preop	CPI postop
	2.665e-4	3.122e-04	2.667e-4	2.667e-04	2.669e-04	2.952e-04
$\Delta P$ [Pa]	20.015	20.024	20.003	15.020	19.998	18.070
CPI [W]	5.738e-03	6.252e-3	5.339e-3	4.010e-3	5.334e-03	5.334e-03
$\Delta$ [%]	17.1 %		25.0 %		10.6 % 10.0%	

# Conclusions

- For the first time, a quantitative and qualitative comparison between CPG, CFR, CPI has been made.

Boundary condition	CPG		CFR		CPI	
cases	Pre-op	Post-op	Pre-op	Post-op	Pre-op	Post-op
	2.665e-04	3.122e-04	2.667e-04	2.667e-04	2.669e-04	2.952e-04
$\Delta P$ [Pa]	20.015	20.024	20.003	15.020	19.998	18.070
CPI [W]	5.738e-03	6.252e-3	5.339e-03	4.010e-03	5.334e-03	5.334e-03
[%]	17.1 %		25.0 %		10.6 % 10.0%	

- Clearly, the differences are large
- No clinician today is able to say which condition is the correct one to use
- CPG is clearly the worst condition since it depends strongly on the shape (for instance trachea length)
- The use of CFR or CPI (or something else) is a future discussion to make with clinicians